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J. P. Dwivedi, V. P. Singh and Radha Krishna Lal

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J. P. DWIVEDI, V. P. SINGH AND RADHA KRISHNA LAL

The dynamic interaction of the twin shells subjected to seismic waves is investigated numerically. It is found that the three-dimensional response of twin shells may differ significantly from the two-dimensional response, and that through-soil interaction between the shells may also be significant.

1. Introduction

The dynamic response of buried pipelines and subway shells for structural stability during earthquakes is important when those structures can potentially be subjected to seismic ground motion. The amplification of seismic motions and stress concentrations may occur as a result of wave scattering around such structures. Mow and Pao [1973] were among the first to study wave diffraction around a cylindrical cavity in an infinite medium and the resulting stress concentration using wave function expansions. An analytic model was presented in [Trifunac 1972] to study horizontal polarized shear (SH) wave scattering at a semicylinder located at the boundary of a half-space. SH wave scattering for various topographies has also been studied in [Wong and Jennings 1975]. Chin et al. [1987] and Liu et al. [1991] studied the response of pipelines buried in back-filled trenches. In reality, many underground structures are constructed in close proximity and the interaction between such closely spaced structures may be significant [Okumura et al. 1992; Guan and Moore 1994]. The purpose of this paper is to investigate the three-dimensional response of a pair of lined cylindrical cavities located in a full-space subjected to incident seismic waves.

2. Governing equations

The geometry of the problem is as shown in Figure 1. The parallel twin shells, denoted I and II, are deeply buried in a viscoelastic ground material described by the shear modulus μ , mass density ρ , and Poisson's ratio σ . The material properties for the linings are described by the sets of variables μ_{I} , ρ_{I} , and σ_{I} and μ_{II} , ρ_{II} , and σ_{II} . For convenience, the index indicating the medium or shell lining is omitted in (2-1)–(2-9), which are valid for each of these three different materials. The displacements and stresses in these solids can be expressed in terms of potentials ϕ , ψ , and χ , which satisfy the following wave equations of motion:

$$\nabla^2 \phi + k_{\alpha}^2 \phi = 0, \quad k_{\alpha} = \frac{\omega}{\upsilon_L}, \tag{2-1}$$

$$\nabla^2 \psi + k_\beta^2 \psi = 0, \tag{2-2}$$

Radha Krishna Lal is the corresponding author.

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$$\nabla^2 \chi + k_\beta^2 \chi = 0, \quad k_\beta = \frac{\omega}{\upsilon_L}, \tag{2-3}$$

where ω is the excitation frequency, υ_L and υ_T denote the longitudinal and transverse wave velocities, and ∇^2 is the Laplace operator in polar cylindrical coordinates.

The factor $e^{-i\omega(t-t_0)}$ is omitted in the text. The general solutions of (2-1), (2-2), and (2-3) take the following forms:

$$\phi(r,\theta,z) = \sum_{n=0}^{\infty} \phi_n = \sum_{n=0}^{\infty} H_n(\alpha_L r) \begin{cases} \cos n\theta \\ \sin n\theta \end{cases} \left(\begin{cases} A \\ B \end{cases}_n e^{\pm i\gamma_L z} \right)$$
$$= \sum_{n=0}^{\infty} H_n(\alpha_L r) (A_n \cos n\theta + B_n \sin n\theta) e^{\pm i\gamma_L z}, \tag{2-4}$$

$$\psi(r,\theta,z) = \sum_{n=0}^{\infty} \psi_n = \sum_{n=0}^{\infty} H_n(\alpha_T r) \begin{cases} \cos n\theta \\ \sin n\theta \end{cases} \left(\begin{cases} C \\ D \end{cases}_n e^{\pm i\gamma_T z} \right)$$
$$= \sum_{n=0}^{\infty} H_n(\alpha_T r) (C_n \cos n\theta + D_n \sin n\theta) e^{\pm i\gamma_T z}, \tag{2-5}$$

$$\chi(r,\theta,z) = \sum_{n=0}^{\infty} \chi_n = \sum_{n=0}^{\infty} H_n(\alpha_T r) \begin{cases} \cos n\theta \\ \sin n\theta \end{cases} \left(\begin{cases} E \\ F \end{cases}_n e^{\pm i\gamma_T z} \right) = \sum_{n=0}^{\infty} H_n(\alpha_T r) (E_n \cos n\theta + F_n \sin n\theta) e^{\pm i\gamma_T z},$$
(2-6)

where H_n are the Bessel functions that satisfy the radiation conditions. Bessel functions of the first kind are used for the liners, and the outgoing wave in the medium is expressed in terms of Hankel functions



Figure 1. Geometry of the problem.

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of the first kind. The wave numbers α_{υ} ($\upsilon = L, T$) are given by

$$\alpha_L = \sqrt{\frac{\omega^2}{\upsilon_L^2 - \gamma_L^2}},\tag{2-7}$$

$$\alpha_T = \sqrt{\frac{\omega^2}{\upsilon_T^2 - \gamma_T^2}}.$$
(2-8)

At the interface between the linings and the medium, the continuity and equilibrium conditions will be enforced:

$$(u_r^{(I)}, u_\theta^{(I)}, u_z^{(I)}) = (u_r, u_\theta, u_z),$$
(2-9)

$$(u_r^{(\text{II})}, u_{\theta}^{(\text{II})}, u_z^{(\text{II})}) = (u_r, u_{\theta}, u_z),$$
(2-10)

$$(\sigma_{rr}^{(\mathrm{I})}, \sigma_{r\theta}^{(\mathrm{I})}, \sigma_{rz}^{(\mathrm{I})}) = (\sigma_{rr}, \sigma_{r\theta}, \sigma_{rz}), \qquad (2-11)$$

$$(\sigma_{rr}^{(\mathrm{II})}, \sigma_{r\theta}^{(\mathrm{II})}, \sigma_{rz}^{(\mathrm{II})}) = (\sigma_{rr}, \sigma_{r\theta}, \sigma_{rz}).$$
(2-12)

For the problem specified in Figure 1, the wave field consists of the incident seismic waves and the waves scattered by shells I and II:

$$\phi = \phi^{i} + \phi^{I}(r, \theta, z) + \phi^{II}(R, \vartheta, z), \qquad (2-13)$$

$$\psi = \psi^{i} + \psi^{I}(r, \theta, z) + \psi^{II}(R, \vartheta, z), \qquad (2-14)$$

$$\chi = \chi^{i} + \chi^{I}(r, \theta, z) + \chi^{II}(R, \vartheta, z), \qquad (2-15)$$

where the superscript i denotes the incident seismic wave.

2.1. Coordinate system transform. In the numerical analysis, the interaction between the two shells is considered by coordinate system transformations (Figure 1(b)). The coordinate transformations from (r, θ) to (R, ϑ) and from (R, ϑ) to (r, θ) are [Watson 1944]

$$H_n(\alpha_{\upsilon}r)\cos n\theta = \sum_{m=0}^{\infty} \frac{\epsilon_m}{2} (H_{n-m}(\alpha_{\upsilon}D) \pm (-1)^m H_{n+m}(\alpha_{\upsilon}D)) J_m(\alpha_{\upsilon}R)\cos m\vartheta, \qquad (2-16)$$

$$H_n(\alpha_{\nu}r)\sin n\theta = \sum_{m=0}^{\infty} \frac{\epsilon_m}{2} (H_{n-m}(\alpha_{\nu}D) \pm (-1)^m H_{n+m}(\alpha_{\nu}D)) J_m(\alpha_{\nu}R)\sin m\vartheta, \qquad (2-17)$$

and

$$H_n(\alpha_{\upsilon}R)\cos n\vartheta = \sum_{m=0}^{\infty} \frac{\epsilon_m}{2} (H_{m-n}(\alpha_{\upsilon}D) \pm (-1)^n H_{m+n}(\alpha_{\upsilon}D)) J_m(\alpha_{\upsilon}r)\cos m\theta, \qquad (2-18)$$

$$H_n(\alpha_{\upsilon}R)\sin n\vartheta = \sum_{m=0}^{\infty} \frac{\epsilon_m}{2} (H_{m-n}(\alpha_{\upsilon}D) \pm (-1)^m H_{m+n}(\alpha_{\upsilon}D)) J_m(\alpha_{\upsilon}r)\sin m\theta, \qquad (2-19)$$

where

$$\epsilon_m = \begin{cases} 1 & \text{if } m = 0, \\ 2 & \text{if } m = 1, 2, \dots \end{cases}$$
(2-20)

2.2. *Incident seismic wave.* The incident seismic wave can be expressed in terms of potentials. The incident *P* wave and *S* wave potentials can be written as

$$\phi^{i} = \phi_0 \exp(ik_\alpha \sin \delta_0 (x \cos \theta_0) + y \sin \theta_0 + ik_\alpha z \cos \delta_0 - i\omega(t - t_0)), \qquad (2-21)$$

$$\psi^{i} = \psi_{0} \exp(ik_{\beta} \sin \delta_{0}(x \cos \theta_{0}) + y \sin \theta_{0} + ik_{\beta}z \cos \delta_{0} - i\omega(t - t_{0})), \qquad (2-22)$$

$$\chi^{i} = \chi_{0} \exp(ik_{\beta} \sin \delta_{0}(x \cos \theta_{0}) + y \sin \theta_{0} + ik_{\beta}z \cos \delta_{0} - i\omega(t - t_{0})), \qquad (2-23)$$

where δ_0 is the angle of the wavefront normal to the *z*-axis (see Figure 1(a)), while θ_0 is the angle that the normal of the intersection between the *XOY* plane and the wavefront makes with the *x*-axis (see Figure 1(b)).

Expressing the wave potentials in polar coordinates yields

$$\phi^{i} = \phi_{0} \sum_{m=0}^{\infty} \frac{\epsilon_{m}}{2} (i)^{m} J_{m}(k_{\alpha}r\sin\delta_{0}) (A_{0,m}\cos m\theta + B_{0,m}\sin m\theta) \exp(ik_{\alpha}z\cos\delta_{0} - i\omega(t-t_{0})), \quad (2-24)$$

$$\psi^{i} = \psi_{0} \sum_{\substack{m=0\\\infty}} \frac{\epsilon_{m}}{2} (i)^{m} J_{m}(k_{\beta}r\sin\delta_{0}) (A_{0,m}\cos m\theta + B_{0,m}\sin m\theta) \exp(ik_{\beta}z\cos\delta_{0} - i\omega(t-t_{0})), \quad (2-25)$$

$$\chi^{i} = \chi_{0} \sum_{m=0}^{\infty} \frac{\epsilon_{m}}{2} (i)^{m} J_{m}(k_{\beta}r\sin\delta_{0}) (A_{0,m}\cos m\theta + B_{0,m}\sin m\theta) \exp(ik_{\beta}z\cos\delta_{0} - i\omega(t-t_{0})), \quad (2-26)$$

where

$$A_{0,m} = \frac{\epsilon_m}{2} (i)^m \cos m\theta_0, \quad B_{0,m} = \frac{\epsilon_m}{2} (i)^m \sin m\theta_0, \tag{2-27}$$

and J_m is a Bessel function of the first kind. Stresses τ_{rr} , $\tau_{\theta\theta}$, and $\tau_{r\theta}$ and displacements u_r and u_{θ} in the medium can be evaluated in terms of potentials (for example, in [Achenbach 1973]).

The potentials in (2-13)–(2-15) must satisfy the stress-free boundary conditions at the half-space and cavity surfaces:

$$\tau_{xy} = \tau_{yy} = 0 \text{ at } y = 0, \quad \tau_{r\theta} = \tau_{rr} = 0 \text{ at } r = r_{I}, \quad \tau_{R\vartheta} = \tau_{RR} = 0 \text{ at } R = R_{I}, \quad (2-28)$$

since ψ^i , χ^i , and ϕ^i already satisfy the stress-free conditions at y = 0.

2.3. *Method of successive reflections.* The approaches of Thiruvenkatachar and Viswanathan and Scheidl and Ziegler are used as follows:

- (1) The incoming wave from shell II is assumed to be zero. Using the boundary conditions, the waves in the lining and the outgoing wave in the medium are determined at shell I.
- (2) Using the boundary conditions, the waves in the lining and the outgoing wave in the medium at shell II are obtained. The incoming wave from shell I is then included using the coordinate transform of the outgoing wave at shell I.
- (3) The waves in the lining and the outgoing wave in the medium at shell I are reevaluated with the incoming wave from shell II included.
- (4) Steps (2) and (3) are repeated until the solution converges.

2.4. *Convergence.* Following the convergence of the method of successive reflections by Thiruvenkatachar and Vishwanathan for wave scattering by a cylinder in a half-space, the convergence of the method is now examined for the specific case of two unlined cavities. Attention is focused on the response of cavity I as a result of the incident wave boundary conditions and wave reflection between the two cavities.

Therefore, with the incoming wave from cavity II excluded, a constant k_1 can be found for the upper bound of the outgoing wave coefficients due to unit incident waves:

$$(|A_n^{\rm I}|, |B_n^{\rm I}|, |C_n^{\rm I}|, |D_n^{\rm I}|, |E_n^{\rm I}|, |F_n^{\rm I}|) < k_{\rm I} \frac{(r\alpha/2)^{2n}}{(n!)^2} (n+1), \quad n = 0, 1, 2, \dots,$$
(2-29)

where α is the maximum of $|\alpha_L|$ and $|\alpha_T|$.

Substituting (2-29) into (2-16) and (2-17), it is then possible to find a positive number k_D so that the following inequality holds for the incoming wave coefficients for cavity II:

$$(|a_m^{\rm II}|, |b_m^{\rm II}|, |c_m^{\rm II}|, |d_m^{\rm II}|, |e_m^{\rm II}|, |f_m^{\rm II}|) < k_{\rm I}k_D \sum_{k=0}^{\infty} \epsilon_m \frac{(r_i\alpha/2)^{2k}(k+1)(m+k-1)!}{(D\alpha/2)^{(m+k)}(k!)^2}.$$
(2-30)

Similarly, the coefficients of the outgoing wave of cavity II are bounded as follows:

$$(|A_m^{\mathrm{II}}|, |B_m^{\mathrm{II}}|, |C_m^{\mathrm{II}}|, |D_m^{\mathrm{II}}|, |E_m^{\mathrm{II}}|, |F_m^{\mathrm{II}}|) < k_{\mathrm{I}}k_Dk_{\mathrm{II}}\sum_{k=0}^{\infty} \epsilon_m \frac{(r_i R_i \alpha^2 / 2)^{2(k+m)} (m+k-1)!(k+1)(m+1)}{(D\alpha/2)^{(m+k)} (k!)^2 (m!)^2}.$$
(2-31)

The incoming wave for cavity I can be obtained from the outgoing wave from cavity II using the coordinate transform once more:

$$(|a_{n}^{\mathrm{I}}|, |b_{n}^{\mathrm{I}}|, |c_{n}^{\mathrm{I}}|, |d_{n}^{\mathrm{I}}|, |e_{n}^{\mathrm{I}}|, |f_{n}^{\mathrm{I}}|) < k_{\mathrm{I}}k_{D}^{2}k_{\mathrm{II}}\sum_{m=0}^{\infty}\epsilon_{n}\sum_{k=0}^{\infty}\epsilon_{m}\frac{(r_{i}R_{i}\alpha^{2}/4)^{2(k+m)}(m+k-1)!(m+n-1)!(k+1)(m+1)}{(D\alpha/2)^{(2m+k+n)}(k!)^{2}(m!)^{2}}.$$
 (2-32)

It can be proved that the series defined by

$$k_D^n = k_D^2 \sum_{m=0}^{\infty} \epsilon_n \sum_{k=0}^{\infty} \epsilon_m \frac{(r_i R_i \alpha^2 / 4)^{2(k+m)} (m+k-1)! (m+n-1)! (k+1) (m+1)}{(D\alpha/2)^{(2m+k+n)} (k!)^2 (m!)^2}$$
(2-33)

is uniformly convergent for a given n. Hence, the outgoing wave for cavity I (with incoming wave from cavity II included) is subject to

$$(|A_{n}^{\mathrm{I}}|, |B_{n}^{\mathrm{I}}|, |C_{n}^{\mathrm{I}}|, |D_{n}^{\mathrm{I}}|, |E_{n}^{\mathrm{I}}|, |F_{n}^{\mathrm{I}}|) < k_{\mathrm{I}}(1 + k_{1}k_{D}^{n}k_{\mathrm{II}})\frac{(r_{i}\alpha^{2}/2)^{2n}}{(n!)^{2}}.$$
(2-34)

Following m iterations of the method of successive reflections, the outgoing wave of cavity I gives

$$(|A_n^{\rm I}|, |B_n^{\rm I}|, |C_n^{\rm I}|, |D_n^{\rm I}|, |E_n^{\rm I}|, |F_n^{\rm I}|) < \sum_{j=0}^{\infty} k_{\rm I} (k_1 k_D^n k_{\rm II}) \frac{(r_i \alpha/2)^{2n}}{(n!)^2} = k_{\rm I} \frac{1 - (k_{\rm I} k_D^n k_{\rm II})^{(m+1)}}{1 - k_{\rm I} k_D^n k_{\rm II}} \frac{(r_i \alpha/2)^{2n}}{(n!)^2}.$$
(2-35)

The method of successive reflections therefore converges provided $k_{I}k_{D}^{n}k_{II} < 1$. The spacing D between the two cavities in (2-33) can be changed for a given r_{i} and R_{i} so that $k_{I}k_{D}^{n}k_{II} < 1$. Therefore the procedure converges if either of the following two conditions is satisfied:

- (1) The wavelength is long enough or there is sufficient spacing between the two cavities.
- (2) The $k_{\rm I}k_D^n k_{\rm II} < 1$ values for the dominant terms (for example, n = 2) are particularly small ($k_{\rm I}k_D^n k_{\rm II} \ll 1$).

It is remarkable to note that, in some specific problems, the expected divergence may also be removed by a large number of iterations of the technique applied in this paper.

3. Numerical study

3.1. *Geometry.* The response of the twin lined shells to seismic ground motion is clearly a function of a considerable number of geometrical and material properties. In order to simplify this study, the response will be examined for shells of equal radius $(r_i = R_i)$ and with identical linings $(r_e = R_e, \mu = \mu_I = \mu_{II})$. The Poisson's ratios for the medium and liners are 0.30 and 0.35, respectively. Furthermore, the center-to-center spacing of the shells will be taken as $D = 3.50r_i$, $4.50r_i$, or ∞ , and the liner thickness, $r_e - r_i = r_i/8$, and the incident seismic waves examined will be parallel to either the *YOZ* ($\theta_0 = \pi/2$) or the *XOZ* ($\theta_0 = 0$) plane. The two shells have axes parallel to *z*, so that for $\theta_0 = \pi/2$ the shells have symmetric response about the *YOZ* plane and only one shell response needs to be reported. Alternatively, for $\theta_0 = 0$ the shell responses are different but symmetric about the horizontal diameters. For all the results presented, the displacements and stresses are normalized by k_β and μk_β^2 , respectively.

3.2. Incident S wave: interaction between the shells. To examine the interaction of the two shells, Figure 2 shows predictions of the normalized hoop stress for shells at three different spacings, $D/r_i = 3.50$, 4.50, and ∞ , and for two different incident angles, $\delta_0 = \pi/2$ and $\delta_0 = \pi/3$. To determine the effect of the frequency of the seismic ground motion, a range of normalized vibration frequencies $a_0 = \omega r_i/\upsilon_T$ is considered.



Figure 2. Maximum stress versus normalized frequency: response to incident *S* waves at the inside surface of the liner, at position z = 0 ($\theta_0 = \pi/2$, $\mu/\mu_m = 3$) for $\delta_0 = \pi/2$ (left) and $\delta_0 = \pi/3$ (right).



Figure 3. Maximum displacements U_r^* (left), U_{θ}^* (middle), and U_Z^* (right) versus normalized frequency: response to incident *S* waves at the outside surface of the liner, at position z = 0 ($\theta_0 = 0$, $D/r_i = 3.50$).

At low vibration frequency, $a_0 < 0.5$, the shell-to-shell interaction for all three shell spacings is insignificant. As normalized frequency a_0 increases, however, the shell response becomes significantly influenced by the spacing. For $0.5 < a_0 < 2.5$, the hoop stress $\sigma_{\theta\theta}^*$ is up to 35% less for an isolated shell $D/r_i = \infty$ than for $D/r_i = 3.50$, 4.50. For shells with $D < 3r_i$, the increase in hoop stress due to through-soil interaction is expected to be even more important.

3.3. Incident *S* wave: influence of lining modulus. To examine the influence of lining design, Figure 3 shows solutions for the maximum normalized displacements U_r^* , U_{θ}^* , and U_z^* for the specific normalized frequency $a_0 = 1$, for a range of modular ratios μ/μ_m and for three different incident angles δ_0 . Figure 4 shows solutions for the maximum normalized stresses $\sigma_{\theta\theta}^*$ and $\sigma_{\theta z}^*$ for the specific normalized frequency $a_0 = 1$, for a range of modular ratios μ/μ_m and for three different incident angles δ_0 .



Figure 4. Maximum normalized stresses $\sigma_{\theta\theta}^*$ (left) and $\sigma_{\theta z}^*$ (right) versus normalized frequency: response to incident *S* waves at the inside surface of the liner, at position z = 0 ($\theta_0 = 0$, $D/r_i = 3.50$).



Figure 5. Maximum stress versus normalized frequency: response to incident S waves at the inside surface of the liner, at position z = 0 ($\delta_0 = \pi/2$, $\mu/\mu_m = 3$) for shells I (left) and II (right).

The normalized radial and circumferential displacements are the greatest responses under plane strain conditions, $\delta_0 = \pi/2$. They peak at low modulus ratio, μ/μ_m , but remain nearly constant when μ/μ_m ranges from 0.25 to 4. Normalized longitudinal displacement also decreases monotonically as μ/μ_m increases, but is generally more sensitive to any changes in the modular ratio.

Normalized stress amplitudes $\sigma_{\theta\theta}^*$ and $\sigma_{\theta z}^*$ at the inner surface of the lining all steadily decrease as the surrounding modulus, μ_m , decreases relative to that of liners μ . All values tend towards zero as modular ratio μ_m/μ approaches zero. As noted for the normalized displacement, the two-dimensional loading condition $\delta_0 = \pi/2$ induces the greatest stress. Circumferential stresses $\sigma_{\theta\theta}^*$ are largest in magnitude.

3.4. *Incident S wave: influence of incident angle* θ_0 . The behavior of the two shells is now examined under the plane strain condition ($\delta_0 = \pi/2$) with incident angles $\theta_0 = 0$, $\pi/3$, and $\pi/2$, and a range of vibration frequencies. The shells are spaced at distance $D/r_i = 3.50$ and have a lining modulus three times that of the ground medium ($\mu/\mu_m = 3$). Figure 5 shows the amplitude of the normalized stress $\sigma_{\theta\theta}^*$ for shells I and II.

For an isolated shell, the response is independent of the incident angle θ_0 . However, for these closely spaced shells, the response is significantly affected by the orientation of the incoming wave. Firstly, Figure 5 clearly shows that the effect of the incident angle θ_0 is frequency dependent. The circumferential stress $\sigma_{\theta\theta}^*$ of shell I has the greatest amplitude when $\theta_0 = \pi/3$ at the normalized frequency $a_0 = 0.8$. For larger values of normalized frequency, $a_0 > 1.5$, the responses are less frequency dependent and the vertically moving seismic wave ($\theta_0 = \pi/2$) induces the highest hoop stresses in both shells. For shell II the peak stress occurs when $0.2 < a_0 < 0.8$, but is not quite as great and is less dependent on θ_0 .

3.5. *Incident P wave.* Figure 6 shows normalized radial, circumferential, and longitudinal displacements for $\theta = \pi/2$, for three incident wave angles δ_0 and a range of the modular ratios. The trends are similar to those observed earlier for shell response to shear waves, but with stress and displacement amplitudes somewhat smaller. The response increases as the modular ratio increases. Figure 7 shows



Figure 6. Maximum displacements U_r^* (left), U_{θ}^* (middle), and U_Z^* (right) versus normalized frequency: response to incident *P* waves at the outside surface of the liner, at position z = 0 ($\theta_0 = \pi/2$, $D/r_i = 3.50$).



Figure 7. Maximum stresses $\sigma_{\theta\theta}^*$ (left), $\sigma_{\theta z}^*$ (middle), and σ_{zz}^* (right) versus normalized frequency; response to incident *P* waves at the inside surface of the liner, at position z = 0 ($\theta_0 = 0$, $D/r_i = 3.50$).

solutions for maximum normalized stresses $\sigma_{\theta\theta}^*$, $\sigma_{\theta z}^*$, and σ_{zz}^* for the specific normalized frequency $a_0 = 1$, for a range of modular ratios μ/μ_m and three different incident angles δ_0 . Compared to the normalized displacement amplitudes, the stresses are more sensitive to changes in the modular ratios.

3.6. Incident *P* wave: interaction between the shells. The interaction between the shells is shown in Figure 8 different angles, $\delta_0 = \pi/2$ and $\pi/4$. Unlike the prediction made for the shell response to an *S* wave, the hoop stress $\sigma_{\theta\theta}^*$ decreases rapidly as the frequency is increased. The through-soil interaction between the shells is not as significant as it is for an incident *S* wave. At most frequencies, the stress predictions for the closely spaced shells are higher than those for an isolated shell, though the difference is not very significant.

3.7. *Incident P wave: influence of incident angle.* Figure 9 shows the hoop stress response for shells I and II. The peak hoop stress is produced for an incident angle $\delta_0 = 0$ and a normalized frequency $a_0 = 1$.



Figure 8. Maximum stress versus normalized frequency: response to incident *P* waves at the inside surface of the liner, at position z = 0 ($\theta_0 = \pi/2$, $\mu/\mu_m = 3$) for $\delta_0 = \pi/2$ (left) and $\delta_0 = \pi/4$ (right).



Figure 9. Maximum stress versus normalized frequency; response to incident *P* waves at the inside surface of the liner, at position z = 0 ($\delta_0 = \pi/2$, $\mu/\mu_m = 3$, and $D/r_i = 3.50$) for shells I (left) and II (right).

However, less response is induced in shell II for incident angle $\theta_0 = 0$ since it is shielded by shell I (shell II is in the "shadow" of shell I).

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J. P. DWIVEDI: jpd@bhu.ac.in

Department of Mechanical Engineering, Indian Institute of Technology (Banaras Hindu University), Varanasi 221005, India

V. P. SINGH: vp_singh56@yahoo.co.in Department of Mechanical Engineering, Indian Institute of Technology (Banaras Hindu University), Varanasi 221005, India

RADHA KRISHNA LAL: radhakrishna7730gmail.com

Department of Mechanical Engineering, Indian Institute of Technology (Banaras Hindu University), Varanasi 221005, India

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