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REDUCIBLE AND IRREDUCIBLE FORMS OF STABILISED GRADIENT ELASTICITY IN DYNAMICS

HARM ASKES AND INNA M. GITMAN

The continualisation of discrete particle models has been a popular tool to formulate higher-order gradient elasticity models. However, a straightforward continualisation leads to unstable continuum models. Padé approximations can be used to stabilise the model, but the resulting formulation depends on the particular equation that is transformed with the Padé approximation. In this contribution, we study two different stabilised gradient elasticity models; one is an irreducible form with displacement degrees of freedom only, and the other is a reducible form where the primary unknowns are not only displacements but also the Cauchy stresses — this turns out to be Eringen's theory of gradient elasticity. Although they are derived from the same discrete model, there are significant differences in variationally consistent boundary conditions and resulting finite element implementations, with implications for the capability (or otherwise) to suppress crack tip singularities.

1. Introduction

Gradient elasticity is a methodology to enrich the continuum equations of elasticity with additional higher-order spatial (and occasionally temporal) derivatives of certain state variables. There are different versions of gradient elasticity, such as those equipped with strain gradients, stress gradients and acceleration gradients; see for instance [Askes and Aifantis 2011] for a recent (but by no means complete) review.

Certain formats of gradient elasticity bear a close relationship with discrete lattice models of materials with microstructure; indeed, it is often possible to derive gradient elasticity theories by continualising the response of a discrete model, for instance using Taylor series approximations [Chang and Gao 1995; Mühlhaus and Oka 1996; Suiker et al. 2001a; Suiker et al. 2001b; Ioannidou et al. 2001; Askes and Metrikine 2005]. However, such models often suffer from intrinsic deficiencies, such as loss of stability in dynamics and loss of uniqueness in statics [Askes et al. 2002]. This can be amended by applying Padé approximations or similar

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Figure 1. One-dimensional chain of masses connected by springs.

techniques, as has for instance been demonstrated in [Rosenau 1984; Rubin et al. 1995; Chen and Fish 2001; Andrianov 2002; Andrianov et al. 2003; Charlotte and Truskinovsky 2008]. Thus, stabilised gradient elasticity theories can be formulated that maintain their close link with discrete lattice models, thereby facilitating simple identification of the higher-order constitutive parameters (usually known as "intrinsic length scales" or "microstructural length scales").

In this paper, we compare two versions of stabilised gradient elasticity. Both can be derived from the response of a discrete lattice model, which is shown for the one-dimensional case. Variational formulations are presented for the multidimensional extensions. Throughout, a distinction is made between the so-called *irreducible form* where the only unknowns are the displacements and the *reducible form* where the unknowns are the displacements as well as the Cauchy stresses. The difference between these two forms has important consequences for the variationally consistent boundary conditions and finite element implementations. A numerical example will show the ability (or otherwise) of the two formulations to suppress singularities — this has historically been an important motivation for using gradient elasticity theories, and certain formats have been demonstrated to remove singularities even under restrictive conditions such as anisotropic material behaviour and bimaterial interface cracks [Kwong and Gitman 2012]. We also discuss the relation of the reducible form with Eringen's [1983] differential theory of nonlocal elasticity.

2. Continualisation of the response of a discrete chain

To illustrate the concepts of continualisation (this section) and stabilisation via Padé approximations (Section 3), the one-dimensional chain of particles and springs in Figure 1 is studied. All particles have mass M, and all springs have stiffness K. Furthermore, the interparticle distance is denoted by d. The equation of motion of particle n thus reads

$$M\ddot{u}_n = K(u_{n+1} - 2u_n + u_{n-1}) \tag{1}$$

where u_i is the displacement of particle *i*. A continuum approximation is obtained by replacing u_n with u(x) and $u_{n\pm 1}$ with $u(x \pm d)$. Taylor series expansions are applied according to

$$u(x \pm d) = u(x) \pm d\frac{\partial u}{\partial x} + \frac{1}{2}d^2\frac{\partial^2 u}{\partial x^2} \pm \frac{1}{6}d^3\frac{\partial^3 u}{\partial x^3} + \frac{1}{24}d^4\frac{\partial^4 u}{\partial x^4} \pm \cdots$$
(2)

so that (1) can be rewritten as

$$\rho \ddot{u} = E \left(\frac{\partial^2 u}{\partial x^2} + \frac{1}{12} d^2 \frac{\partial^4 u}{\partial x^4} + \cdots \right)$$
(3)

where the mass density $\rho = M/Ad$ and the Young's modulus E = Kd/A, with A the (unit) cross-sectional area of the system. Multidimensional formulations in the spirit of (3) have been derived by Chang and Gao [1995], Mühlhaus and Oka [1996] and Suiker et al. [2001a; Suiker et al. [2001b], among others.

Apart from the lowest-order, standard terms, (3) also contains higher-order terms proportional to d^2 , d^4 , etc. These additional terms capture the microstructural effects that are present in the discrete model of (1) but that are absent in standard continuum theories as retrieved by taking d = 0 in (3). The simplest continuum model that incorporates microstructural effects is obtained by truncating the series in (3) after the term that is proportional to d^2 ; unfortunately, such a model is unstable and its solutions in a boundary-value problem may lack uniqueness [Askes et al. 2002]. Although stability and uniqueness can be restored by incorporating the next term, i.e., truncating after the d^4 term, the numerical implementation of such a model is complicated [Askes et al. 2002]; thus, alternative solution strategies are explored here.

3. Stabilising the continuum equations

Unstable gradient theories can be turned into stable gradient theories by means of Padé approximations, as has been explored in [Andrianov et al. 2003; Andrianov and Awrejcewicz 2008; Andrianov et al. 2010]. However, there are various ways to do this, and the format of the resulting equations depends on which equations are transformed by the Padé approximation.

3.1. *Irreducible form.* Firstly, (3) is truncated after the first nonstandard term. The various spatial derivatives are factorised as

$$\rho \ddot{u} = \left(1 + \frac{1}{12}d^2\frac{\partial^2}{\partial x^2}\right)E\frac{\partial^2 u}{\partial x^2}.$$
(4)

A [0, 1]-Padé approximation is used according to

$$1 + a \approx \frac{1}{1 - a} \quad \text{for } a \ll 1. \tag{5}$$

For *a* in (5), we will substitute the operator $\frac{1}{12}d^2\partial^2/\partial x^2$, which allows us to rewrite (4) as

$$\left(1 - \frac{1}{12}d^2\frac{\partial^2}{\partial x^2}\right)\rho\ddot{u} = E\frac{\partial^2 u}{\partial x^2}.$$
(6)

The higher-order gradient term now appears on the inertia side of the equation, and for this reason, it has been called microinertia, internal inertia or higher-order inertia in the literature [Vardoulakis and Aifantis 1994; Wang and Sun 2002; Bennett et al. 2007]. Equation (6), or slight variations thereof, has also been obtained by various other researchers using asymptotic series equivalence; see for instance the work of Rubin et al. [1995], Chen and Fish [2001] and Pichugin et al. [2008].

Note that the only unknown appearing in (6) is the displacement; for this reason, this format is denoted as *irreducible*. Although at first sight it may appear that the micromechanical background of the higher-order terms is lost through the Padé approximation, an alternative interpretation of the microinertia contribution in terms of long-range interactions has been provided in [Askes and Gitman 2014].

3.2. *Reducible form.* It is also possible to extract a (one-dimensional) relation between stress σ and strain ε from (3) such that

$$\rho \ddot{u} = \frac{\partial \sigma}{\partial x} \quad \text{and} \quad \varepsilon = \frac{\partial u}{\partial x}.$$
(7)

The stress-strain relation then follows as

$$\sigma = E\left(\varepsilon + \frac{1}{12}d^2\frac{\partial^2\varepsilon}{\partial x^2}\right) = E\left(1 + \frac{1}{12}d^2\frac{\partial^2}{\partial x^2}\right)\varepsilon\tag{8}$$

where series have again been truncated after the first nonstandard term. Applying the [0, 1]-Padé approximation to (8) yields

$$\left(1 - \frac{1}{12}d^2\frac{\partial^2}{\partial x^2}\right)\sigma = E\varepsilon.$$
(9)

Equations (7) and (9) can be combined into a system of coupled equations,

$$\rho \ddot{u} = \frac{\partial \sigma}{\partial x} \tag{10a}$$

together with

$$\sigma - \frac{1}{12}d^2\frac{\partial^2\sigma}{\partial x^2} = E\frac{\partial u}{\partial x}$$
(10b)

where the unknowns are the displacement u as well as the stress σ . In contrast to the single fourth-order equation (6), (10) is a set of two second-order equations. They are termed *reducible* because it is possible to eliminate one of the unknowns, namely the stress σ . To do this, the second-order spatial derivative of (10a) must be taken and, multiplied with $\frac{1}{12}d^2$, subtracted from the original expression (10a):

$$\rho\left(\ddot{u} - \frac{1}{12}d^2\frac{\partial^2\ddot{u}}{\partial x^2}\right) = \frac{\partial}{\partial x}\left(\sigma - \frac{1}{12}d^2\frac{\partial^2\sigma}{\partial x^2}\right).$$
(11)

If (10b) is substituted into the right-hand side of (11), the stress will disappear

from the expressions and thus it is possible to retrieve (6). This reduction of the number of unknowns, and its consequences, will be discussed in more depth below in Section 4.2.

4. Energy functionals for the multidimensional case

Above, the governing equations have been derived from simple mechanical and mathematical arguments in a one-dimensional context. Next, we will show how the analogous multidimensional equations can be derived from variational principles. Hamilton's action S is defined as

$$S = \int_{t_0}^{t_1} L \,\mathrm{d}t.$$
 (12)

The governing equations of the models can be derived by requiring stationarity of S, that is, $\delta S = 0$. The energy functional (or Lagrangian function) L is defined individually for the two different models below, but we will assume that L depends on the displacements u_i and their spatial and temporal derivatives, as well as on the stresses σ_{ij} and their spatial derivatives:

$$L = L(u_i; u_{i,j}; \dot{u}_i; \dot{u}_{i,j}; \sigma_{ij}; \sigma_{ij,k}).$$
(13)

Substituting (13) into (12) and requiring $\delta S = 0$ yields

$$\int_{t_0}^{t_1} \delta L \, dt = \int_{t_0}^{t_1} \left(\delta u_i \frac{\partial L}{\partial u_i} + \delta u_{i,j} \frac{\partial L}{\partial u_{i,j}} + \delta \dot{u}_i \frac{\partial L}{\partial \dot{u}_i} + \delta \dot{u}_{i,j} \frac{\partial L}{\partial \dot{u}_{i,j}} + \delta \sigma_{ij} \frac{\partial L}{\partial \sigma_{ij}} + \delta \sigma_{ij,k} \frac{\partial L}{\partial \sigma_{ij,k}} \right) dt = 0,$$
(14)

which, as usual, can be rewritten as

$$\int_{t_0}^{t_1} \delta u_i \left(\frac{\partial L}{\partial u_i} - \frac{\partial}{\partial x_j} \frac{\partial L}{\partial u_{i,j}} - \frac{\partial}{\partial t} \frac{\partial L}{\partial \dot{u}_i} + \frac{\partial^2}{\partial x_j \partial t} \frac{\partial L}{\partial \dot{u}_{i,j}} \right) dt + \int_{t_0}^{t_1} \frac{\partial}{\partial x_j} \left(\delta u_i \frac{\partial L}{\partial u_{i,j}} - \delta u_i \frac{\partial}{\partial t} \frac{\partial L}{\partial \dot{u}_{i,j}} \right) dt + \int_{t_0}^{t_1} \frac{\partial}{\partial t} \left(\delta u_i \frac{\partial L}{\partial \dot{u}_i} + \delta u_{i,j} \frac{\partial L}{\partial \dot{u}_{i,j}} \right) dt + \int_{t_0}^{t_1} \delta \sigma_{ij} \left(\frac{\partial L}{\partial \sigma_{ij}} - \frac{\partial}{\partial x_k} \frac{\partial L}{\partial \sigma_{ij,k}} \right) dt + \int_{t_0}^{t_1} \frac{\partial}{\partial x_k} \left(\delta \sigma_{ij} \frac{\partial L}{\partial \sigma_{ij,k}} \right) dt = 0.$$
(15)

The third integral cancels through the requirement that $\delta u_i = 0$ and $\delta u_{i,j} = 0$ for $t = t_0$ and for $t = t_1$. The first and fourth integrals will lead to field equations, whereas the second and fifth will contribute to the natural boundary conditions.

4.1. *Irreducible form.* The Lagrangian function of the irreducible form can be written as

$$L_{\text{irred}} = \int_{\Omega} \frac{1}{2} \rho(\dot{u}_i \dot{u}_i + \ell^2 \dot{u}_{i,j} \dot{u}_{i,j}) \, \mathrm{d}V - \int_{\Omega} \frac{1}{2} u_{i,j} C_{ijkl} u_{k,l} \, \mathrm{d}V + \int_{\Gamma_n} u_i t_i \, \mathrm{d}S \quad (16)$$

where the first integral is the kinetic energy, the second integral is the stored strain energy and the last two terms represent the work of the external forces. Thus, for this model, the Lagrangian takes the usual format of "kinetic energy minus potential energy", whereby the nonstandard contributions are included in the kinetic energy only [Lazar and Anastassiadis 2007; Polizzotto 2012]. Note that for the internal length scale we have now used the generic notation ℓ rather than the notation dthat was used in the previous section in relation to the discrete model.

Substituting (16) into (15) and noting that $\delta u_i = 0$ on Γ_e leads to

$$\int_{t_0}^{t_1} \int_{\Omega} \delta u_i (b_i + C_{ijkl} u_{k,jl} - \rho \ddot{u}_i + \rho \ell^2 \ddot{u}_{i,jj}) \, \mathrm{d}V \, \mathrm{d}t + \int_{t_0}^{t_1} \int_{\Gamma_n} \delta u_i (t_i - n_j (C_{ijkl} u_{k,l} + \rho \ell^2 \ddot{u}_{i,j})) \, \mathrm{d}S \, \mathrm{d}t = 0, \quad (17)$$

where, as usual, the boundary Γ of the domain Ω is decomposed into parts Γ_n and Γ_e associated with natural and essential boundary conditions: $\Gamma = \Gamma_n \cup \Gamma_e$ and $\emptyset = \Gamma_n \cap \Gamma_e$.

A symmetric Hookean stress $\tau_{ij}^{\text{H}} = C_{ijkl}u_{k,l}$ can be identified in terms of which the field equations and natural boundary conditions can be written as

$$\rho(\ddot{u}_i - \ell^2 \ddot{u}_{i,jj}) = \tau^{\mathrm{H}}_{ij,j} + b_i \quad \text{in } \Omega,$$
(18a)

$$n_j(\tau_{ij}^{\mathrm{H}} + \rho \ell^2 \ddot{u}_{i,j}) = t_i \qquad \text{on } \Gamma_n.$$
(18b)

In our opinion, Hookean stress is appropriate terminology for τ_{ij}^{H} , not Cauchy stress, since the equations of motion and the natural boundary conditions contain additional gradients of the acceleration that are not included in the definition of τ_{ij}^{H} . In Appendix A this particular terminology is motivated.

Remark. A nonsymmetric stress tensor τ_{ij}^{B} can be identified as (see [Lazar and Anastassiadis 2007])

$$\tau_{ij}^{\mathrm{B}} = C_{ijkl} u_{k,l} + \rho \ell^2 \ddot{u}_{i,j} \tag{19}$$

This would enable one to write the equations of motion and natural boundary conditions in terms of a stress tensor that is similar in role to a standard Cauchy stress as explained in Appendix A. However, since τ_{ij}^{B} is nonsymmetric, using the term Cauchy stress for this tensor is not obvious. This issue of nomenclature is left for future debate and discussion. **4.2.** *Reducible form.* For the reducible form, the Lagrangian function adopts a less common appearance, which, to the authors' best knowledge, is novel:

$$L_{\text{red}} = \int_{\Omega} \frac{1}{2} \rho \dot{u}_i \dot{u}_i \, dV - \int_{\Omega} u_{i,j} \sigma_{ij} \, dV + \int_{\Omega} \frac{1}{2} (\sigma_{ij} S_{ijkl} \sigma_{kl} + \ell^2 \sigma_{ij,m} S_{ijkl} \sigma_{kl,m}) \, dV + \int_{\Omega} u_i b_i \, dV + \int_{\Gamma_n} u_i t_i \, dS, \quad (20)$$

where S_{ijkl} is the elastic compliance tensor. The first integral is again the kinetic energy, whilst the last two integrals contain the external work. The third integral contains the stored complementary energy with a positive rather than negative sign, but the effects of the lower-order part are offset by the effects of the second integral, which couples the effects of the two sets of unknowns, namely displacements and stresses. In the reducible form, the displacement derivative $u_{i,j}$ is no longer energyconjugated to the (symmetric) stress σ_{ij} , unless $\ell = 0$. Therefore, the second integrand does *not* have the meaning of internal work. Expression (20) can also be rewritten as a Hellinger–Reissner functional whereby the displacements act as Lagrange multipliers to enforce balance of momentum in Ω and on Γ [Askes and Gutiérrez 2006; Polizzotto 2015].

Again making use of $\delta u_i = 0$ on Γ_e , substitution of (20) into (15) yields

$$\int_{t_0}^{t_1} \int_{\Omega} \delta u_i (b_i + \sigma_{ij,j} - \rho \ddot{u}_i) \, \mathrm{d}V \, \mathrm{d}t + \int_{t_0}^{t_1} \int_{\Gamma_n} \delta u_i (t_i - n_j \sigma_{ij}) \, \mathrm{d}S \, \mathrm{d}t + \int_{t_0}^{t_1} \int_{\Omega} \delta \sigma_{ij} (-u_{i,j} + S_{ijkl} \sigma_{kl} - \ell^2 S_{ijkl} \sigma_{kl,mm}) \, \mathrm{d}V \, \mathrm{d}t + \int_{t_0}^{t_1} \oint_{\Gamma} \delta \sigma_{ij} n_m S_{ijkl} \sigma_{kl,m} \, \mathrm{d}S \, \mathrm{d}t = 0 \quad (21)$$

so that the following set of coupled governing equations can be identified:

$$\rho \ddot{u}_i = \sigma_{ij,j} + b_i \qquad \qquad \text{in } \Omega, \qquad (22a)$$

$$n_j \sigma_{ij} = t_i$$
 on Γ_n , (22b)

$$S_{ijkl}(\sigma_{kl} - \ell^2 \sigma_{kl,mm}) = \frac{1}{2}(u_{i,j} + u_{j,i})$$
 in Ω , (22c)

$$n_m \ell^2 S_{ijkl} \sigma_{kl,m} = 0 \qquad \text{on } \Gamma. \tag{22d}$$

From the format of (22a) and (22b), it is clear that the meaning of σ_{ij} in the reducible model is that of the Cauchy stress. Equations (22) have also been derived, using different arguments, by Eringen [1983]; see Appendix B for a discussion.

Equations (22) form a set of coupled equations with independent unknowns u_i and σ_{ij} , but they are reducible in the sense that it is possible to eliminate the stresses σ_{ij} . To do so, firstly the Laplacian of (22a) is taken and multiplied with ℓ^2 , after which the result is subtracted from the original expression (22a). This gives

$$\rho(\ddot{u}_{i} - \ell^{2} \ddot{u}_{i,jj}) = \sigma_{ij,j} - \ell^{2} \sigma_{ij,kk} + b_{i} - \ell^{2} b_{i,jj}.$$
(23)

Next, (22c) is premultiplied with the elastic stiffness tensor C_{ijkl} and substituted into (23), leading to

$$\rho(\ddot{u}_i - \ell^2 \ddot{u}_{i,jj}) = C_{ijkl} u_{k,jl} + b_i - \ell^2 b_{i,jj}, \qquad (24)$$

which is equivalent to (18a) except for the presence of the Laplacian of the body forces $b_{i,jj}$ and a mismatch in the associated variationally consistent boundary conditions. Note that the effect of the higher-order gradients disappears altogether in statics in the case $b_{i,jj} = 0$.

Remark. From (22c) it is clear that the gradient enrichment affects the constitutive part of the field equations, and therefore the term "gradient elasticity" seems appropriate for what is here denoted as the reducible form. In contrast, it could be argued that using the term "gradient elasticity" is less suitable for the irreducible format represented in (24), because the gradient enrichment operates on the accelerations, not stresses or strains—i.e., the elasticity part of the irreducible form retains its classical format. However, we still prefer to refer to the irreducible form as a particular variant of gradient elasticity, because of the close relation between the reducible form and the irreducible form. Due to the coupling between the equations of motion and the constitutive equations, the gradient enrichment of the accelerations will affect the stresses and strains, albeit indirectly.

5. Finite element equations

In order to obtain solutions of the relevant partial differential equations for domains of arbitrary geometry, a numerical solution strategy is required. Here, the finite element method will be used for the spatial discretisation, whereas the Newmark time integrator will be adopted to progress the solution in the time domain. The finite element equations of the irreducible form are well established and need not be revisited here — the interested reader is referred to [Fish et al. 2002a; 2002b; Askes and Aifantis 2011].

For the reducible form, we write $\underline{u} = N_u d$ and $\underline{\sigma} = N_\sigma s$ where \underline{u} and $\underline{\sigma}$ are column vectors containing the relevant components of the displacements and Cauchy stresses, respectively. Furthermore, the matrices N_u and N_σ contain the shape functions for displacements and Cauchy stresses whereas d and s are the nodal displacements and nodal Cauchy stresses. The spatial discretisation of (20) can thus be written as

$$L_{\text{red}}^{\text{FE}} = \int_{\Omega} \frac{1}{2} \rho \dot{d}^T N_u^T N_u \dot{d} \, \mathrm{d}V - \int_{\Omega} d^T B_u^T N_\sigma s \, \mathrm{d}V + \int_{\Omega} \frac{1}{2} s^T \left(N_\sigma^T S N_\sigma + \sum_{i=1}^3 \ell^2 \frac{\partial N_\sigma^T}{\partial x_i} S \frac{\partial N_\sigma}{\partial x_i} \right) s \, \mathrm{d}V + \int_{\Omega} d^T N_u^T \underline{b} \, \mathrm{d}V + \int_{\Gamma_n} d^T N_u^T \underline{t} \, \mathrm{d}S \quad (25)$$

where \underline{b} and \underline{t} contain the components of the distributed body and surface forces, respectively. Furthermore, B_u is the standard strain-displacement matrix with derivatives of the displacement shape functions N_u and S is the matrix counterpart of the compliance tensor S_{ijkl} .

Requiring $\delta L_{\rm red}^{\rm FE} = 0$ leads to a system of finite element equations according to

$$\begin{bmatrix} M_{uu} & \mathbf{0} \\ \mathbf{0} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{d} \\ \ddot{s} \end{bmatrix} + \begin{bmatrix} \mathbf{0} & K_{u\sigma} \\ K_{\sigma u} & K_{\sigma\sigma} \end{bmatrix} \begin{bmatrix} d \\ s \end{bmatrix} = \begin{bmatrix} f \\ \mathbf{0} \end{bmatrix},$$
(26)

where

$$M_{uu} = \int_{\Omega} \rho N_u^T N_u \, \mathrm{d}V, \tag{27a}$$

$$\boldsymbol{K}_{u\sigma} = \boldsymbol{K}_{\sigma u}^{T} = \int_{\Omega} \boldsymbol{B}_{u}^{T} \boldsymbol{N}_{\sigma} \, \mathrm{d}\boldsymbol{V}, \qquad (27b)$$

$$\boldsymbol{K}_{\sigma\sigma} = -\int_{\Omega} \left(\boldsymbol{N}_{\sigma}^{T} \boldsymbol{S} \boldsymbol{N}_{\sigma} + \sum_{i=1}^{3} \ell^{2} \frac{\partial \boldsymbol{N}_{\sigma}^{T}}{\partial x_{i}} \boldsymbol{S} \frac{\partial \boldsymbol{N}_{\sigma}}{\partial x_{i}} \right) \mathrm{d}V.$$
(27c)

Finite-element implementation of (26) was carried out using the recommendations of the statics theory given in [Askes and Gutiérrez 2006], in particular the use of quadratic shape functions for s and linear shape functions for d. This particular choice of shape functions avoids oscillations in the displacement field, although a formal investigation of the inf-sup condition may require further refinement of the two sets of interpolations.

6. Numerical example

Although the reducible form can be transformed into the irreducible form as shown in (23) and (24), the associated change in variationally consistent boundary conditions has implications when it comes to the simulation of crack tip stresses. This will be demonstrated by means of the numerical example shown in Figure 2.



Figure 2. Strip with central crack: geometry and loading conditions.

A square strip with dimension 2L = 2 m has a central crack of length 2a = 0.5 m. The material properties are mass density $\rho = 1 \text{ kg/m}^3$, Young's modulus $E = 100 \text{ N/m}^2$ and Poisson's ratio $\nu = \frac{1}{4}$, whilst a plane stress assumption has been made. Furthermore, the gradient elasticity length scale $\ell = 0.1$ m. The strip is subjected to outward vertical velocities $\dot{\bar{u}} = 10 \text{ m/s}$ imposed on the top and bottom edges, as indicated, which leads to stress waves propagating towards the centre of the strip. Away from the crack, the stress waves will have the shape of a block wave due to the nature of the loading conditions, but the presence of the crack will disturb this pattern, and indeed in a classical elasticity setting, this will lead to singular stresses and strains at the tips of the crack. It is the aim of this example to verify whether these singularities can be avoided in the reducible and irreducible formulations of gradient elasticity discussed above. For reasons of symmetry, only the top quarter of the strip is modelled.

The irreducible format of gradient elasticity is implemented with four-noded quadrilateral elements for the displacements. The reducible format is implemented with eight-noded elements for the stresses and four-noded quadrilateral elements for the displacements — see [Askes and Gutiérrez 2006] for details on this particular choice. Structured finite element meshes consisting of square elements are used, and a sequence of uniformly refined meshes is taken to monitor the behaviour of the stresses at the crack tip. Since in the irreducible format the stresses are postprocessed from linear displacements whereas in the reducible format the stresses are primary unknowns interpolated with quadratic shape functions, there is an obvious mismatch in stress resolution between the two formats. To address this mismatch, the meshes used range from 16×16 to 128×128 elements for the irreducible format.

Regarding the imposition of traction boundary conditions, it must be realised that the stresses are primary variables in the reducible formulation, whereas they are derived quantities in the irreducible formulation. In the reducible formulation, traction boundary conditions are thus essential boundary conditions and are imposed by assigning prescribed values to the relevant stress components (e.g., $\sigma_{yy} = 0$ on the crack face). On the other hand, traction boundary conditions are natural boundary conditions in the irreducible formulation; applying zero tractions on the crack face means that the left-hand-side of (18b) is set equal to zero, which is handled straightforwardly in a finite element context. Finally, and for the sake of completeness, it is noted that displacement (and velocity) boundary conditions have been implemented using Lagrange Multipliers in the reducible formulation.

The Newmark constant average acceleration scheme is used for the time integration. This scheme is unconditionally stable; therefore, the only criterion for selecting the time step is accuracy. Following the recommendations given in [Askes et al. 2008; Bennett and Askes 2009], the time step is chosen such that waves



Figure 3. Vertical normal stress τ_{yy}^{H} (N/m²) versus *x* (m) for the irreducible format — 16 × 16 elements (dotted), 32 × 32 elements (dashed), 64 × 64 elements (dot-dashed) and 128 × 128 elements (solid).

propagate approximately half an element per time step. Time domain simulations were carried out from time t = 0 s to t = 0.2 s.

Figures 3 and 4 show the profiles of the vertical normal stress for both formats and the indicated range of finite element meshes, where the origin of the coordinate system is chosen at the centre of the crack. For the irreducible format (Figure 3), we have plotted the Hookean stress τ_{yy}^{H} (see Section 4.1) whilst for the reducible format the Cauchy stress σ_{yy} is plotted (Figure 4).

The stress profiles for the irreducible formulation appear to converge towards a unique solution, except for the crack tip value. At the crack tip, the stress increases significantly for every refinement of the mesh. This is an indication that a stress singularity is present at the crack tip. To analyse this in more depth, Richardson extrapolations have been carried out for the crack tip stresses. Table 1 reports the

mesh	$ au_{yy}^{ m H}$	extrapolation
16 × 16	4.8077	
32×32	6.8091	8.8105
64×64	9.6223	13.6438
128×128	13.5923	20.0751

Table 1. Crack tip stress and Richardson extrapolation in N/m^2 for irreducible form.



Figure 4. Vertical normal stress σ_{yy} (N/m²) versus *x* (m) for the reducible format — 8 × 8 elements (dotted), 16 × 16 elements (dashed), 32 × 32 elements (dot-dashed) and 64 × 64 elements (solid).

values of the crack tip stress and their extrapolations. (The first extrapolation is a two-point extrapolation based on the coarsest two meshes, the second is a three-point extrapolation based on the coarsest three meshes, and mutatis mutandis for the last extrapolation.) The numerical results confirm that the crack tip stress grows in a seemingly unbounded manner, whereas the difference between numerical stress and extrapolated stress increases with refinement of the mesh. This confirms the suggestion that a singularity is present. Thus, it must be concluded that the irreducible format is not capable of avoiding stress singularities. This is reported for the Hookean stress $\tau_{yy}^{\rm H}$ but will carry over to the pseudo Cauchy stress $\tau_{yy}^{\rm B}$ since the latter quantity includes the former.

On the other hand, the results of the reducible format clearly converge towards a unique, nonsingular solution, and the singularities that plague classical elasticity formulations are avoided. However, it must be noted that the maximum stress occurs not at the crack tip but further inside the material. This is in line with the analysis and results reported in [Simone et al. 2004].

7. Conclusions

We have reviewed and systematically compared two formats of gradient elasticity. Both formats can be derived by continualising a one-dimensional discrete model and stabilising the resulting equations, but the models differ in respect of which particular equation is stabilised — either the field equation (leading to what is denoted as the "irreducible format") or the constitutive equation (leading to the "reducible format"). The multidimensional case, including the associated boundary conditions, has been derived from a variational principle. It is noted that the field equations of the irreducible format can be retrieved from those of the reducible format (assuming that the Laplacian of the body forces vanishes), but the variationally consistent boundary conditions are different for the two models.

This has implications for the solution of initial-boundary-value problems. We have presented a crack problem, and it was demonstrated that the irreducible format is not capable of avoiding singularities in the stress field. On the other hand, no singularities were found when the reducible format was used. Thus, for the dynamic analysis of stresses around sharp cracks, the reducible format is to be preferred.

Appendix A: Nomenclature in gradient elasticity: Cauchy stress

In the literature, there is a lack of consistency in which quantity is denoted as the Cauchy stress in gradient elasticity theories. Some eminent authors have used this term to indicate the derivative of the strain energy density with respect to the strain — see for instance [Mindlin 1964, p. 57] or [Shu et al. 1999, p. 375]. However, we have followed the arguments set out by Borino and Polizzotto [2003, Remark 3], who state that the term Cauchy stress should be used for the total stress quantity as it appears in the equilibrium equations; conversely, we have used the term Hookean stress for the derivative of the strain energy density with respect to the strain. We believe the former is in line with the conceptualisation of Cauchy himself, who discussed stresses as forming equilibrium (or indeed accelerating) systems by acting on surfaces, rather than as derivatives of energy functionals — see for instance [Cauchy 1823; 1827; 1843].

However, it is also noted that extending the concept of Cauchy stress as "force divided by area" to gradient-enriched continua leads, in general, to much more complicated expressions. This is illustrated by the format of the natural boundary conditions in Mindlin's [1964, pp. 67–68] theory of gradient elasticity. Askes and Metrikine [2005] as well as Froiio et al. [2010] have provided physical interpretations of the nonstandard boundary conditions.

Appendix B: Eringen's 1983 differential theory of nonlocal elasticity

The reducible format presented in Section 4 has been derived earlier in [Eringen 1983] from an integral formulation. Because the coupled nature of the governing equations of Eringen's theory is not always appreciated, it is worthwhile to summarise Eringen's theory. Adopting his notation unless stated otherwise, the equations of motion are given by [Eringen 1983, (2.1)] as

$$t_{kl,k} + \rho(f_l - \ddot{u}_l) = 0 \tag{28}$$

where t_{kl} is the Cauchy stress tensor and f_l is the body force density. With the restriction to isotropic linear elasticity, a Hookean stress σ_{kl}^0 is defined via [Eringen 1983, (2.3) and (2.4)] as

$$\sigma_{kl}^0 = \lambda \delta_{kl} u_{j,j} + \mu u_{k,l} + \mu u_{l,k} \tag{29}$$

where a superscript 0 is included in σ^0 to avoid confusion with the Cauchy stress of the reducible theory discussed in Section 4.2. Furthermore, λ and μ are the Lamé constants and δ_{kl} is the Kronecker delta.

The field equations are completed by a differential relation between the Cauchy stress t_{kl} and the Hookean stress σ_{kl}^0 . The particular relation that seems to have attracted most interest in the literature is given in [Eringen 1983, (3.19)] as

$$t_{kl} - \ell^2 t_{kl,jj} = \sigma_{kl}^0 \tag{30}$$

where the higher-order coefficient is simply indicated by ℓ^2 (Eringen uses a more intricate notation with multiple symbols, which are not required in the present discussion).

Eringen [1983, pp. 4704–4705] also discusses the elimination of the stress t_{kl} from the system of equations. Combining (3.13) and (3.18), he arrives at the irreducible form

$$\sigma_{kl,k}^{0} + (1 - \ell^2 \nabla^2)(\rho f_l - \rho \ddot{u}_l) = 0.$$
(31)

Next, he notes that the particular case of statics with vanishing body forces leads to

$$\sigma_{kl,k}^0 = 0. \tag{32}$$

However, regarding natural boundary conditions, Eringen [1983, p. 4704] explicitly states that "[b]oundary conditions involving tractions [are] based on the stress tensor t_{kl} , not on σ_{kl}^0 ", while Eringen [2002, p. 100] also emphasises that "the real stress is not σ_{kl}^0 but t_{kl} " — in both quotations we have added the superscript 0 to σ as explained above. This means that (32) cannot be used in isolation to solve general boundary-value problems involving prescribed tractions.

In summary, in our opinion, a divergence-free Hookean stress σ^0 should not be considered as a fundamental equation of the Eringen theory because, firstly, it can only be retrieved by making the assumptions of zero body force and zero acceleration and, secondly, it cannot be used to solve general equilibrium problems due to a lack of associated traction boundary conditions. In this respect, we disagree with Lazar and Polyzos [2015], who suggest that (32) is an equilibrium equation in its own right — although these authors do confirm that the correct natural boundary conditions are in terms of t_{kl} rather than σ_{kl}^0 .

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DATING HYPATIA'S BIRTH: A PROBABILISTIC MODEL

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We propose a probabilistic approach as a dating methodology for events like the birth of a historical figure. The method is then applied to the controversial birth date of the Alexandrian scientist Hypatia, proving to be surprisingly effective.

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1. Introduction

Although in historical investigation it may appear meaningless to do experiments on the basis of a preexisting theory - and in particular, it does not make sense to prove theorems of history - it can make perfect sense to use forms of reasoning typical of the exact sciences as an aid to increase the degree of reliability of a particular statement regarding a historical event. This paper deals with the problem of dating the birth of a historical figure when the only information available about it is indirect — for example, a set of testimonies, or scattered statements, about various aspects of his/her life. The strategy is then based on the construction of a probability distribution for the birth date out of each testimony and subsequently combining the distributions so obtained in a sensible way. One might raise several objections to this program. According to Charles Sanders Peirce [1901], a probability "is the known ratio of frequency of a specific event to a generic event", but a birth is neither a specific event nor a generic event but an "individual event". Nevertheless, probabilistic reasoning is used quite often in situations dealing with events that can be classified as "individual". In probabilistic forecasting, one tries to summarize what is known about future events with the assignment of a probability to each of a number of different outcomes that are often events of this kind. For instance, in sport betting, a summary of bettors' opinions about the likely outcome of a race

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is produced in order to set bookmakers' pay-off rates. By the way, this type of observation lies at the basis of the theoretical formulation of the subjective approach in probability theory [de Finetti 1931]. Although we do not endorse de Finetti's approach in all its implications, we embrace its severe criticism of the exclusive use of the frequentist interpretation in the application of probability theory to concrete problems. In particular, we feel entitled to look at an "individual" event of the historical past with a spirit similar to that with which one bets on a future outcome (this is a well known issue in the philosophy of probability; see, e.g., [Dubucs 1993]). Plainly, as the information about an event like the birth of an historical figure is first extracted by material drawn from various literary sources and then treated with mathematical tools, both our approach and goal are interdisciplinary in their essence.

2. A probabilistic method for combining testimonies

Let $X = [x_-, x_+] \subset \mathbb{Z}$ be the time interval that includes all possible birth dates of a given subject (*terminus ad quem*). X can be regarded as a set of mutually exclusive statements about a singular phenomenon (the birth of a given subject in a given year), only one of which is true, and can be made a probability space (X, \mathcal{F}, P_0) , with \mathcal{F} the σ -algebra made of the $2^{|X|}$ events of interest and P_0 the uniform probability measure on \mathcal{F} (reference measure): $P_0(A) = |A|/|X|$ (where |A| denotes the number of elements of A). In the context of decision theory, the assignment of this probability space can be regarded as the expression of a basic state of knowledge, in the absence of any information that can be used to discriminate among the possible statements on the given phenomenon, namely a situation in which Laplace's *principle of indifference* can be legitimately applied.

Now suppose we have k testimonies T_i , i = 1, ..., k, which in first approximation we may assume independent of each other, each providing some kind of information about the life of the subject, and which can be translated into a probability distribution p_i on \mathcal{F} so that $p_i(x)$ is the probability that the subject is born in the year $x \in X$ based on the information given by the testimony T_i , assumed true, along with supplementary information such as, e.g., life tables for the historical period considered. The precise criteria for the construction of these probability distributions depends on the kind of information carried by each testimony and will be discussed case by case in the next section. Of course, we shall also take into account the possibility that some testimonies are false, thereby not producing any additional information. We model this possibility by assuming that the corresponding distributions equal the reference measure P_0 .

The problem that we want to discuss in this section is the following: how can one combine the distributions p_i in such a way to get a single probability distribution Q

that somehow optimizes the available information? To address this question, let us observe that from the *k* testimonies taken together, each one with the possibility to be true or false, one gets $N = 2^k$ combinations, corresponding to as many binary words $\sigma_s = \sigma_s(1) \cdots \sigma_s(k) \in \{0, 1\}^k$, which can be ordered lexicographically according to $s = \sum_{i=1}^k \sigma_s(i) \cdot 2^{i-1} \in \{0, 1, \dots, N-1\}$, and given by

$$P_{s}(\cdot) = \frac{\prod_{i=1}^{k} p_{i}^{\sigma_{s}(i)}(\cdot)}{\sum_{x \in X} \prod_{i=1}^{k} p_{i}^{\sigma_{s}(i)}(x)}, \quad p_{i}^{\sigma_{s}(i)} = \begin{cases} p_{i}, & \sigma_{s}(i) = 1, \\ P_{0}, & \sigma_{s}(i) = 0. \end{cases}$$
(2-1)

In particular, one readily verifies that P_0 is but the reference uniform measure.

Now, if Ω denotes the class of probability distributions $Q: X \to [0, 1]$, we look for a *pooling operator* $T: \Omega^N \to \Omega$ that combines the distributions P_s by weighing them in a sensible way. The simplest candidate has the general form of a linear combination

$$T(P_0, \dots, P_{N-1}) = \sum_{s=0}^{N-1} w_s P_s, \quad w_s \ge 0, \ \sum_{s=0}^{N-1} w_s = 1,$$
(2-2)

which, as we shall see, can also be obtained by minimizing some informationtheoretic function.

Remark 2.1. The issue we are discussing here has been the object of a vast amount of literature regarding the normative aspects of the formation of aggregate opinions in several contexts (see, e.g., [Genest and Zidek 1986] and references therein). In particular, it has been shown by McConway [1981] that, if one requires the existence of a function $F : [0, 1]^N \rightarrow [0, 1]$ such that

$$T(P_0, \dots, P_{N-1})(A) = F(P_0(A), \dots, P_{N-1}(A))$$
 for all $A \in \mathcal{F}$ (2-3)

with $P_s(A) = \sum_{x \in A} P_s(x)$, then whenever $|X| \ge 3$, *F* must necessarily have the form of a linear combination as in (2-2). The above condition implies in particular that the value of the combined distribution on coordinates depends only on the corresponding values on the coordinates of the distributions P_s , namely that the pooling operator commutes with marginalization.

However, some drawbacks of the linear pooling operator have also been highlighted. For example, it does not "preserve independence" in general: if $|X| \ge 5$, it is not true that $P_s(A \cap B) = P_s(A)P_s(B)$, s = 0, ..., N-1, entails

$$T(P_0, \ldots, P_{N-1})(A \cap B) = T(P_0, \ldots, P_{N-1})(A)T(P_0, \ldots, P_{N-1})(B)$$

unless $w_s = 1$ for some *s* and 0 for all others [Lehrer and Wagner 1983; Genest and Wagner 1987].

(Another form of the pooling operator considered in the literature to overcome the difficulties associated with the use of (2-2) is the log-linear combination

$$T(P_0, \dots, P_{N-1}) = C \prod_{s=0}^{N-1} P_s^{w_s}, \quad w_s \ge 0, \ \sum_{s=0}^{N-1} w_s = 1,$$
(2-4)

where C is a normalizing constant [Genest and Zidek 1986; Abbas 2009].)

On the other hand, in our context, the independence preservation property does not seem so desirable: the final distribution $T(P_0, \ldots, P_{N-1})$ relies on a set of information much wider than that associated with the single distributions P_s , and one can easily imagine how the alleged independence between two events can disappear as the information about them increases.

2.1. *Optimization.* The linear combination (2-2) can also be viewed as the marginal distribution¹ of $x \in X$ under the hypothesis that one of the distributions P_0, \ldots, P_{N-1} is the "true" one (without knowing which) [Genest and McConway 1990]. In this perspective, (2-2) can be obtained by minimizing the expected loss of information due to the need to compromise, namely a function of the form

$$I(w, Q) = \sum_{s=0}^{N-1} w_s D(P_s \parallel Q) \ge 0,$$
(2-5)

where

$$D(P \parallel Q) = \sum_{x \in X} P(x) \log\left(\frac{P(x)}{Q(x)}\right)$$
(2-6)

is the *Kullback–Leibler divergence* [1951], representing the information loss using the measure Q instead of P. Note that the concavity of the logarithm and the Jensen inequality yield

$$-\sum_{x} P(x) \log \frac{P(x)}{Q(x)} \le \log \sum_{x} P(x) \frac{Q(x)}{P(x)} = 0$$

and therefore

$$D(P \parallel Q) \ge 0$$
 and $D(P \parallel Q) = 0 \iff Q \equiv P.$ (2-7)

We have the following result.

Lemma 2.2. *Given a probability vector* $w = (w_0, w_1, \ldots, w_{N-1})$,

$$\underset{Q\in\Omega}{\arg\min} I(w, Q) = Q_w \equiv \sum_s w_s P_s.$$
(2-8)

Moreover,

$$I(w, Q_w) = H\left(\sum_s w_s P_s\right) - \sum_s w_s H(P_s), \qquad (2-9)$$

where $H(Q) = -\sum_{x \in X} Q(x) \log Q(x)$ is the entropy of $Q \in \Omega$.

¹In the sense that a marginal probability can be obtained by averaging conditional probabilities.

Proof. Equation (2-8) can be obtained using the method of Lagrange multipliers. An alternative argument makes use of the easily derived "parallelogram rule":

$$\sum_{s} w_{s} D(P_{s} \parallel Q) = \sum_{s} w_{s} D(P_{s} \parallel Q_{w}) + D(Q_{w} \parallel Q) \quad \text{for all } Q \in \Omega.$$
(2-10)

From (2-7), we thus get $I(w, Q_w) \le I(w, Q)$ for all $Q \in \Omega$. The uniqueness of the minimum follows from the convexity of $D(P \parallel Q)$ with respect to Q. Finally, checking (2-9) is a simple exercise.

Remark 2.3. It is worth mentioning that, if we took $\sum_s w_s D(Q \parallel P_s)$ (instead of $\sum_s w_s D(P_s \parallel Q)$) as the function to be minimized (still varying Q with w fixed), then instead of the "arithmetic mean" (2-2), the "optimal" distribution would have been the "geometric mean" (2-4) (see also [Abbas 2009]).

2.2. Allocating the weights. We have seen that for each probability vector w in the *N*-dimensional simplex $\{w_s \ge 0 : \sum_{s=0}^{N-1} w_s = 1\}$ the distribution $Q_w = \sum_s w_s P_s$ is the "optimal" one. We are now left with the problem of determining a sensible choice for w. This cannot be achieved by using the same criterion, in that by (2-7) $\inf_w I(w, Q_w) = 0$ and the minimum is realized whenever $w_s = 1$ for some s and 0 for all others.

A suitable expression for the weights w_s can be obtained by observing that the term $\sum_{x \in X} \prod_{i=1}^{k} p_i^{\sigma_s(i)}(x)$ is proportional to the probability of the event (in the product space $X^{[1,k]}$) that the birth dates of k different subjects, with the *i*-th birth date distributed according to $p_i^{\sigma_s(i)}$, coincide, and thus, it furnishes a measure of the degree of compatibility of the distributions p_i involved in the product associated with the word σ_s .

It thus appears natural to consider the weights

$$w_{s} = \frac{\sum_{x \in X} \prod_{i=1}^{k} p_{i}^{\sigma_{s}(i)}(x)}{\sum_{s=0}^{N-1} \sum_{x \in X} \prod_{i=1}^{k} p_{i}^{\sigma_{s}(i)}(x)},$$
(2-11)

which, once inserted in (2-2), yield the expression

$$T(P_0, \dots, P_{N-1})(\cdot) = \frac{\sum_{s=0}^{N-1} \prod_{i=1}^k p_i^{\sigma_s(i)}(\cdot)}{\sum_{x \in X} \sum_{s=0}^{N-1} \prod_{i=1}^k p_i^{\sigma_s(i)}(x)}.$$
 (2-12)

Remark 2.4. There are at least k + 1 strictly positive coefficients w_s . They correspond to the words $\sigma_s^{(i)}$ with $\sigma_s^{(i)}(i) = 1$ for some $i \in \{1, ..., k\}$ and $\sigma_s^{(i)}(j) = 0$ for $j \neq i$, plus one to the word 0^k , that is, to the distributions $P_{s^{(i)}} \equiv p_i, i \in \{0, 1, ..., k\}$, where $p_0 \equiv P_0$.

2.3. Weights as likelihoods. A somewhat complementary argument to justify the choice (2-11) for the coefficients w_s can be formulated in the language of probabilistic inference, showing that they can be interpreted as (normalized) average

likelihoods associated with the various combinations corresponding to the words σ_s . More precisely, with each pair of "hypotheses" of the form

$$D_i^e = \begin{cases} \{T_i \text{ true}\}, & e = 1, \\ \{T_i \text{ false}\}, & e = 0, \end{cases}$$

we associate its *likelihood*, given the event that the birth date is $x \in X$, with the expression²

$$V(D_i^e \mid x) = \frac{P(x \mid D_i^e)}{P(x)} = \begin{cases} p_i(x)/p_0(x), & e = 1, \\ 1, & e = 0, \end{cases}$$
(2-13)

with $i \in \{1, ..., k\}$ and $p_0 \equiv P_0$. In this way, the posterior probability $P(D_i^e | x)$ (the probability of D_i^e in light of the event that the subject was born in the year $x \in X$) is given by the product of $V(D_i^e | x)$ with the prior probability $P(D_i^e)$, according to Bayes's formula.

If we now consider two pairs of "hypotheses" $D_i^{e_i}$ and $D_j^{e_j}$, which we assume conditionally independent (without being necessarily independent), that is,

$$P(D_i^{e_i}, D_j^{e_j} \mid x) = P(D_i^{e_i} \mid x)P(D_j^{e_j} \mid x), \quad e_i, e_j \in \{0, 1\},$$

then we find

$$\begin{split} P(D_i^{e_i}, D_j^{e_j} \mid x) &= \frac{P(x \mid D_i^{e_i}, D_j^{e_j})}{P(x)} = \frac{P(D_i^{e_i}, D_j^{e_j} \mid x)}{P(D_i^{e_i}, D_j^{e_j})} = \frac{P(D_i^{e_i} \mid x)P(D_j^{e_j} \mid x)}{P(D_i^{e_i}, D_j^{e_j})} \\ &= \frac{P(D_i^{e_i})P(D_j^{e_j})}{P(D_i^{e_i}, D_j^{e_j})} \cdot V(D_i^{e_i} \mid x)V(D_j^{e_j} \mid x). \end{split}$$

More generally, given k testimonies T_i , to each of which there corresponds the pair of events D_i^e , and given a word $\sigma_s \in \{0, 1\}^k$, if we assume the conditional independence of the events $(D_1^{\sigma_s(1)}, \ldots, D_k^{\sigma_s(k)})$, we get

$$V(D_1^{\sigma_s(1)}, \dots, D_k^{\sigma_s(k)} \mid x) = \rho_s \prod_{i=1}^k V(D_i^{\sigma_s(i)} \mid x)$$
(2-14)

where

$$\rho_{s} = \frac{\prod_{i=1}^{k} P(D_{i}^{\sigma_{s}(i)})}{P(D_{1}^{\sigma_{s}(1)}, \dots, D_{k}^{\sigma_{s}(k)})}.$$
(2-15)

If, in addition, there is grounds to assume unconditional independence, i.e., $\rho_s = 1$, then (2-14) simply reduces to the product rule. Under this assumption, we can

²Here the symbol *P* denotes either the reference measure P_0 or any probability measure on *X* compatible with it.

evaluate the *average likelihood* of the set of information $(D_1^{\sigma_s(1)}, \ldots, D_k^{\sigma_s(k)})$ with the expression

$$V_s = \frac{1}{|X|} \sum_{x \in X} V(D_1^{\sigma_s(1)}, \dots, D_k^{\sigma_s(k)} \mid x) = |X|^{k-1} \sum_{x \in X} \prod_{i=1}^k p_i^{\sigma_s(i)}(x).$$
(2-16)

Comparing with (2-11), we see that

$$w_{s} = \frac{V_{s}}{\sum_{s=0}^{N-1} V_{s}}.$$
(2-17)

In other words, within the hypotheses made so far, the allocation of the coefficients (2-11) corresponds to assigning to each distribution P_s a weight proportional to the average likelihood of the set of information from which it is constructed.

3. Application to Hypatia

This method is now applied to a particular dating process, the one of Hypatia's birth. This choice stems from the desire to study a case both easy to handle and potentially useful in its results. The problem of dating Hypatia's birth is indeed open, in that there are different possible resolutions of the constraints imposed by the available data. According to the reconstruction given by Deakin [2007, p. 51], "Hypatia's birth has been placed as early as 350 and as late as 375. Most authors settle for 'around 370'". There are not many testimonies (historical records) concerning the birth of the Alexandrian scientist (far more are about her infamous death), but they have the desirable feature of being independent of one another, as will be apparent in the sequel, so that the scheme discussed in the previous section can be directly applied. The hope is to obtain something that is qualitatively significant when compared to the preexisting proposals, based on a qualitative discussion of the sources, and quantitatively unambiguous. A probability distribution for the year of Hypatia's birth is extracted from each testimony, the specific reasoning being briefly discussed in each case. Eventually all distributions are combined according to the criteria outlined in the previous section.

3.1. Hypatia was at her peak between 395 and 408. Under the entry Υπατία, the Suda (a Byzantine lexicon) informs us that she flourished under the emperor Arcadius (ἤχμασεν ἐπί τῆς βασιλείας Ἀρχαδίου).³

It is well established that Arcadius, the first ruler of the Byzantine Empire, reigned from 395 to 408. Guessing an age or age interval based on the Greek $\ddot{\eta}$ ×µ $\alpha\sigma\varepsilon\nu$, however, is less straightforward. The word is related to $\dot{\alpha}$ ×µ $\dot{\eta}$, 'peak',

 $^{^{3}\}Upsilon$ 166. See http://www.stoa.org/sol-bin/search.pl?field=adlerhw_gr&searchstr=upsilon,166.



Figure 1. The probability distribution f(x) assumed associated with one's peak years.



Figure 2. The probability distribution $\Upsilon_f(\xi)$ for Hypatia's birth based on her peak years.

and we follow the rule of thumb, going back to Antiquity, that it refers to the period of one's life around 40 years of age. Specifically, we adopt the probability distribution f(x) in Figure 1 to model how old Hypatia would have been at her "peak" in Arcadius' reign.

Figure 2 shows $\Upsilon_f(\xi)$, the probability distribution for the year of Hypatia's birth deduced from this historical datum; it is obtained by averaging fourteen copies of the triangular f(x), each centered around one of the years from 355 through 368—the beginning and end points of Arcadius's empire, shifted back by the 40 years corresponding to the peak of f(x).

3.2. *Hypatia was intellectually active in 415.* The sources ascribe Hypatia's martyrdom at the hands of a mob of Christian fanatics to the envy that many felt on account of her extraordinary intelligence, freedom of thought, and political influence, being a woman. Her entry in the Suda, already mentioned, states:

Τοῦτο δὲ πέπονθε διὰ φθόνον καὶ τὴν ὑπερβάλλουσαν σοφίαν, καὶ μάλιστα εἰς τὰ περὶ ἀστρονομίαν.⁴

Socrates Scholasticus, in his Εχκλησιαστική Ιστορία, reports:

On account of the self-possession and ease of manner, which she had acquired in consequence of the cultivation of her mind, she not infrequently appeared in public in presence of the magistrates. Neither did she feel abashed in coming to an assembly of men. For all men on account of her extraordinary dignity and virtue admired her the more. Yet even she fell a victim to the political jealousy which at that time prevailed. For as she had frequent interviews with Orestes, it was calumniously reported among the Christian populace, that it was she who prevented Orestes from being reconciled to the bishop.⁵

Because of these and similar testimonies, it seems reasonable to mark 415 as a year of intellectual activity in Hypatia's life.

To get from this information a probability distribution for the year of birth, it is necessary to have the probability distribution of being intellectually active at a given age. This can be calculated given the probability of being alive at any given age and of being active at any given age (if alive), by simple multiplication.

To derive the first of these probability distributions we have used data from a 1974 mortality table for Italian males,⁶ clipping off ages under 18 since the subject was known to be intellectually active. The resulting probability distribution, a(x), is shown in Figure 3.



Figure 3. The probability distribution a(x) for an adult to reach a given age. The life expectancy comes to 71.8 years.

⁴She suffered this [violent death] because of the envy for her extraordinary wisdom, especially in the field of astronomy.

⁵Book VII, Chapter 15; translation from [Socrates Scholasticus, p. 160].

⁶All data are taken from http://www.mortality.org.

Name	Dates of birth and death	Lifespan
Accius, Lucius	170-circa 86 BC	~ 84
Adrianus (Hadrianus) of Tyre	circa AD 113–193	~ 80
Aelian (Claudius Aelianus)	AD 165/170-230/235	~ 65
Aeschines	circa 397–circa 322 BC	~ 65
Aeschylus	524/525-456/455 BC	~ 70
Agathocles (2) (of Cyzicus)	circa 275/265–circa 200/190 BC	~ 75
Alexander of Tralles	AD 525–605	80
Alexis	circa 375–circa 275 BC	~ 100
Ammianus Marcellinus	circa AD 330–395	~ 65
Anaxagoras	probably 500–428 BC	~ 72
Anaximenes (2) of Lampsacus	circa 380–320 BC	~ 60
Andocides	circa 440–circa 390 BC	~ 50
Androtion	circa 410–340 BC	~ 70
Antiphon	circa 480–411 BC	~ 69
Apollonius of Citium	circa 90–15 BC?	~ 75
Arcesilaus	316/315-242/241 BC	~ 74
Aristarchus of Samothrace	circa 216–144 BC	~ 72
Aristophanes of Byzantium	circa 257–180 BC	~ 77
Aristotle	384–322 BC	62
Arius	circa AD 260–336	~ 76
Arrian (Lucius Flavius Arrianus)	circa AD 86–160	\sim 74
Aspasius	circa AD 100–150	~ 50
Athanasius	circa AD 295–373	~ 78
Atticus	circa AD 150–200	~ 50
Augustine. Saint	AD 354–430	76
Bacchius of Tanagra	probably 275–200 BC	~ 75
Bacchylides	circa 520–450 BC	~ 70
Basil of Caesarea	circa AD 330–379	~ 49
Bion of Borysthenes	circa 335–circa 245 BC	~ 90
Carneades	214/213-129/128 BC	~ 85
Cassius (1)	31 BC-AD 37	68
Cassius Longinus	circa AD 213–273	~ 60
Cato (Censorius)	234–149 BC	85
Chrysippus of Soli	circa 280–207 BC	~ 73
Chrysostom, John	circa AD 354–407	\sim 53
Cinesias	circa 450–390 BC	~ 60
Claudius Atticus Herodes (2) Tiberius	circa AD 101–177	~ 76
Cleanthes of Assos	331–232 BC	99
Clitomachus	187/186–110/119 BC	~ 77
Colotes (RE 1) of Lampsacus	circa 325–260 BC	~ 65
Cornelius (RE 157) Fronto, Marcus	circa AD 95–circa 166	~ 71
Crantor of Soli in Cilicia	circa 335–275 BC	~ 60
Crates (2)	circa 368/365–288/285 BC	~ 80
Demades	circa 380–319 BC	~61
Demochares	circa 360–275 BC	~ 85
Democritus (of Abdera)	circa 460–370 BC	~ 90
Demosthenes (2)	384–322 BC	62
Dinarchus	circa 360-circa 290 BC	~ 70
Dio Cocceianus	circa 40/50-110/120 BC	~ 70
Diodorus (3) of Agyrium, Sicily	circa 90–30 BC	~ 60
Diogenes (3) (of Babylon)	circa 240–152 BC	${\sim}88$

Diogenes (2) the Cynic	circa 412/403-circa 324/321 BC	~ 85
Duris	circa 340–circa 260 BC	$\sim \! 80$
Empedocles	circa 492–432 BC	~ 60
Ennius, Quintus	239–169 BC	70
Ennodius, Magnus Felix	AD 473/474–521	$\sim \!\! 48$
Ephorus of Cyme	circa 405–330 BC	~ 75
Epicurus	341–270 BC	71
Epiphanius	circa AD 315–403	~ 88
Erasistratus	circa 315–240 BC	~ 75
Eratosthenes of Cyrene	circa 285–194 BC	~ 91
Eubulus (1)	circa 405–circa 335 BC	~ 70
Euclides (1) of Megara	circa 450–380 BC	~ 70
Euripides	probably 480s-407/406 BC	~ 78
Eusebius of Caesarea	circa AD 260–339	~ 79
Evagrius Scholasticus	circa AD 535–circa 600	~ 65
Favorinus	circa AD 85–155	~ 70
Fenestella	52 BC-AD 19 or 35 BC-AD 36	71
Galen of Pergamum	AD 129–216	87
Gorgias (1) of Leontini	circa 485–circa 380 BC	~ 105
Gregory (2) of Nazianzus	AD 329–389	60
Gregory (3) of Nyssa	circa AD 330–395	~ 65
Gregory (4) Thaumaturgus	circa AD 213–circa 275	~ 62
Hecataeus (2) of Abdera	circa 360–290 BC	~ 70
Hegesippus (1)	circa 390–circa 325 BC	~ 65
Hellanicus (1) of Lesbos	circa 480–395 BC	~ 85
Hellanicus (2)	circa 230/220–160/150 BC	~ 70
Herophilus of Chalcedon	circa 330–260 BC	~ 70
Hieronymus (2) of Rhodes	circa 290–230 BC	~ 60
Himerius	circa AD 310-circa 390	~ 80
Horace (Quintus Horatius Flaccus)	65–8 BC	57
Idomeneus (2)	circa 325–circa 270 BC	~ 55
Irenaeus	circa AD 130-circa 202	~ 72
Isaeus (1)	circa 420–340s BC	~ 75
Isocrates	436–338 BC	98
Ister	circa 250–200 BC	~ 50
Jerome (Eusebius Hieronymus)	circa AD 347–420	~ 73
Laberius, Decimus	circa 106–43 BC	~ 63
Libanius	AD 314-circa 393	~ 63
Livius Andronicus, Lucius	circa 280/270–200 BC	~ 75
Livy (Titus Livius)	59 BC-AD 17 or 64 BC-AD 12	76
Lucilius (1) Gaius	probably 180–102/101 BC	~ 75
Lucretius (Titus Lucretius Carus)	circa 94–55/51 BC	~ 41
Lyco	circa 300/298–226/224 BC	~ 74
Lycurgus (3)	circa 390–circa 325/324 BC	~ 65
Lydus	AD 490–circa 560	~ 70
Lysias	459/458-circa 380 BC or circa 445-circa 380 BC	~ 72
Malalas	circa AD 480–circa 570	~ 90
Mantias	circa 165–85 BC	$\sim \! 80$
Megasthenes	circa 350–290 BC	~ 60

Table 1. Life spans of the first 100 "ancient intellectuals" in *The Ox- ford Classical Dictionary*. The average, 71.7 years, is taken as typical.



Figure 4. The probability distribution $a_a(x)$ for being active at a given age, if alive.

The choice made for this distribution might appear questionable on two grounds: Is it appropriate to use modern data in studying an Alexandrian scholar of the fourth century AD? And assuming this is so, is the particular mortality table chosen adequate?

Our chief justification for keeping this choice of a(x) is that its most important feature for our purposes, the life expectancy, is in excellent agreement with a control value calculated for this purpose: the average lifespan of the first one hundred (in alphabetical order) "well dated" intellectuals found in The Oxford Classical Dictionary [Hornblower et al. 2012]⁷ (see Table 1). This suggests that using a(x)as an approximation for the mortality distribution of the population of interest is consistent with the available quantitative evidence.

To model the probability $a_a(x)$ of being intellectually active at a given age if alive at that age we make some reasonable, if somewhat arbitrary, assumptions reflected in the graph in Figure 4.

Combining the two distributions a(x) and $a_a(x)$ as explained, the probability of being active at any given age is calculated and - knowing that Hypatia was so in 415 — the probability distribution $\Upsilon_a(\xi)$ for the year of Hypatia's birth deduced from this historical datum is obtained in a straightforward manner (see Figure 5).

3.3. Hypatia reached old age. In his Χρονογραφία, John Malalas tells us that our subject was an old woman when she died:

Κατ΄ ἐκεῖνον δὲ τὸν καιρὸν παρρησίαν λαβόντες ὑπὸ τοῦ ἐπισκόπου οἱ Ἀλεξανδρεῖς ἔχαυσαν φρυγάνοις αὐθεντήσαντες Ὑπατίαν τὴν περι-

⁷The cutoff at 100 gives a convenient sample size large enough to be representative. Using all "ancient intellectuals" as the control population and not only those who lived in the third and fourth centuries AD is necessary in order to obtain a statistically significant sample.



Figure 5. The probability distribution $\Upsilon_a(\xi)$ for Hypatia's birth based on her being active when she died.



Figure 6. The probability distribution o(x) for being regarded as an old woman.

βόητον φιλόσοφον, περί ῆς μεγάλα ἐφέρετο ἦν δὲ παλαιὰ γυνή.⁸

In light of the average lifespan of ancient intellectuals (Table 1), even a conservative interpretation of "old woman" would preclude an age much below 50.9 Hence we model the probability distribution of someone being "old woman" by the function o(x) shown in Figure 6. The resulting probability distribution, $\Upsilon_o(\xi)$, for the year of Hypatia's birth based on this datum is then easily obtained; see Figure 7.

⁸At that time the Alexandrians, given free rein by their bishop, seized and burnt on a pyre of brushwood Hypatia the famous philosopher, who had a great reputation and who was an old woman [Malalas, XIV.12].

⁹This agrees with the authoritative opinion of many historians; thus Maria Dzielska [1995]: "John Malalas argues persuasively that at the time of her ghastly death Hypatia was an elderly woman not twenty-five years old (as Kingsley wants), nor even forty-five, as popularly assumed. Following Malalas, some scholars, including Wolf, correctly argue that Hypatia was born around 355 and was about sixty when she died".



Figure 7. The probability distribution $\Upsilon_o(\xi)$ for Hypatia's birth given that she reached old age.

3.4. *Hypatia, daughter of Theon.* Theon of Alexandria, best known for allowing the transmission of Euclid's *Elements* to the present day, was Hypatia's father. By knowing his birth year, one might think of deducing a probability distribution for the year of Hypatia's birth; sadly, this is unknown as well. Therefore, it is necessary to calculate a probability distribution for the year of Theon's birth first. To this end, two recorded facts are useful:

- Theon was intellectually active between 364 and 377.¹⁰
- Hypatia overhauled the third book of Theon's *Commentary on the Almagest* (Theon refers to this in the *Commentary* itself).

This second datum makes it unlikely that Hypatia was born in Theon's old age; it also make it less probable that he stopped being intellectually active at a young age, since he was still active while his daughter made her contribution to his work. To quantify this reasoning, we define notation for the relevant events:

- F_i , Theon becomes a father at age i.
- $A_i^{T/I}$, Theon/Hypatia is intellectually active at the age of *i*.
- C, Theon is able to collaborate with Hypatia (both are intellectually active).
- $B_k^{T/I}$, Theon/Hypatia begins being intellectually active at age k.
- $S_k^{T/I}$, Theon/Hypatia stops being intellectually active at age k.

The probability of Theon becoming a father at various ages is described approximately by the model distribution F(x) shown in Figure 8.

¹⁰In the *Little Commentary on Ptolemy's Handy Tables*, Theon mentioned some astronomical observations that can be dated with certainty: the two solar eclipses of June 15th and November 26th, 364 and an astral conjunction in 377. It is reasonable to assume that he was also active in the interval between those two years.



Figure 8. The probability distribution F(x) for Theon's age at the time of Hypatia's birth.

The probability of a subject (Theon or Hypatia) beginning their intellectual activity at a given age is described approximately by the model distribution B(x)shown in Figure 9.

The probability distribution S(x) for the subject ending her intellectual activity at a given age is taken to be, up to age 70, just the probability of dying (derived from the distribution a(x) of Figure 3), while after that it is the probabily of dying conditioned to that of being active, as obtained in Section 3.2. See Figure 10.

The probability of event C is therefore

$$P(C) = \sum_{i} \sum_{k} P(A_{i+k}^{T} \cap F_{i} \cap B_{k}^{I}).$$

By the definition of conditional probability,



Figure 9. The probability distribution B(x) for the starting point of one's intellectual career.



Figure 10. The probability distribution S(x) for the endpoint of one's intellectual career.

and since the beginning of the active life of Hypatia does not depend on her father's activity, the following simplification can be made:

$$\sum_{i} \sum_{k} P(A_{i+k}^T \cap B_k^I \mid F_i) \cdot P(F_i) = \sum_{i} \sum_{k} P(B_k^I) \cdot P(A_{i+k}^T \mid F_i) \cdot P(F_i).$$

Without committing a large error, it is possible to confuse the probability of being active at age i + k having had a daughter at age i, $P(A_{i+k}^T | F_i)$, with the one of being active at age i + k having been alive at age i (V_i) ,¹¹ $P(A_{i+k}^T | V_i)$:

$$P(A_{i+k}^T \mid F_i) \approx P(A_{i+k}^T \mid V_i) = \frac{P(A_{i+k}^T)}{P(V_i)}.$$

In the end, the following equation can be written:

$$P(C) = \sum_{i} \sum_{k} P(B_k^I) \cdot \frac{P(A_{i+k}^T)}{P(V_i)} \cdot P(F_i).$$

Based on the idea previously introduced, the next step is to calculate $P(F_i | C)$ and $P(S_k^T | C)$ (and so $P(A_i^T | C) = 1 - \sum_k P(S_k^T | C)$):

$$P(F_{i} | C) = \frac{P(F_{i} \cap C)}{P(C)} = \frac{\sum_{k} P(B_{k}^{I}) \cdot (P(A_{i+k}^{T})/P(V_{i})) \cdot P(F_{i})}{\sum_{i} \sum_{k} P(B_{k}^{I}) \cdot (P(A_{i+k}^{T})/P(V_{i})) \cdot P(F_{i})},$$

$$P(S_{k}^{T} | C) = \frac{P(S_{k} \cap C)}{P(C)} = \frac{\sum_{i,j:i+j \le k} P(S_{k}^{T}) \cdot P(F_{i}) \cdot P(B_{j}^{I})}{\sum_{i} \sum_{j} P(B_{j}^{I}) \cdot (P(A_{i+j}^{T})/P(V_{i})) \cdot P(F_{i})}.$$

 $A_C^T(x)$ is the probability distribution of Theon being active at a given age, conditioned to the *C* event; see Figure 11.

 $^{^{11}}V_i$ is obtained from the above-mentioned 1974 Italian male mortality data set.


Figure 11. The probability distribution $A_C^T(x)$ for Theon being active at a given age, given that his and Hypatia's periods of activity overlap.



Figure 12. The probability distributions $364(\xi)$ and $377(\xi)$.

Keeping in mind the two years in which Theon was surely active (364 and 377), two distributions $364(\xi)$ and $377(\xi)$ for Theon's year of birth are deduced as previously shown in Section 3.2 (see Figure 12). Then, following the procedure introduced in Section 2, a single distribution $\Theta(\xi)$ is obtained (see Figure 13).

Finally, in order to calculate $\Upsilon_d(\xi)$, the probability distribution for the year of Hypatia's birth based on her being Theon's daughter, the probability of the various events "the age difference between father and daughter is i years" conditioned on event C must be known. This is indeed the above-calculated $P(F_i | C)$, now written as the function $F_C(x)$ (see Figure 14) so that $\Upsilon_d(\xi)$ is straightforward to calculate:12

$$\Upsilon_d(\xi) = \sum_x \Theta(\xi) \cdot F_C(\xi - \xi).$$

(See Figure 15.)

¹²The sum is taken over the whole domain of $\Theta(\xi)$.



Figure 13. The probability distribution $\Theta(\xi)$ for Theon's birth.



Figure 14. The probability distribution $F_C(x)$ for the difference in age between father and daughter, given that their periods of activity overlap.



Figure 15. The probability distribution $\Upsilon_d(\xi)$ for Hypatia's birth based on her relationship to Theon.



Figure 16. The probability distribution T(x) for the age gap between teacher and disciple.

3.5. *Hypatia, teacher of Synesius.* Synesius of Cyrene, neo-Platonic philosopher and bishop of Ptolemais, was a disciple of Hypatia, as shown by a close correspondence between the two.

For instance, from his deathbed, Synesius wrote:

Τῆ φιλοσόφω. Κλινοπετὴς ὑπηγόρευσα τὴν ἐπιστολήν, ἢν ὑγιαίνουσα κομίσαιο, μῆτερ καὶ ἀδελφὴ καὶ διδάσκαλε καὶ διὰ πάντων τούτων εὐεργετικὴ καὶ πᾶν ὅ τι τίμιον καὶ πρᾶγμα καὶ ὄνομα.¹³

The distribution T(x) is introduced as a model to describe the probability of a difference of x years of age between teacher and pupil (see Figure 16).

 $\Upsilon_t(\xi)$, the probability distribution for the year of Hypatia's birth deduced from this historical datum, is obtained in a straightforward manner by taking 370 as the year of birth of Synesius¹⁴ (see Figure 17).

3.6. *Combined distribution.* Combining the five probability distributions deduced above for the year of Hypatia's birth, one final distribution, $\Upsilon(\xi)$, can be obtained following the rules introduced in Section 2. This final distribution $\Upsilon(\xi)$ can be compared to the distribution given by the simple arithmetic mean of the various distributions resulting from every possible combination of testimonies being considered true at the same time, $\Upsilon_A(\xi)$ (see Figure 18).

Therefore, the most probable year for the birth of Hypatia is $355 \ (\sim 14.5\%)$ with a total probability of the interval [350, 360] of about 90%.

¹³To the Philosopher. I am dictating this letter to you from my bed, but may you receive it in good health, mother, sister, teacher, and withal benefactress, and whatsoever is honored in name and deed [Synesius of Cyrene, Incipit of Letter 16].

¹⁴See, for example, [Hornblower et al. 2012].



Figure 17. The probability distribution $\Upsilon_t(\xi)$ for Hypatia's age based on her having been a teacher of Synesius.



Figure 18. The final probability distribution $\Upsilon(\xi)$ calculated for the birth of Hypatia using the method in Section 2 and an average distribution $\Upsilon_A(\xi)$ based on the same historical data.

4. Conclusions

The probabilistic dating model proposed in this work, structured in three steps, could be summarized by making use of a culinary analogy. The first step is represented by the collection of enough raw ingredients (testimonies) to be refined or "cooked" in the second step (turned into probability distributions) and — finally, in the third step — put together following a recipe (provided in Section 2) so that they blend well (as a single probability distribution).

Its application to the case of Hypatia proved to be satisfactory in that the final probability distribution shows a marked peak, making it possible to give a date with good precision. The result so obtained contradicts the prevalent opinion (cf. page 25) but is in agreement with the minority view held by some highly-regarded scholars working on the issue. We have already mentioned the authoritative opinion of Maria Dzielska, who deems that Hypatia died at about age 60, having been,

consequently, born around the year 355. A similar opinion is expressed in [Deakin 2007, p. 52].

Future applications appear to be far-reaching as the method could serve not only in cases strictly analogous to the one presented here but also in dating any event provided with a sufficient number of testimonies able to be turned into probability distributions.

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ON THE POSSIBLE EFFECTIVE ELASTICITY TENSORS OF 2-DIMENSIONAL AND 3-DIMENSIONAL PRINTED MATERIALS

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The set GU_f of possible effective elastic tensors of composites built from two materials with elasticity tensors $C_1 > 0$ and $C_2 = 0$ comprising the set U = $\{C_1, C_2\}$ and mixed in proportions f and 1 - f is partly characterized. The material with tensor $C_2 = 0$ corresponds to a material which is void. (For technical reasons C_2 is actually taken to be nonzero and we take the limit $C_2 \rightarrow 0$). Specifically, recalling that GU_f is completely characterized through minimums of sums of energies, involving a set of applied strains, and complementary energies, involving a set of applied stresses, we provide descriptions of microgeometries that in appropriate limits achieve the minimums in many cases. In these cases the calculation of the minimum is reduced to a finite-dimensional minimization problem that can be done numerically. Each microgeometry consists of a union of walls in appropriate directions, where the material in the wall is an appropriate *p*-mode material that is easily compliant to $p \leq 5$ independent applied strains, yet supports any stress in the orthogonal space. Thus the material can easily slip in certain directions along the walls. The region outside the walls contains "complementary Avellaneda material", which is a hierarchical laminate that minimizes the sum of complementary energies.

1. Introduction

Here we consider what effective elasticity tensors can be produced in the limit $\delta \rightarrow 0$ if we mix in prescribed proportions two materials with positive definite and bounded elasticity tensors C_1 and $C_2 = \delta C_0$. In the limit $\delta \rightarrow 0$ this represents a mixture of an elastic phase and an extremely compliant phase. Thus we are given a set $U = \{C_1, \delta C_0\}$ and we are aiming to characterize as best we can the set GU_f of all possible effective tensors of composites having a volume fraction f of phase 1. The elasticity tensor C_1 need not be isotropic but if it is anisotropic we require that it has a fixed orientation throughout the composite. Our results are summarized by the theorems in Section 10.

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To get an idea of the enormity of the problem one has to recognize that in three dimensions elasticity tensors can be represented by 6×6 matrices and these have 21 independent elements. The set of possible elasticity tensors is thus represented as a set in a 21-dimensional space. Even a distorted multidimensional cube in a 21dimensional space needs about 44 million real numbers to represent it (specifying the position in 21-dimensional space of each of the 2^{21} vertices). In the case where the two phases are isotropic, one is free to rotate the material to obtain an equivalent structure. Thus the set of possible elasticity tensors is invariant under rotations. As rotations involve three parameters (the Euler angles) this reduces the number of constants needed to describe the elasticity tensor from 21 to 21 - 3 = 18, and thus the elasticity tensor can be represented in an 18-dimensional space of tensor invariants. For example, in the generic case, one can take these 18 invariants as follows: the six eigenvalues of the elasticity tensor; the two independent elements of the normalized eigenstrain associated with the lowest eigenvalue that can be assumed to be diagonal by an appropriate choice of the coordinate axes (which then fixes these axes); the four independent elements of the normalized eigenstrain associated with the second lowest eigenvalue that is orthogonal to the first eigenstrain; the three independent elements of the normalized eigenstrain associated with the third lowest eigenvalue that is orthogonal to the first two eigenstrains; the two independent elements of the normalized eigenstrain associated with the third lowest eigenvalue that is orthogonal to the first three eigenstrains; and the one independent element of the normalized eigenstrain associated with the third lowest eigenvalue that is orthogonal to the first four eigenstrains. This brings the total to 6+2+4+3+2+1=18. In the same way that it takes two parameters (the bulk and shear moduli) to specify the elastic behavior of an isotropic material, it takes 18 parameters to specify the elastic behavior of a fully anisotropic material.

A distorted cube in this 18-dimensional space still requires about 4.7 million numbers to represent it. This makes exploring the range of possible elasticity tensors a daunting, if not impossible, numerical task. Some numerical exploration of this space has been done by Sigmund [1994; 1995], but we emphasize that this exploration covers only a tiny fraction of the number of possibilities.

Furthermore, the microstructures we found that lie near the boundary of GU_f have quite complicated multiscale architectures and thus would be difficult to find numerically. Also, it is not clear whether there are significantly simpler microstructures that can do the job. The numerical route of Sigmund should provide some simpler alternatives for the strut configurations in the multimode structures in the walls, although even then one needs to make subtle multiscale replacements (such as those appearing later in Figures 9 and 10) to achieve the desired performance. Numerical tests need to be made to see whether one can achieve the same performance with simpler structures. While strut configurations might be suitable at low

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volume fractions they are unlikely to be ideal at high volume fractions. Work by Allaire and Aubry [1999] shows that sometimes optimal microstructures necessarily have structure on multiple length scales. Even if one could numerically explore the question, it is not clear how one could summarize the results in a concise way.

From the applied side there is growing interest in trying to characterize the effective elasticity tensors of microstructures that can be produced by 3-dimensional or 2-dimensional printing. A dramatic example of such a microstructure is given in Figure 1. Our results have obvious relevance to this problem in the case where the 3dimensional printed material uses only one isotropic material plus void. Although our microstructures are somewhat extreme, they provide benchmarks that show what is theoretically possible. What is possible in practice will be a subset of this.

The microstructures we consider involve taking three limits. First, as they have structure on multiple length scales, the homogenization limit where the ratio between length scales goes to infinity needs to be taken. Second, the limit $\delta \rightarrow 0$ needs to be taken. Third, as the structure involves thin walls of width ϵ , along which the material can "slip", the limit $\epsilon \rightarrow 0$ needs to be taken so the contribution to the complementary energy of these walls goes to zero, when the structure supports an applied stress. (Here ϵ should not be confused with the size of the unit cell, as is common in homogenization theory). The limits should be taken in this order, as, for example, standard homogenization theory is justified only if $\delta \neq 0$, so we need to take the homogenization limit before taking the limit $\delta \rightarrow 0$. In the walled structures the material may only occupy a small volume fraction, but this is ultimately irrelevant as the thin walled structures the material (which goes to zero as $\epsilon \rightarrow 0$).

The case, applicable to printed materials, when phase 2 is actually void, rather than almost void, requires special care. To justify the homogenization steps taken here one has to first replace the void phase 2 with a composite foam having a small amount of phase 1 as the matrix phase, so that its effective elasticity tensor is nonzero, but approaches zero as the proportion of phase 1 in it tends to zero. The microgeometry in this composite needs to be much smaller than the scales in the geometries discussed here, which would involve mixtures of it and phase 1.

We emphasize, too, that our analysis is valid only for linear elasticity, and ignores nonlinear effects such as buckling. In reality the structures will easily buckle under compression. This buckling will occur, for example, in the square beam array structure of Figure 10. Additionally, some of the multimode materials are constructed via a superposition of appropriately shifted and deformed pentamode materials, and these substructures will interact under finite deformations. Also, in practice it would be difficult to realize the delicate multiscale materials that come close to attaining the bounds. Thus what is practically realizable will be just a subset, dependent on the current state of technology, of the set GU_f . While the title refers only to printed materials, the results are also applicable to any periodic, or statistically homogeneous, material containing voids or pores in a homogeneous material. Printed materials are more interesting than typical porous materials as they allow one to explore a wider range of interesting structures.

In a companion paper [Milton et al. 2017] we consider the opposite limit $\delta \to \infty$, corresponding to a mixture of an elastic material and an almost rigid material.

2. Review of some bounds on the elastic moduli of two-phase composites and geometries that attain them

Here we review a selection of results on sharp bounds on the elastic response of twophase composites and the associated problem of identifying optimal geometries that attain them. The interested reader is encouraged to look at the books of Nemat-Nasser and Hori [1998], Cherkaev [2000], Milton [2002], Allaire [2002], Torquato [2002] and Tartar [2009], which provide a much more comprehensive survey.

The most elementary bounds on the elastic properties of composites are the classical bounds of Hill [1952], who implicitly showed that

$$\langle [\boldsymbol{C}(\boldsymbol{x})]^{-1} \rangle^{-1} \leq \boldsymbol{C}_* \leq \langle \boldsymbol{C}(\boldsymbol{x}) \rangle.$$
(2-1)

Here the angular brackets $\langle \cdot \rangle$ denote a volume average, and the inequality holds in the sense of quadratic forms, i.e., for fourth-order tensors A and B satisfying the symmetries of elasticity tensors we say that $A \ge B$ if $\epsilon : A\epsilon \ge \epsilon : B\epsilon$ for all matrices ϵ . While these bounds were not explicitly stated by Hill in his 1952 paper they are an immediate and obvious consequence of his equation (2). If the two phases are isotropic the spectral decomposition of the elasticity tensors C_1 and C_2 of the two phases is

$$C_1 = 3\kappa_1 \Lambda_h + 2\mu_1 \Lambda_s$$
 and $C_2 = 3\kappa_2 \Lambda_h + 2\mu_2 \Lambda_s$, (2-2)

where κ_1 and κ_2 are the bulk moduli of the two phases, μ_1 and μ_2 are the shear moduli, and

$$\{\mathbf{\Lambda}_h\}_{ijk\ell} = \frac{1}{3}\delta_{ij}\delta_{k\ell}, \quad \{\mathbf{\Lambda}_s\}_{ijk\ell} = \frac{1}{2}[\delta_{ik}\delta_{j\ell} + \delta_{i\ell}\delta_{kj}] - \frac{1}{3}\delta_{ij}\delta_{k\ell}$$
(2-3)

act as projections. The tensor Λ_h projects onto the 1-dimensional space of matrices proportional to the second-order identity matrix, while Λ_s projects onto the 5-dimensional space of trace-free matrices. Similarly if the effective elasticity tensor C_* is isotropic we have that $C_* = 3\kappa_1\Lambda_h + 2\mu_1\Lambda_s$, where κ_* and μ_* are the effective bulk and shear moduli of the composite. In this paper we are interested in the case where the two phases are well-ordered in the sense that

$$\boldsymbol{C}_1 \ge \boldsymbol{C}_2, \tag{2-4}$$

and we will take the limit as $C_2 \rightarrow 0$, meaning that all the eigenvalues of C_2 approach zero. In the case of isotropic components this well-ordering assumption is satisfied if $\kappa_1 \ge \kappa_2$ and $\mu_1 \ge \mu_2$, and we will take the limit as $\kappa_2, \mu_2 \rightarrow 0$.

For isotropic composites of two well-ordered materials Hashin and Shtrikman [1963] and Hill [1963] obtained the celebrated bounds

$$\kappa_{*} \geq f\kappa_{1} + (1-f)\kappa_{2} - \frac{f(1-f)(\kappa_{1}-\kappa_{2})^{2}}{(1-f)\kappa_{1} + f\kappa_{2} + 4\mu_{2}/3},$$

$$\kappa_{*} \leq f\kappa_{1} + (1-f)\kappa_{2} - \frac{f(1-f)(\kappa_{1}-\kappa_{2})^{2}}{(1-f)\kappa_{1} + f\kappa_{2} + 4\mu_{1}/3},$$

$$\mu_{*} \geq f\mu_{1} + (1-f)\mu_{2} - \frac{f(1-f)(\mu_{1}-\mu_{2})^{2}}{(1-f)\mu_{1} + f\mu_{2} + \mu_{2}(9\kappa_{2} + 8\mu_{2})/[6(\kappa_{2} + 2\mu_{2})]},$$

$$\mu_{*} \leq f\mu_{1} + (1-f)\mu_{2} - \frac{f(1-f)(\mu_{1}-\mu_{2})^{2}}{(1-f)\mu_{1} + f\mu_{2} + \mu_{1}(9\kappa_{1} + 8\mu_{1})/[6(\kappa_{1} + 2\mu_{1})]}.$$
(2-5)

In fact these bounds (and the variational principles they derive from) hold even if one component has a negative bulk modulus, so long as the composite is stable [Kochmann and Milton 2014]. For 2-dimensional composites (fiber reinforced materials) analogous bounds on the effective elastic moduli were found by Hill [1964] and Hashin [1965]. Bounds on the complex effective bulk and shear moduli of isotropic two-phase 2-dimensional or 3-dimensional composites were also obtained [Gibiansky and Milton 1993; Milton and Berryman 1997; Gibiansky et al. 1993; 1999; Gibiansky and Lakes 1993; 1997]: these are appropriate to the propagation of fixed frequency elastic waves in composites when one or both of the phases is viscoelastic, and when the wavelength is much larger than the microstructure.

An important "attainability principle" is that bounds obtained by substituting a trial field in a variational principle will be attained when the geometry is such that the actual field matches this trial field. This principle was used, for example, in [Milton 1981c] to find geometries that attain the Hashin–Shtrikman bounds on the effective bulk modulus of composites with three or more phases (see also [Gibian-sky and Sigmund 2000]). The Hashin–Shtrikman variational principles involve a minimization over trial polarization fields, and the actual polarization field depends on the choice of the elasticity tensor C_0 of a "reference medium" (typically chosen to be positive definite) and is defined by

$$P(x) = (C(x) - C_0)\epsilon(x) = \sigma(x) - C_0\epsilon(x).$$
(2-6)

The variational principles require that $C(x) - C_0$ be either positive semidefinite or negative semidefinite, so in the case of a well-ordered material natural choices of C_0 are C_1 or C_2 and correspondingly the field will be zero in phase 1 or phase 2, respectively. The bounds are obtained by assuming it is constant in the other phase (proportional to the identity in case of the bulk modulus bounds, and trace-free for the shear modulus bounds). Hashin and Shtrikman [1963] recognized that the effective bulk modulus would be attained by the Hashin assemblage of coated spheres [Hashin 1962]. A single coated sphere can be a neutral inclusion: if the surrounding "matrix" material has an appropriate bulk modulus (with a specific value between κ_1 and κ_2) one can insert it in the matrix material without disturbing a surrounding hydrostatic field (this is the principle behind the unfeelability cloak of Bückmann, Thiel, Kadic, Schittny and Wegener [Bückmann et al. 2014]). The inclusion is invisible to the surrounding field and one can continue to insert similar inclusions, scaled to sizes ranging to the very small, until one essentially obtains a two-phase composite with effective bulk modulus the same as the original matrix material. Due to radial symmetry the forces acting on the spherical inner core will be equally distributed around the boundary and directed radially: thus the field inside the core material is hydrostatic and constant, and hence by the attainability principle, and due to their neutrality, sphere assemblages must attain the effective bulk modulus bounds in (2-5).

One very important class of microgeometries for which the field is constant in one phase are the sequentially layered laminates (first introduced by Maxwell [1873]) built by layering phase 2 with phase 1 in a direction n_1 (by which we mean n_1 is perpendicular to the layers), then taking this laminate and layering it again on a much larger length scale with phase 1 in a direction n_2 to obtain a "rank 2" laminate, and continuing this process until one obtains a rank *m* laminate, containing in a sense a "core" of phase 2 surrounded by layers of phase 1. The field is then constant in the core material of phase 2. An explicit formula for the effective elasticity tensor of such sequentially layered laminates was obtained by Francfort and Murat [1986], generalizing the analogous formulas obtained by Tartar [1985] for conductivity. Of course one can switch the roles of the phases in this construction and thus obtain a material where the field is constant in phase 1. It then immediately follows from the attainability principle (without requiring any calculation!) that one can attain the Hashin-Shtrikman shear modulus bounds (2-5) (and simultaneously the bulk modulus bounds) if one can find a sequentially layered laminate that has an isotropic elasticity tensor, and the easiest way to do this is to do the lamination sequentially by adding infinitesimal layers in random directions. This established the attainability of the Hashin-Shtrikman shear modulus bound [Milton 1986], also established independently and at the same time by Norris [1985], using the differential scheme that was known to be realizable [Milton 1985; Avellaneda 1987a] — in fact Roscoe [1973] had earlier realized the differential approximation scheme could produce the desired shear modulus - and at the same time elegantly by Francfort and Murat [1986], using sequentially layered laminates with just five directions of lamination (in the case of 3-dimensional composites).

Hill [1963] proved that the bulk modulus bounds are valid also in the non-wellordered case where $\mu_1 \ge \mu_2$ but $\kappa_1 \le \kappa_2$. As far as we know, the tightest bounds on the effective shear modulus of 3-dimensional composites in the non-well-ordered case where $\mu_1 \ge \mu_2$ but $\kappa_1 \le \kappa_2$ are those of Milton and Phan-Thien [1982]:

$$\min_{\substack{\zeta \\ 0 \leq \zeta \leq 1}} \frac{8\langle 6/\mu + 7/\kappa \rangle_{\zeta} + 15/\mu_{2}}{2(\langle 21/\mu + 2/\kappa \rangle_{\zeta}/\mu_{2} + 40\langle 1/\mu \rangle_{\zeta} \langle 1/\kappa \rangle_{\zeta})} \\
\leq \frac{f(1-f)(\mu_{1}-\mu_{2})^{2}}{f\mu_{1} + (1-f)\mu_{2} - \mu_{*}} - (1-f)\mu_{1} - f\mu_{2} \\
\leq \max_{\substack{\zeta \\ 0 \leq \zeta \leq 1}} \frac{8\mu_{1}\langle 6\kappa + 7\mu \rangle_{\zeta} + 15\langle \mu \rangle_{\zeta} \langle \kappa \rangle_{\zeta}}{2(\langle 21\kappa + 2\mu \rangle_{\zeta} + 40\mu_{1})},$$
(2-7)

where for any quantity *a* taking values a_1 and a_2 in phase 1 and phase 2, respectively, we define $\langle a \rangle_{\zeta} \equiv \zeta a_1 + (1 - \zeta)a_2$. These bounds are obtained by eliminating the geometric parameters from the bounds of Milton and Phan-Thien [1982] and are tighter than the better-known Walpole bounds [1966], and are in fact sharp (as they coincide with the Hashin–Shtrikman formula, which corresponds to particular geometries as we have discussed) when the moduli are slightly non-well-ordered. Specifically, the first bound in (2-7) is sharp when the minimum over ζ is attained at $\zeta = 0$, which occurs when

$$\kappa_1 - \kappa_2 \ge -\frac{(3\kappa_2 + 8\mu_2)^2}{42\kappa_2^2} \frac{\kappa_1\kappa_2}{\mu_1\mu_2} (\mu_1 - \mu_2), \tag{2-8}$$

and the second bound in (2-7) is sharp when the maximum over ζ is attained at $\zeta = 1$, which occurs when

$$\kappa_1 - \kappa_2 \ge -\frac{(3\kappa_1 + 8\mu_1)^2}{42\mu_1^2}(\mu_1 - \mu_2).$$
(2-9)

The bounds (2-5) and (2-7) constrain the pair (κ_* , μ_*) to lie in a rectangular box. Berryman and Milton [1988] obtained tighter coupled bounds which slice off two opposing corner regions of the box by eliminating the geometric parameters from the bulk modulus bounds of Beran and Molyneux [1966] (as simplified by Milton [1981b]) and from the shear modulus bounds of Milton and Phan-Thien [1982]. There is good reason to believe these bounds can be improved as the analogous 2-dimensional bounds are not as tight as the bounds of Cherkaev and Gibiansky [1993] coupling κ_* and μ_* , which were derived using the translation method.

For anisotropic composites with an effective tensor C_* , the microstructure independent bounds that are directly analogous to the Hashin–Shtrikman–Hill bounds,

given by (2-5), are the "trace bounds"

$$f \operatorname{Tr}[\mathbf{\Lambda}_{h}(\mathbf{C}_{*}-\mathbf{C}_{2})^{-1}] \leq \frac{1}{3(\kappa_{1}-\kappa_{2})} + \frac{1-f}{3\kappa_{2}+4\mu_{2}},$$

$$(1-f) \operatorname{Tr}[\mathbf{\Lambda}_{h}(\mathbf{C}_{1}-\mathbf{C}_{*})^{-1}] \leq \frac{1}{3(\kappa_{1}-\kappa_{2})} - \frac{f}{3\kappa_{1}+4\mu_{1}},$$

$$f \operatorname{Tr}[\mathbf{\Lambda}_{s}(\mathbf{C}_{*}-\mathbf{C}_{2})^{-1}] \leq \frac{5}{2(\mu_{1}-\mu_{2})} + \frac{3(\kappa_{2}+2\mu_{2})(1-f)}{\mu_{2}(3\kappa_{2}+4\mu_{2})},$$

$$(1-f) \operatorname{Tr}[\mathbf{\Lambda}_{s}(\mathbf{C}_{1}-\mathbf{C}_{*})^{-1}] \leq \frac{5}{2(\mu_{1}-\mu_{2})} - \frac{3(\kappa_{1}+2\mu_{1})f}{\mu_{1}(3\kappa_{1}+4\mu_{1})},$$

$$(2-10)$$

obtained independently by Milton and Kohn [1988] and Zhikov [1988; 1991a; 1991b]. In these expressions the fourth-order tensors Λ_h multiply the fourth-order tensors on their right, and

$$\operatorname{Tr}[\mathbf{A}] = A_{ijij} \tag{2-11}$$

defines the "trace" of a fourth-order tensor (see also [Francfort and Murat 1986] and [Nemat-Nasser and Hori 1993] for related bounds). From the attainability principle it follows that these bounds will be achieved whenever the composite is a sequentially layered laminate, with a core of one phase, surrounded by layers (on widely separated length scales) of the other phase. When C_* is isotropic these bounds (2-10) reduce to the Hashin–Shtrikman–Hill bounds (2-5). In the case where the two phases, and hence the composite, are incompressible we can define the five effective shear moduli $\mu_1^*, \mu_2^*, \mu_3^*, \mu_4^*, \mu_5^*$ to be the five finite eigenvalues of $\frac{1}{2}C_*$, and the second pair of bounds in (2-10) reduce to

$$\sum_{i=1}^{5} \frac{f}{2(\mu_{*i} - \mu_2)} \le \frac{5}{2(\mu_1 - \mu_2)} + \frac{3(\kappa_2 + 2\mu_2)(1 - f)}{\mu_2(3\kappa_2 + 4\mu_2)},$$

$$\sum_{i=1}^{5} \frac{1 - f}{2(\mu_1 - \mu_{*i})} \le \frac{5}{2(\mu_1 - \mu_2)} - \frac{3(\kappa_1 + 2\mu_1)f}{\mu_1(3\kappa_1 + 4\mu_1)}.$$
(2-12)

Lipton [1988] established that the analogous bounds for the two effective shear moduli μ_1^* and μ_2^* of 2-dimensional composites of two incompressible isotropic phases completely characterize GU_f .

Earlier, Willis [1977] considered anisotropic composites and used the Hashin– Shtrikman variational principle with a trial polarization that was zero in one phase and constant in the other to obtain bounds on the elastic energy of a two-phase composite. He found that these bounds are not microgeometry independent, but rather involve the two-point correlation function, i.e., the probability that a rod with fixed orientation lands with both ends in phase 1 when thrown randomly in a composite. It follows from the attainability principle that the Willis bounds will be achieved when the composite is a sequentially layered laminate, with a core of one phase, surrounded by layers (on widely separated length scales) of the other phase.

In a major advance, Avellaneda [1987b] recognized that for any composite of two phases with well-ordered tensors not all the information contained in the twopoint correlation function was relevant to determining the bounds: what was relevant was a "reduced two-point correlation function" that could be represented as a positive measure $\mu(\boldsymbol{\xi})$ (with unit integral) on the sphere $|\boldsymbol{\xi}| = 1$. Roughly speaking one takes the Fourier transform of the two-point correlation function and integrates it over rays $\mathbf{k} = k\boldsymbol{\xi}$ in "Fourier space" keeping $\boldsymbol{\xi}$ fixed and integrating over k from 0 to infinity. Most importantly, every such measure could be realized to an arbitrarily high degree of approximation by the measure of a suitable sequentially layered laminate. For example, a measure with weighted delta functions in directions ξ_1 and $\boldsymbol{\xi}_2$ would be realized by a second-rank sequentially layered laminate with layers normal to ξ_1 and ξ_2 . (We note in passing that these reduced two-point correlation functions of Avellaneda are a special case of the H-measures introduced at the same time by Tartar [1989; 1990], in terms of which he could calculate second-order corrections to the effective tensor of a nearly homogeneous composite. *H*-measures were also introduced independently by Gérard [1989; 1994] under the name of microlocal defect measures. For composites of two isotropic phases the Hashin-Shtrikman conductivity bounds, and indeed variational conductivity bounds at any order, can be naturally expressed in terms of the series expansion coefficients of the effective tensor up to a corresponding order for a nearly homogeneous composite, as shown by Milton and McPhedran [1982].)

The fantastic implication was that by summing the Willis bounds [1977], and then minimizing over all positive measures on the sphere, one would get sharp bounds on the sum of elastic complementary energies

$$W_{f}^{0}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\sigma}_{5}^{0}, \boldsymbol{\sigma}_{6}^{0}) = \min_{\boldsymbol{C}_{*} \in GU_{f}} \sum_{j=1}^{6} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0}, \qquad (2-13)$$

and similarly one could get sharp bounds on the sum of elastic energies

$$W_{f}^{6}(\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}, \boldsymbol{\epsilon}_{6}^{0}) = \min_{\boldsymbol{C}_{*} \in GU_{f}} \sum_{i=1}^{6} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0}.$$
(2-14)

Here some of the applied stresses σ_j^0 or the applied strains ϵ_i^0 could be zero. So the evaluation of the functions $W_f^0(\sigma_1^0, \sigma_2^0, \sigma_3^0, \sigma_4^0, \sigma_5^0, \sigma_6^0)$ and $W_f^6(\epsilon_1^0, \epsilon_2^0, \epsilon_3^0, \epsilon_4^0, \epsilon_5^0, \epsilon_6^0)$ reduces to a finite-dimensional minimization problem which can be done numerically. Hence we will treat the functions $W_f^0(\sigma_1^0, \sigma_2^0, \sigma_3^0, \sigma_4^0, \sigma_5^0, \sigma_6^0)$ as being known, and we will call an "Avellaneda material" an associated sequentially layered laminate material with effective tensor $C_* = C_f^A(\epsilon_1^0, \epsilon_2^0, \epsilon_3^0, \epsilon_4^0, \epsilon_5^0, \epsilon_6^0)$ that

attains the minimum in (2-14), and similarly we call a "complementary Avellaneda material" an associated sequentially layered laminate material with effective tensor $C_* = \widetilde{C}_f^A(\sigma_1^0, \sigma_2^0, \sigma_3^0, \sigma_4^0, \sigma_5^0, \sigma_6^0) \in GU_f$ that attains the minimum in (2-13). Explicit analytical formulas for the tensors $C_f^A(\epsilon_1^0, \epsilon_2^0, \epsilon_3^0, \epsilon_4^0, \epsilon_5^0, \epsilon_6^0)$ and $\widetilde{C}_f^A(\sigma_1^0, \sigma_2^0, \sigma_3^0, \sigma_4^0, \sigma_5^0, \sigma_6^0)$ are not generally available, but rather have to be found by numerical computation. When $C_1 \ge C_2$ one needs to take the minimum in (2-13) over the C_* of sequentially layered laminates with a core material of phase 2. Similarly, when $C_1^{-1} \ge C_2^{-1}$ the minimum in (2-14) also can be taken over the C_* of sequentially layered laminates with a core material of phase 2. We remark that although Avellaneda assumed the tensors C_1 and C_2 were isotropic, his analysis easily extends to the case where the tensors are anisotropic but well-ordered (either with $C_1 \ge C_2$ or $C_2 \ge C_1$) and with constant orientation throughout the composite: see, for example, Section 23.3 in [Milton 2002].

These C_* of sequentially layered laminates are given by the formula of Francfort and Murat [1986] and Gibiansky and Cherkaev [1997b]:

$$(1-f)(\boldsymbol{C}_1 - \boldsymbol{C}_*)^{-1} = (\boldsymbol{C}_1 - \boldsymbol{C}_2)^{-1} - f \sum_{j=1}^r c_j \boldsymbol{\Gamma}(\boldsymbol{n}_j), \qquad (2-15)$$

where *r* is the rank of the sequential laminate, the positive weights c_j sum to 1, the n_i are the lamination directions, and $\Gamma(n)$ is the fourth-order tensor with elements given by

$$\{\boldsymbol{\Gamma}(\boldsymbol{n})\}_{hik\ell} = \frac{1}{4} \Big(n_h \{\boldsymbol{C}(\boldsymbol{n})^{-1}\}_{ik} n_\ell + n_h \{\boldsymbol{C}(\boldsymbol{n})^{-1}\}_{i\ell} n_k + n_i \{\boldsymbol{C}(\boldsymbol{n})^{-1}\}_{hk} n_\ell + n_i \{\boldsymbol{C}(\boldsymbol{n})^{-1}\}_{h\ell} n_k \Big), \quad (2\text{-}16)$$

in which $C(n) = n \cdot C_1 n$ is the 3 × 3 matrix known as the acoustic tensor, with elements

$$\{C(n)\}_{ik} = \{n \cdot C_1 n\}_{ik} = n_h \{C_1\}_{hik\ell} n_\ell.$$
(2-17)

Thus the minimum needs to be taken over the rank r of the sequential laminate, over the positive weights c_j , which sum to 1, and over the lamination directions n_j . In the case where phase 1 is isotropic, with bulk modulus κ_1 and shear modulus μ_1 , C(n) can be easily calculated and one obtains

$$\{\boldsymbol{\Gamma}(\boldsymbol{n}_{j})\}_{hik\ell} = \frac{3n_{h}n_{i}n_{k}n_{\ell}}{3\kappa_{1}+4\mu_{1}} + \frac{1}{4\mu_{1}}(n_{h}\delta_{ik}n_{\ell}+n_{h}\delta_{i\ell}n_{k}+n_{i}\delta_{hk}n_{\ell}+n_{i}\delta_{h\ell}n_{k}-4n_{h}n_{i}n_{k}n_{\ell}).$$
(2-18)

Francfort, Murat, and Tartar [Francfort et al. 1995] proved that when C_1 is isotropic it suffices to limit attention to laminates of rank $r \le 6$. When C_1 is anisotropic we extend an argument due to Avellaneda [1987b]. Consider the set A consisting of

all fourth-order tensors A of the form

$$A = \int_{|\boldsymbol{n}|=1} \boldsymbol{\Gamma}(\boldsymbol{n}) \, \boldsymbol{m}(d\boldsymbol{n}), \qquad (2-19)$$

where m(dn) is a nonnegative measure on the unit sphere having an integral of 1 over the sphere. Since A satisfies

$$\{A\}_{hik\ell}\{C_1\}_{hik\ell} = \int_{|\boldsymbol{n}|=1} \{C(\boldsymbol{n})^{-1}\}_{ik}\{C(\boldsymbol{n})\}_{ik} \, m(d\boldsymbol{n}) = 3, \qquad (2-20)$$

it follows that A is a convex set in a space of dimension v = 20 (with 20 of the 21 independent matrix elements of A as coordinates, and the remaining element being determined by (2-20)). The extreme points correspond to point masses on the unit sphere. Hence any tensor of the form (2-19) is a convex combination of at most v + 1 extreme points. Thus the sum (2-15) can be limited to $r \le 21$; i.e., it suffices to consider laminates up to rank 21. Lipton [1991; 1992; 1994] obtained a complete algebraic characterization of the possible sequentially layered laminates having transverse or orthotropic symmetry and derived explicit expressions for many of the associated bounds. The Avellaneda materials are of course difficult to build in practice since they have structure on multiple length scales. However, if f is small and one phase is void, Bourdin and Kohn [2008] showed that it suffices to use a walled structure (similar to the structure on the right in Figure 4, but with walls in many directions, not just two, and with the wall thickness depending on orientation).

As observed by Avellaneda [1987b], the implications of course also apply to 2-dimensional elasticity. Define

$$W_{f}^{0}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}) = \min_{\boldsymbol{C}_{*} \in GU_{f}} \sum_{j=1}^{3} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0}$$
(2-21)

and

$$W_f^3(\boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0) = \min_{\boldsymbol{C}_* \in GU_f} \sum_{i=1}^3 \boldsymbol{\epsilon}_i^0 : \boldsymbol{C}_* \boldsymbol{\epsilon}_i^0.$$
(2-22)

Then there is an Avellaneda material with effective tensor $C_* = C_f^A(\epsilon_1^0, \epsilon_2^0, \epsilon_3^0)$ that attains the minimum in (2-22), and a complementary Avellaneda material with effective tensor $C_* = \widetilde{C}_f^A(\sigma_1^0, \sigma_2^0, \sigma_3^0) \in GU_f$ that attains the minimum in (2-21). In 2-dimensional elasticity, sequentially layered laminates have elasticity tensors given by (2-15)–(2-17) when the tensor C_1 is anisotropic. When the elasticity tensor C_1 of phase 1 is isotropic, the sequentially layered laminates of rank r have effective compliance tensors $S_* = (C_*)^{-1}$ given by the Gibiansky–Cherkaev formula

$$(1-f)(\mathbf{S}_1 - \mathbf{S}_*)^{-1} = (\mathbf{S}_1 - \mathbf{S}_2)^{-1} - f[(4\kappa_2)^{-1} + (4\mu_2)^{-1}]\mathbf{M}$$
(2-23)

(see [Gibiansky and Cherkaev 1997b, equations (2.37) and (2.38)] and see also [Lurie et al. 1982], in which Lurie, Cherkaev, and Fedorov derived an equivalent, but less simple, formula), where $S_1 = (C_1)^{-1}$ and $S_2 = (C_2)^{-1}$ are the compliance tensors of the two phases, occupying volume fractions f and 1 - f, respectively, and M has elements

$$\{\boldsymbol{M}\}_{hik\ell} = \sum_{j=1}^{r} c_j \boldsymbol{t}_{j_h} \boldsymbol{t}_{j_\ell} \boldsymbol{t}_{j_\ell}, \qquad (2-24)$$

in which the t_j are unit vectors perpendicular to the directions of lamination (i.e., parallel to the layer boundaries), and the c_j are any set of positive weights, summing to 1, giving the proportions of phase 1 laminated in the various directions. The tensor M is clearly positive semidefinite and has the property that

$$\{M\}_{hkhk} = \{M\}_{hhkk} = 1.$$
(2-25)

Conversely, Avellaneda and Milton [1989] have shown that given a positive semidefinite fourth-order tensor M satisfying (2-25) there is a sequential layered laminate of rank $r \leq 3$ that corresponds to it, i.e., such that (2-23) holds for some choice of unit vectors t_j and weights c_j (see also Theorem 2.2 of [Francfort et al. 1995]). Thus when C_1 is isotropic, the computation of the complementary Avellaneda tensor $\tilde{C}_f^A(\sigma_1^0, \sigma_2^0, \sigma_3^0)$ reduces to a minimization over positive semidefinite fourth-order tensors M satisfying (2-25). When C_1 is anisotropic, by the same argument as in the 3-dimensional case, it suffices to consider sequential layered laminates of rank at most 6.

We also remark that aside from hierarchical laminates there are many other structures that have a uniform field in one phase, sometimes only for certain applied fields. These include assemblages of confocal ellipses and ellipsoids [Milton 1980; 1981a; Grabovsky and Kohn 1995a], the periodic Vigdergauz geometries [Vigdergauz 1986; 1994; 1996; 1999; Grabovsky and Kohn 1995b], the Sigmund structures [2000], and the periodic *E*-inclusions of Liu, James, and Leo [Liu et al. 2007] (see also Section 23.9 of [Milton 2002]). Usually these attain the bounds when the measure $\mu(\boldsymbol{\xi})$ minimizing the sum of Willis bounds is not required to be a discrete measure. Allaire and Aubry [1999] have shown that sometimes the best microstructure necessarily has structure on multiple length scales (like sequentially layered laminates).

For single energies for anisotropic two-phase composites, the Hill bounds (2-1) imply

$$\epsilon_{0} : [fC_{1}^{-1} + (1-f)C_{2}^{-1}]^{-1}\epsilon_{0} \le \epsilon_{0} : C_{*}\epsilon_{0} \le \epsilon_{0} : [fC_{1} + (1-f)C_{2}]\epsilon_{0},$$

$$\sigma_{0} : [fC_{1} + (1-f)C_{2}]^{-1}\sigma_{0} \le \sigma_{0} : C_{*}^{-1}\sigma_{0} \le \sigma_{0} : [fC_{1}^{-1} + (1-f)C_{2}^{-1}]\sigma_{0}.$$
 (2-26)

Improved, and in fact sharp, upper and lower bounds on the elastic energy $\epsilon_0 : C_* \epsilon_0$ in terms of the given applied strain ϵ_0 and sharp upper and lower bounds on the complementary elastic energy $\sigma_0 : C_*^{-1} \sigma_0$ in terms of the given applied stress σ_0 were obtained for isotropic component materials by Gibiansky and Cherkaev [1997a], Kohn and Lipton [1988], and Allaire and Kohn [1993a; 1993b; 1994]. The paper of Gibiansky and Cherkaev [1997a] was for the fourth-order plate equation, but this can be mapped to the equivalent 2-dimensional elasticity problem considered by Allaire and Kohn [1993b]. Their lower bounds on $\sigma^0 : C_*^{-1}\sigma^0$ are equivalent to the bounds that for any tensor $C_* \in GU_f$,

$$\boldsymbol{\sigma}^{0}: \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}^{0} \geq \boldsymbol{\sigma}^{0}: [\boldsymbol{\widetilde{C}}_{f}^{A}(\boldsymbol{\sigma}^{0}, 0, 0)]^{-1}\boldsymbol{\sigma}^{0},$$
(2-27)

and they provided an explicit formula for the right-hand side for any 2×2 symmetric matrix σ^0 representing the applied stress. This bound can be viewed in two ways: in the way originally interpreted, i.e., as a bound on the possible (elastic energy, average stress, volume fraction) triplets; or as a bound

$$\boldsymbol{\sigma}^{0}:\boldsymbol{\epsilon}^{0} \geq \boldsymbol{\sigma}^{0}: [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}^{0},0,0)]^{-1}\boldsymbol{\sigma}^{0}$$
(2-28)

on the possible (average stress, average strain, volume fraction) triplets. Here $\epsilon^0 = C_*^{-1}\sigma^0$ is the strain associated with σ^0 . Significantly, Milton, Serkov, and Movchan [Milton et al. 2003] found that the inequality (2-28) completely characterizes the possible (average stress, average strain, volume fraction) triplets in the limit in which one phase becomes void, when the other phase is isotropic. Specifically, given any triplet (σ^0 , ϵ^0 , f) satisfying (2-28) as an inequality, they give a recipe for constructing a 2-dimensional microstructure with effective tensor C_* and having phase 1 occupy a volume fraction f such that $\sigma^0 = C_*\epsilon^0$.

For 3-dimensional composites explicit expressions for the optimal upper energy bound were found by Gibiansky and Cherkaev [1997b] and Allaire [1994] for the case of a two-phase composite where one of the phases is void or rigid [Gibiansky and Cherkaev 1997b]. Grabovsky [1996] obtained energy bounds for two-phase composites containing anisotropic phases, each with a constant orientation.

Another major advance was made by Milton and Cherkaev [1995], who showed that any desired positive definite fourth-order tensor which has the symmetries of an elasticity tensor could be realized as the effective elasticity tensor C_* of a composite of a sufficiently stiff isotropic material and a sufficiently compliant isotropic material. One key to this advance was the realization that certain structures called pentamode materials could be (arbitrarily) stiff to one applied stress σ_1^0 and yet have five mutually orthogonal strains ϵ_1^0 , ϵ_2^0 , ϵ_3^0 , ϵ_4^0 , ϵ_5^0 , each orthogonal to σ_1^0 as five (arbitrarily compliant) easy modes of deformations (hence the name pentamode).



Figure 1. An electron micrograph of the pentamode structure created by Kadic, Bückmann, Stenger, Thiel and Wegener [Kadic et al. 2012] using a 3-dimensional lithography technique. (Used with the kind permission of Martin Wegener.)

For such a pentamode

$$W_{f}^{5}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}) = \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\left(\sum_{i=1}^{5} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} \right) + \boldsymbol{\sigma}_{1}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{1}^{0} \right] \quad (2\text{-}29)$$

approaches zero as the constituent stiff isotropic material becomes increasingly stiff and the constituent compliant isotropic material becomes increasingly compliant. The lattice structure of a pentamode is similar to that of diamond with a stiff double cone structure replacing each carbon bond. This structure ensures that the tips of four double cone structures meet at each vertex. This is the essential feature: treating the double cone structures as struts, the tension in one determines uniquely the tension in the other three. This is simply the balance of forces. Thus the structure as a whole can essentially support only one stress. Pentamode structures were experimentally realized by Kadic, Bückmann, Stenger, Thiel and Wegener [Kadic et al. 2012] in an incredible feat of precision three-dimensional lithography. One of their electron micrographs of the structure is shown in Figure 1. Pentamode structures were also independently discovered in 1995 by Sigmund, although he did not find the complete span of pentamode structures needed here: one needs pentamodes that can support any chosen stress, not just a hydrostatic one. It is this aspect of pentamodes that makes them more interesting than, for example, a gel. Gels are examples of pentamodes as they are easy to shear, but difficult to compress under a hydrostatic loading $\sigma_1 = I$. By contrast the pentamodes of Milton and Cherkaev could be stiff to any desired stress σ_1^0 : this desired stress

may be a mixture of shear and compression, and may have eigenvalues of mixed signs. A simple argument for seeing that these pentamodes can achieve any desired elasticity tensor was given in the foreword of the book edited by Phani and Hussein [2017]. To recapitulate that argument, one expresses the desired C_* in terms of its eigenvectors and eigenvalues,

$$\boldsymbol{C}_* = \sum_{i=1}^6 \lambda_i \boldsymbol{v}_i \otimes \boldsymbol{v}_i.$$
(2-30)

The idea, roughly speaking, is to find six pentamode structures each supporting a stress represented by the vector v_i for i = 1, 2, ..., 6. The stiffness of the material and the necks of the junction regions at the vertices need to be adjusted so each pentamode structure has an effective elasticity tensor close to

$$\boldsymbol{C}_{*}^{(i)} = \lambda_{i} \boldsymbol{v}_{i} \otimes \boldsymbol{v}_{i}. \tag{2-31}$$

Then one successively superimposes all these six pentamode structures, with their lattice structures being offset to avoid collisions. Additionally one may need to deform the structures appropriately to avoid these collisions as described in [Milton and Cherkaev 1995], and when one does this it is necessary to readjust the stiffness of the material in the structure to maintain the value of λ_i . Then the remaining void in the structure is replaced by an extremely compliant material. (Its presence is needed just for technical reasons, to ensure that the assumptions of homogenization theory are valid so that the elastic properties can be described by an effective tensor.) But it is so compliant that essentially the effective elasticity tensor is just a sum of the effective elasticity tensors of the six pentamodes, i.e., the elastic interaction between the six pentamodes is negligible. In this way we arrive at a material with (approximately) the desired elasticity tensor C_* .

It is worth mentioning that with extremely high contrast materials the homogenized equations are not necessarily the usual linear elasticity equations, but can also include nonlocal terms. Nonlocal interactions can be obtained for example with an extremely stiff dumbbell-shaped inclusion with the balls arbitrarily distant. If the bar joining them is not only extremely stiff but also extremely thin, then it does not directly couple with the surrounding elastic material (except in the very near vicinity of the bar, where it is obviously deformed by it), but provides a nonlocal interaction between the balls. In fact, amazingly, Camar-Eddine and Seppecher [2003] have completely characterized all possible linear macroscopic behaviors of any high contrast composite: they showed that any energetically stable behavior can be obtained using materials with such dumbbell-shaped inclusions interacting at many length scales. Some interesting examples of high contrast materials with exotic effective behaviors have been given by Seppecher, Alibert, and dell'Isola [Seppecher et al. 2011].



Figure 2. Left: A convex set is the envelope of its tangent planes. The positions of the two tangent planes with normal n are determined by the Legendre transform f(n) and f(-n) defined by (3-1). Specifically f(n) and f(-n) give the distances of the tangent planes from the origin. Right: An example highlighting an interesting case discussed in the text that helps give a geometrical interpretation of the results of the paper.

3. Characterizing convex sets and G-closures for elasticity

Let *G* be a convex set of real *d*-dimensional vectors, meaning that if $c_1, c_2 \in G$ then $\theta c_1 + (1 - \theta)c_2 \in G$ for all $\theta \in [0, 1]$. As shown in Figure 2 (left) for d = 2such a convex set can be completely characterized by its Legendre transform,

$$f(\boldsymbol{n}) = \min_{\boldsymbol{c} \in G} \boldsymbol{n} \cdot \boldsymbol{c}. \tag{3-1}$$

Clearly this function satisfies the homogeneity property that

$$f(\lambda \boldsymbol{n}) = \lambda f(\boldsymbol{n}) \quad \text{for all } \lambda > 0, \tag{3-2}$$

and consequently it suffices to know f(n) for all unit vectors n to recover the function f(n) for any vector n. The values of f(n) and f(-n) give the positions of the two planes with normals $\pm n$ that are tangent to G: specifically |f(n)| and |f(-n)| give the distances from these tangent planes to the origin. By varying n and taking the intersection of the regions between the planes one recovers G: the set G is the envelope of its tangent planes as illustrated in Figure 2 (left). Thus the Legendre transform function f(n) with |n| = 1 completely characterizes G.

The example of Figure 2 (right) is also illuminating for the purposes of this paper. Let n and m be the vectors

$$\boldsymbol{n} = \begin{pmatrix} 0\\1 \end{pmatrix}, \quad \boldsymbol{m} = \begin{pmatrix} 1\\0 \end{pmatrix}, \tag{3-3}$$

and consider $f(\mathbf{n} + \alpha \mathbf{m})$ for $\alpha \ge 0$ in the context of this example. (Of course $\mathbf{n} + \alpha \mathbf{m}$ is only a unit vector when $\alpha = 0$.) As the boundary of G contains a flat section

orthogonal to \boldsymbol{n} , the vector \boldsymbol{c} which attains the minimum in (3-1) is not unique. In the diagram both \boldsymbol{c}^A and \boldsymbol{c}_* are minimizers. However, for an infinitesimal value of $\alpha > 0$, \boldsymbol{c}_* is selected as the unique minimizer and remains the minimizer no matter how large $\alpha > 0$ becomes. Furthermore, since \boldsymbol{c}_* is orthogonal to \boldsymbol{m} the value of $f(\boldsymbol{n} + \alpha \boldsymbol{m})$ remains constant for all $\alpha \ge 0$.

If G is a convex set of, say, real $d \times d$ matrices it can be similarly characterized by its Legendre transform,

$$f(N) = \min_{C \in G}(N, C), \tag{3-4}$$

defined for all $d \times d$ matrices N, where (N, C) is an inner product on the space of matrices which we may take to be

$$(N, C) = N_{ij}C_{ij} \equiv N : C, \tag{3-5}$$

where we have adopted the Einstein summation convention that sums over repeated indices are assumed, and the double dot ":" denotes a double contraction of indices. This is exactly equivalent to (3-1) if we think of the matrix C being represented by the vector c of its matrix elements. Note that if G only contains symmetric matrices, then it suffices to take N as a symmetric matrix since (A, C) = 0 if C is symmetric and A is antisymmetric.

Similarly, if G is a convex set of fourth-order elasticity tensors C satisfying the usual symmetries

$$C_{ijk\ell} = C_{jik\ell} = C_{k\ell ij},\tag{3-6}$$

then it can be characterized by the Legendre transform (3-4) with an inner product

$$(N, C) = N_{ijk\ell} C_{ijk\ell}, \qquad (3-7)$$

and again it suffices to assume N has the same symmetries as C, i.e., those in (3-6).

However, *G*-closures (i.e., sets of all possible effective tensors) are not generally convex sets. Nevertheless, they do have some convexity properties as a consequence of their stability under lamination. In the case of the set GU_f where $U = \{C_1, \delta C_2\}$, we can take two materials with effective tensors $C_1^*, C_2^* \in GU_f$ and laminate them together in a direction *n* (representing the vector perpendicular to the layers) in proportions θ and $1 - \theta$ to obtain an effective tensor $C_*(n, \theta)$ which necessarily lies in the set GU_f for all $\theta \in [0, 1]$. While $C_*(n, \theta)$ is not a linear average of C_1^* and C_2^* , there exist fractional linear transformations T_n of fourth-order tensors such that lamination in direction *n* reduces to a linear average [Backus 1962; Milton 1990] (see also [Tartar 1979]):

$$T_{\boldsymbol{n}}(\boldsymbol{C}_{*}(\boldsymbol{n},\theta)) = \theta T_{\boldsymbol{n}}(\boldsymbol{C}_{1}^{*}) + (1-\theta)T_{\boldsymbol{n}}(\boldsymbol{C}_{2}^{*}) \quad \text{for all } \theta \in [0,1].$$
(3-8)

Thus $T_n(GU_f)$ must be a convex set of fourth-order tensors. In the particular case where a set of effective tensors has no interior, i.e., is constrained to lie on a manifold of dimension *m* smaller than the dimension of the space of fourth-order tensors

satisfying the symmetries of elasticity tensors (i.e., m < 21 for 3-dimensional composites and m < 6 for 2-dimensional composites), then as recognized by Grabovsky [1998] (see also [Grabovsky and Sage 1998]) T_n must map this manifold to a subset of a hyperplane of dimension m for any value of n. This places rather severe constraints on the form of such manifolds. Identifying such manifolds is important as they represent exact relations satisfied by effective tensors, no matter what the geometry of the composite happens to be. Thus these constraints provide necessary conditions for an exact relation. Later, sufficient conditions for an exact relation to hold were obtained [Grabovsky et al. 2000].

Unfortunately, the use of Legendre transforms of the convex set $T_n(GU_f)$ is not useful to us as we are unaware of any direct variational principles for $T_n(C_*)$. An alternative approach was prompted by work of Cherkaev and Gibiansky [1992; 1993], who found that bounding sums of energies and complementary energies could lead to very useful bounds on *G*-closures. It was proved by Francfort and Milton [Francfort and Milton 1994; Milton 1994] that minimums over $C_* \in GU_f$ of such sums of energies and complementary energies completely characterize GU_f in much the same way that Legendre transforms characterize convex sets: the stability under lamination of GU_f is what allows one to recover GU_f from the values of these minimums (see also Chapter 30 in [Milton 2002]). Figure 3 captures the idea of this characterization.



Figure 3. *G*-closures are characterized by minimums of sums of energies and complementary energies. The coordinates here represent the elements of the effective elasticity tensor C_* . Then a plane represents a surface where a sum of energies is constant, and when this sum takes its minimum value the plane is tangent to the *G*-closure. The convexity properties of the *G*-closure guarantee that the surfaces corresponding to the minimums of sums of energies and complementary energies wrap around the *G*-closure and touch each point on its boundary. (Reproduction of Figure 30.1 in [Milton 2002].)

Specifically, in the case of 3-dimensional elasticity, the set GU_f is completely characterized if we know the seven "energy functions",

$$\begin{split} W_{f}^{0}(\sigma_{1}^{0}, \sigma_{2}^{0}, \sigma_{3}^{0}, \sigma_{4}^{0}, \sigma_{5}^{0}, \sigma_{6}^{0}) &= \min_{\boldsymbol{C}_{*} \in GU_{f}} \sum_{j=1}^{6} \sigma_{j}^{0} : \boldsymbol{C}_{*}^{-1} \sigma_{j}^{0}, \\ W_{f}^{1}(\sigma_{1}^{0}, \sigma_{2}^{0}, \sigma_{3}^{0}, \sigma_{4}^{0}, \sigma_{5}^{0}, \epsilon_{1}^{0}) &= \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\boldsymbol{\epsilon}_{1}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{1}^{0} + \sum_{j=1}^{5} \sigma_{j}^{0} : \boldsymbol{C}_{*}^{-1} \sigma_{j}^{0} \right], \\ W_{f}^{2}(\sigma_{1}^{0}, \sigma_{2}^{0}, \sigma_{3}^{0}, \sigma_{4}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}) &= \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{4} \sigma_{j}^{0} : \boldsymbol{C}_{*}^{-1} \sigma_{j}^{0} \right], \\ W_{f}^{3}(\sigma_{1}^{0}, \sigma_{2}^{0}, \sigma_{3}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}) &= \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\sum_{i=1}^{3} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{3} \sigma_{j}^{0} : \boldsymbol{C}_{*}^{-1} \sigma_{j}^{0} \right], \\ W_{f}^{4}(\sigma_{1}^{0}, \sigma_{2}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}) &= \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\sum_{i=1}^{4} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{2} \sigma_{j}^{0} : \boldsymbol{C}_{*}^{-1} \sigma_{j}^{0} \right], \\ W_{f}^{5}(\sigma_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}) &= \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\sum_{i=1}^{5} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{2} \sigma_{j}^{0} : \boldsymbol{C}_{*}^{-1} \sigma_{j}^{0} \right], \\ W_{f}^{5}(\sigma_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}) &= \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\left(\sum_{i=1}^{5} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} \right) + \sigma_{1}^{0} : \boldsymbol{C}_{*}^{-1} \sigma_{1}^{0} \right], \\ W_{f}^{6}(\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}) &= \min_{\boldsymbol{C}_{*} \in GU_{f}} \sum_{i=1}^{6} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0}. \\ \end{array} \right\}$$

In fact, it suffices [Milton and Cherkaev 1995] to know these functions for sets of applied strains ϵ_i^0 and applied stresses σ_i^0 that are mutually orthogonal:

$$(\boldsymbol{\epsilon}_{i}^{0}, \boldsymbol{\sigma}_{j}^{0}) = 0, \quad (\boldsymbol{\epsilon}_{i}^{0}, \boldsymbol{\epsilon}_{k}^{0}) = 0, \quad (\boldsymbol{\sigma}_{j}^{0}, \boldsymbol{\sigma}_{\ell}^{0}) = 0,$$

for all i, j, k, ℓ with $i \neq j, i \neq k, j \neq \ell$. (3-10)

Each of these terms in the minimums has a physical significance. For example, in the expression for W_f^2 ,

$$\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{4} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0}$$
(3-11)

has the physical interpretation of being the sum of energies per unit volume stored in the composite with effective elasticity tensor C_* when successively subjected to the two applied strains ϵ_1^0 and ϵ_2^0 and then to the four applied stresses $\sigma_1^0, \sigma_2^0, \sigma_3^0$ and σ_4^0 . To distinguish the terms $\epsilon_i^0 : C_* \epsilon_i^0$ and $\sigma_j^0 : C_*^{-1} \sigma_j^0$, the first is called an energy (it is really an energy per unit volume associated with the applied strain ϵ_i^0) and the second is called a complementary energy, although it too physically represents an energy per unit volume associated with the applied stress σ_j^0 . Note that the quantity (3-11) can be equivalently written as

$$(C_*, N) + (C_*^{-1}, N'),$$
 (3-12)

where

$$N = \sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} \otimes \boldsymbol{\epsilon}_{i}^{0}, \quad N' = \sum_{j=1}^{4} \boldsymbol{\sigma}_{j}^{0} \otimes \boldsymbol{\sigma}_{j}^{0}, \quad (3-13)$$

in which for any $d \times d$ symmetric matrix A, the tensor $A \otimes A$ is defined to be the fourth-order tensor with elements

$$\{A \otimes A\}_{ijk\ell} = \{A\}_{ij}\{A\}_{k\ell}.$$
(3-14)

If we decompose the positive semidefinite tensors N and N' into their spectral decompositions

$$N = \sum_{i=1}^{2} \lambda_i \boldsymbol{v}_i \otimes \boldsymbol{v}_i, \quad N' = \sum_{j=1}^{4} \lambda'_j \boldsymbol{v}'_j \otimes \boldsymbol{v}'_j, \quad (3-15)$$

with eigenmatrices v_i and v'_j and corresponding nonnegative eigenvalues λ_i and λ'_j , then, with the orthogonality constraints (3-10), we can make the identifications

$$\boldsymbol{\epsilon}_{i}^{0} = \sqrt{\lambda_{i}} \boldsymbol{v}_{i}, \quad \boldsymbol{\sigma}_{j}^{0} = \sqrt{\lambda_{j}'} \boldsymbol{v}_{j}.$$
(3-16)

Note that due to the orthogonality conditions (3-10) the fourth-order tensors N and N' have the property that the product NN' is zero. Here the product of two fourth-order tensors C and C' is given by

$$\{CC'\}_{ijk\ell} = \{C\}_{ijmn}\{C'\}_{mnk\ell}.$$
(3-17)

Thus in the same way that convex sets are the envelope of planes, the *G*-closure GU_f is the envelope of special surfaces parametrized by positive semidefinite fourth-order tensors N and N' satisfying the symmetries of elasticity tensors, and having zero product NN' = N'N = 0 (i.e., the range of N' is in the null space of N, and conversely the range of N is in the null space of N'). These special surfaces consist of all positive definite fourth-order tensors C satisfying

$$(C, N) + (C^{-1}, N') = c,$$
 (3-18)

where c is a positive real constant. In the case N' = 0 this does represent a hyperplane, but its orientation is restricted by the fact that the outward normal to the surface N is restricted to be a positive definite fourth-order tensor (by outward normal we mean the normal pointing away from the origin). Knowledge of the seven

functions W_f^i given by (3-9) is clearly equivalent to knowledge of the function

$$W_f(N, N') = \min_{C_* \in GU_f} (C_*, N) + (C_*^{-1}, N')$$
(3-19)

for all positive semidefinite fourth-order tensors N and N' satisfying the symmetries of elasticity tensors and having NN' = 0. The formula for recovering GU_f from $W_f(N, N')$ is then

$$\bigcap_{\substack{N,N' \ge 0 \\ NN'=0}} \left\{ \boldsymbol{C} : (\boldsymbol{C}, N) + (\boldsymbol{C}^{-1}, N') \ge W_f(N, N') \right\} = GU_f.$$
(3-20)

More generally if we replace GU_f in (3-19) by another set G of positive definite matrices, and if the left-hand side of (3-20) is again G, then we may say G is "W-convex".

An explicit definition of W-convexity is as follows: a set G of positive definite symmetric matrices is said to be strictly W-convex if G is simply connected and if for every pair of positive semidefinite symmetric matrices N and N', not both zero, the minimum in

$$\min_{C \in G} (C, N) + (C^{-1}, N')$$
(3-21)

is uniquely attained by only one $C \in G$. Geometrically, G is strictly W-convex if for all positive semidefinite symmetric matrices N and N', not both zero, the surface that consists of all positive definite matrices C satisfying

$$(C, N) + (C^{-1}, N') = k, (3-22)$$

where k is chosen as the smallest value for which this surface touches G, has the property that it touches G at only one point. A set G is W-convex if it is a limit of strictly W-convex sets. If the set G has a smooth boundary, then the condition for W-convexity can be expressed in terms of the curvature of the boundary of G: when G is a set of matrices, this curvature at each point on the surface of G is a fourth-order tensor; when G is a set of fourth-order elasticity tensors, this curvature is an eighth-order tensor. (See equation (3.51) in [Milton 1994], or equation (30.11) in [Milton 2002], for the explicit inequalities that the curvature must satisfy.)

The stability of GU_f under lamination implies it is *W*-convex, but *W*-convexity probably does not imply stability under lamination, as stability under lamination depends on the underlying partial differential equations. Associated with any set *G* of symmetric positive definite matrices *C* is its *W*-transform, defined as

$$W(N, N') = \min_{C \in G} (C, N) + (C^{-1}, N'), \qquad (3-23)$$

where N and N' are symmetric positive semidefinite matrices satisfying NN' = 0, and the inner product of two symmetric matrices A and B can be taken as (A, B) =

Tr(*AB*), where Tr denotes the trace (sum of diagonal elements) of a matrix. To see some of the properties of *W*-transforms it is helpful to extend the definition of the transform to allow for matrices *N* and *N'* that have a nonzero product, $NN' \neq 0$. The defining equation, (3-23), remains the same. Then consider a weighted average of (N_1, N'_1) and (N_2, N'_2) , with weights θ and $1 - \theta$, where the four matrices N_1, N'_1, N_2, N'_2 are positive semidefinite. Then for any $\theta \in (0, 1)$, we have

$$W(\theta N_{1} + (1-\theta)N_{2}, \theta N'_{1} + (1-\theta)N'_{2}) = \min_{C \in G} \{\theta[(C, N_{1}) + (C^{-1}, N'_{1})] + (1-\theta)[(C, N_{2}) + (C^{-1}, N'_{2})]\} \\ \geq \theta \{\min_{C \in G} (C, N_{1}) + (C^{-1}, N'_{1})\} + (1-\theta) \{\min_{C \in G} (C, N_{2}) + (C^{-1}, N'_{2})\} \\ \geq \theta W_{f}(N_{1}, N'_{1}) + (1-\theta)W_{f}(N_{2}, N'_{2}),$$
(3-24)

which (by definition) implies W(N, N') is a jointly concave function of N and N'. This concavity is a well-known property of Legendre transforms.

4. Variational principles

Upper bounds on the sums of energies and complementary energies can easily be obtained from classic energy minimization variational principles. For example, in the case of the sum (3-11), we have

$$\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{4} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0}$$
$$= \min_{\boldsymbol{\epsilon}_{1}, \boldsymbol{\epsilon}_{2}, \boldsymbol{\sigma}_{1}, \boldsymbol{\sigma}_{2}, \boldsymbol{\sigma}_{3}, \boldsymbol{\sigma}_{4}} \left\langle \sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}(\boldsymbol{x}) : \boldsymbol{C}(\boldsymbol{x}) \boldsymbol{\epsilon}_{i}(\boldsymbol{x}) + \sum_{j=1}^{4} \boldsymbol{\sigma}_{j}(\boldsymbol{x}) : [\boldsymbol{C}(\boldsymbol{x})]^{-1} \boldsymbol{\sigma}_{j}(\boldsymbol{x}) \right\rangle, \quad (4-1)$$

where the minimum is over a set of two trial strain fields $\underline{\epsilon}_1(x)$ and $\underline{\epsilon}_2(x)$ and a set of four trial stress fields $\underline{\sigma}_1(x)$, $\underline{\sigma}_2(x)$, $\underline{\sigma}_3(x)$, and $\underline{\sigma}_4(x)$ that have the prescribed average values

$$\langle \underline{\epsilon}_i \rangle = \epsilon_i^0 \quad \text{for } i = 1, 2, \qquad \langle \underline{\sigma}_j \rangle = \sigma_j^0 \quad \text{for } j = 1, 2, 3, 4,$$
 (4-2)

and are subject to the differential constraints that

$$\underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) = \frac{1}{2} \left(\nabla \underline{\boldsymbol{u}}_{i}(\boldsymbol{x}) + (\nabla \underline{\boldsymbol{u}}_{i}(\boldsymbol{x}))^{T} \right) \quad \text{for } i = 1, 2,$$

$$\nabla \cdot \boldsymbol{\sigma}_{i}(\boldsymbol{x}) = 0 \quad \text{for } j = 1, 2, 3, 4,$$
(4-3)

where *T* denotes the transpose (reflecting the matrix about its diagonal) and $\underline{u}_i(x)$ is the trial displacement field associated with the trial stress field $\underline{\epsilon}_i(x)$. The trial strain fields $\underline{\epsilon}_i(x)$ and the trial stress fields $\underline{\sigma}_j(x)$ (but not the trial displacement fields) should be chosen to be periodic (if the composite is periodic), quasiperiodic (if the composite is quasiperiodic), or statistically homogeneous (if the composite

is statistically homogeneous). It may be the case that the material has structure on widely separated length scales. Maybe it can be viewed as a mixture of two composites, one with effective tensor C_*^1 and a second with effective tensor C_*^2 , so that at the mesoscale it has a geometry described by a characteristic function $\chi_*(\mathbf{x})$, where $\chi_*(\mathbf{x})$ is 1 in the composite with effective tensor C_*^1 and 0 in the material with effective tensor C_*^2 . Naturally the length scale, or length scales, of variations in $\chi_*(\mathbf{x})$ should be much larger than the variations in the microstructure of the materials that have the effective tensors C_*^1 and C_*^2 . Then we can treat the material having effective tensor as a composite of the materials C_*^1 and C_*^2 and we have the variational principle

$$\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{4} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0}$$

$$= \min_{\underline{\boldsymbol{\epsilon}}_{1}, \underline{\boldsymbol{\epsilon}}_{2}, \underline{\boldsymbol{\sigma}}_{1}, \underline{\boldsymbol{\sigma}}_{2}, \underline{\boldsymbol{\sigma}}_{3}, \underline{\boldsymbol{\sigma}}_{4}} \left\langle \sum_{i=1}^{2} \underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) : \left[\chi_{*}(\boldsymbol{x}) \boldsymbol{C}_{*}^{1} + (1 - \chi_{*}(\boldsymbol{x})) \boldsymbol{C}_{*}^{2} \right] \underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x})$$

$$+ \sum_{j=1}^{4} \underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x}) : \left[\chi_{*}(\boldsymbol{x}) \boldsymbol{C}_{*}^{1} + (1 - \chi_{*}(\boldsymbol{x})) \boldsymbol{C}_{*}^{2} \right]^{-1} \underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x}) \right\rangle, \quad (4-4)$$

where again the minimum is over fields subject to the appropriate average values and differential constraints. Particular choices of trial fields will then lead to an upper bound on this sum of energies and complementary energies. To bound the quantities on the right one may again use variational principles. When x is in the material C_*^k for k = 1 or 2, one has the variational principles

$$\underline{\epsilon}_{i}(\boldsymbol{x}): \boldsymbol{C}_{*}^{k} \underline{\epsilon}_{i}(\boldsymbol{x}) = \min_{\underline{\epsilon}_{i}} \langle \underline{\epsilon}_{i}(\boldsymbol{x}, \boldsymbol{y}): \boldsymbol{C}^{k}(\boldsymbol{y}) \underline{\epsilon}_{i}(\boldsymbol{x}, \boldsymbol{y}) \rangle_{\boldsymbol{y}},$$

$$\underline{\sigma}_{j}(\boldsymbol{x}): [\boldsymbol{C}_{*}^{k}]^{-1} \underline{\sigma}_{j}(\boldsymbol{x}) = \min_{\underline{\sigma}_{j}} \langle \underline{\sigma}_{j}(\boldsymbol{x}, \boldsymbol{y}): [\boldsymbol{C}^{k}(\boldsymbol{y})]^{-1} \underline{\sigma}_{j}(\boldsymbol{x}, \boldsymbol{y}) \rangle_{\boldsymbol{y}},$$
(4-5)

where $\langle \cdot \rangle_y$ now denotes an average over the y variable (x is the "slow variable" and y is the "fast variable") and

$$C^{k}(\mathbf{y}) = \chi^{k}(\mathbf{y})C_{1} + (1 - \chi^{k}(\mathbf{y}))C_{2}, \qquad (4-6)$$

in which $\chi^k(\mathbf{y})$ is the characteristic function representing the geometry associated with the effective tensor C_*^k , taking a value 1 in the material with tensor C_1 and 0 in the material with tensor C_2 . Here the trial fields have the prescribed average values

$$\langle \underline{\underline{\epsilon}}_i(\mathbf{x}, \mathbf{y}) \rangle_{\mathbf{y}} = \underline{\epsilon}_i(\mathbf{x}) \text{ for } i = 1, 2, \quad \langle \underline{\underline{\sigma}}_j(\mathbf{x}, \mathbf{y}) \rangle_{\mathbf{y}} = \underline{\sigma}_j(\mathbf{x}) \text{ for } j = 1, 2, 3, 4, (4-7)$$

and are subject to the differential constraints

$$\underline{\underline{\epsilon}}_{i}(\boldsymbol{x}, \boldsymbol{y}) = \frac{1}{2} \left(\nabla_{y} \underline{\underline{\mu}}_{i}(\boldsymbol{x}, \boldsymbol{y}) + (\nabla \underline{\underline{\mu}}_{i}(\boldsymbol{x}, \boldsymbol{y}))^{T} \right) \quad \text{for } i = 1, 2,$$

$$\nabla \cdot_{y} \underline{\underline{\sigma}}_{j}(\boldsymbol{x}, \boldsymbol{y}) = 0 \quad \text{for } j = 1, 2, 3, 4,$$
(4-8)

where ∇_y and ∇_y are the gradient and divergence with respect to the *y* variables. We call the step of replacing the variational principle (4-1) by the variational principles (4-4) and (4-5) the "homogenization at intermediate scales step".

In this paper we will choose trial fields that satisfy the *local orthogonality condition* that

$$\underline{\boldsymbol{\epsilon}}_i(\boldsymbol{x}) : \underline{\boldsymbol{\sigma}}_j(\boldsymbol{x}) = 0, \quad \text{for all } \boldsymbol{x}. \tag{4-9}$$

Using the differential constraints satisfied by the trial fields, and integration by parts, one sees that the associated average fields are necessarily orthogonal too:

$$\boldsymbol{\epsilon}_{i}^{0}:\boldsymbol{\sigma}_{j}^{0}=\langle \underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x})\rangle:\langle \underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x})\rangle=\langle \underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}):\underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x})\rangle=0.$$
(4-10)

5. Finding most of the energy functions

Recall from Section 2 that an complementary Avellaneda material is a sequentially layered laminate material with phase 1 occupying a volume fraction f and with effective tensor

$$\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\sigma}_{3}^{0},\boldsymbol{\sigma}_{4}^{0},\boldsymbol{\sigma}_{5}^{0},0)$$

that attains equality in (2-13). It is found by minimizing the right-hand side of (2-13) as C_* varies within the class of tensors given by (2-15)–(2-17) with $C_2 = 0$, as the rank r, the positive weights c_j which sum to 1, and the unit vectors \mathbf{n}_i are varied. Here some of the applied stresses $\boldsymbol{\sigma}_j^0$ could be zero. Since the energy $\boldsymbol{\sigma}_j^0 : C_*^{-1} \boldsymbol{\sigma}_j^0$ associated with any applied stress $\boldsymbol{\sigma}_j^0$ is necessarily nonnegative, we obtain from (3-9) the bounds

$$\begin{split} \sum_{j=1}^{5} \boldsymbol{\sigma}_{j}^{0} &: [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\sigma}_{5}^{0}, 0)]^{-1} \boldsymbol{\sigma}_{j}^{0} \leq W_{f}^{1}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\sigma}_{5}^{0}, \boldsymbol{\epsilon}_{1}^{0}), \\ \sum_{j=1}^{4} \boldsymbol{\sigma}_{j}^{0} : [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, 0, 0)]^{-1} \boldsymbol{\sigma}_{j}^{0} \leq W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}), \\ \sum_{j=1}^{3} \boldsymbol{\sigma}_{j}^{0} : [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, 0, 0, 0)]^{-1} \boldsymbol{\sigma}_{j}^{0} \leq W_{f}^{3}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}), \quad (5-1) \\ \sum_{j=1}^{2} \boldsymbol{\sigma}_{j}^{0} : [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, 0, 0, 0, 0)]^{-1} \boldsymbol{\sigma}_{j}^{0} \leq W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}), \\ \boldsymbol{\sigma}_{1}^{0} : [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, 0, 0, 0, 0, 0)]^{-1} \boldsymbol{\sigma}_{j}^{0} \leq W_{f}^{5}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}), \\ \boldsymbol{\sigma}_{1}^{0} : [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, 0, 0, 0, 0, 0, 0)]^{-1} \boldsymbol{\sigma}_{1}^{0} \leq W_{f}^{5}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}), \\ \boldsymbol{0} \leq W_{f}^{6}(\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}, \boldsymbol{\epsilon}_{6}^{0}). \end{aligned}$$

The last inequality is clearly sharp, being attained when the composite consists of islands of phase 1 surrounded by a phase 2 (so that C_* approaches 0 as $\delta \rightarrow 0$).

The objective of this paper is to show that many of the other inequalities are also sharp in the limit $\delta \to 0$, at least when the spaces spanned by the applied strains ϵ_j^0 for j = 1, 2, ..., p satisfy certain properties. This space of applied strains \mathcal{V}_p , associated with W_f^p , has dimension p and is spanned by $\epsilon_1^0, \epsilon_2^0, ..., \epsilon_p^0$.

The recipe for doing this is to simply insert into a relevant complementary Avellaneda material a microstructure occupying a thin walled region, such that the material can slip along the walls when the applied strain lies in appropriate spaces \mathcal{V}_p , yet which is such that the combination of Avellaneda material and walled material can support without slip any applied stress in the subspace orthogonal to \mathcal{V}_p . This will be possible only when \mathcal{V}_p is spanned by symmetrized rank 1 matrices, taking the form

$$\boldsymbol{\epsilon}^{(k)} = \frac{1}{2} (\boldsymbol{a}_k \boldsymbol{n}_k^T + \boldsymbol{n}_k \boldsymbol{a}_k^T), \quad \text{for } k = 1, \dots, p.$$
 (5-2)

The existence of such matrices $\epsilon^{(k)}$ is proved in Section 7. The proof uses small perturbations of the applied stresses and strains. But, due to the continuity of the energy functions W_f^k established in Section 9, the small perturbations do not modify the generic result. The vectors n_k determine the orientation of the walls in the structure. For each n_k there is a set of parallel walls perpendicular to n_k that allow slip given by the strain $\epsilon^{(k)}$. We say slip but it should be recognized that $\epsilon^{(k)}$ is not generally a pure shear, but rather a combination of dilation and shear, since it does not generally have zero trace.

To define the thin walled structure, introduce the periodic function $H_c(x)$ with period 1 which takes the value 1 if $x - [x] \le c$, where [x] is the greatest integer less than x, and $c \in [0, 1]$ gives the thickness of each wall relative to the spacing between walls (which is unity). Then for the unit vectors $\mathbf{n}_1, \mathbf{n}_2, \ldots, \mathbf{n}_p$ appearing in (5-2), and for a small relative wall thickness $c = \epsilon$, define the characteristic functions

$$\eta_k(\boldsymbol{x}) = H_\epsilon(\boldsymbol{x} \cdot \boldsymbol{n}_k + k/p). \tag{5-3}$$

This characteristic function defines a series of parallel walls, as shown on the left in Figure 4, each perpendicular to the vector \mathbf{n}_j , where $\eta_j(\mathbf{x}) = 1$ in the wall material. The additional shift term k/p in (5-3) ensures the walls associated with k_1 and k_2 do not intersect when it happens that $\mathbf{n}_{k_1} = \mathbf{n}_{k_2}$, at least when ϵ is small. Note that ϵ is a volume fraction, not a homogenization parameter. We will be taking the limit $\epsilon \rightarrow 0$ after taking the homogenization limit.

Now define the characteristic function

$$\chi_*(\mathbf{x}) = \prod_{k=1}^p (1 - \eta_k(\mathbf{x})).$$
 (5-4)

If $p \le 3$, this is usually a periodic function of x, an exception being if p = 3 and there are no nonzero integers z_1 , z_2 , and z_3 such that $z_1n_1 + z_2n_2 + z_3n_3 = 0$. More



Figure 4. Example of walled structures. On the left we have a "rank 1" walled structure and on the right a "rank 2" walled structure. The generalization to walled structures of any rank is obvious, and precisely defined by the characteristic function (5-4) that is 0 in the walls, and 1 in the remaining material.

generally, $\chi_*(x)$ is a quasiperiodic function of x. The walled structure is where $\chi_*(x)$ takes the value 0. In the case p = 2 the walled structure is illustrated on the right in Figure 4.

Recall that a *p*-mode material is a material for which there are *p* independent strains to which the material is easily compliant, yet the material is much more resistant to any strain in the (6-p)-dimensional orthogonal subspace. In this sense the microstructure of Figure 1 is a pentamode material. We consider a subclass of multimode materials which can still support stresses in the limit $\delta \to 0$. We say a composite with effective tensor C_* built from the two materials C_1 and $C_2 = \delta C_0$ is easily compliant to a strain ϵ_i^0 if the elastic energy $\epsilon_i^0 : C_* \epsilon_i^0$ goes to zero as $\delta \to 0$, and supports a stress σ_j^0 if the complementary energy $\sigma_j^0 : C_*^{-1} \sigma_j^0$ has a nonzero limit as $\delta \to 0$. We desire *p*-mode materials for which there are *p* independent strains to which the material is easily compliant, yet for which the material supports any stress in the (6-p)-dimensional orthogonal subspace. The pentamode structure of Figure 1 needs to be modified as all its elastic moduli go to zero as $\delta \to 0$. The multimode structures we will introduce have structure on multiple length scales and it is important that one takes the limit of an infinite separation of length scales (so one can apply homogenization theory) before taking the limit $\delta \to 0$.

Inside the walled structure, where $\chi_*(\mathbf{x}) = 0$, we put a *p*-mode material with effective tensor $C_*^2 = C_*(\mathcal{V}_p)$ that supports any applied stress σ^0 in the space orthogonal to \mathcal{V}_p and which is easily compliant to any strain ϵ^0 in the space \mathcal{V}_p . When we take the six matrices

$$\boldsymbol{v}_1 = \boldsymbol{\sigma}_1^0 / |\boldsymbol{\sigma}_1^0|, \dots, \boldsymbol{v}_{6-p} = \boldsymbol{\sigma}_{6-p}^0 / |\boldsymbol{\sigma}_{6-p}^0|, \boldsymbol{v}_{7-p} = \boldsymbol{\epsilon}_1^0 / |\boldsymbol{\epsilon}_1^0|, \dots, \boldsymbol{v}_6 = \boldsymbol{\epsilon}_p^0 / |\boldsymbol{\epsilon}_p^0|$$
(5-5)

as an orthonormal basis for the space of 6×6 matrices, we need to find a *p*-mode material for which the elasticity tensor C_*^2 in this basis is such that

$$\lim_{\delta \to 0} \boldsymbol{C}_*^2 = \begin{pmatrix} \boldsymbol{A} & \boldsymbol{0} \\ \boldsymbol{0} & \boldsymbol{0} \end{pmatrix}, \tag{5-6}$$

where A represents a (strictly) positive definite $(6-p) \times (6-p)$ matrix and the 0 on the diagonal represents the $p \times p$ zero matrix.

Outside the walled structure, where $\chi_*(x) = 1$, we put the complementary Avellaneda material with effective elasticity tensor

$$\boldsymbol{C}_*^1 = \widetilde{\boldsymbol{C}}_f^A(\boldsymbol{\sigma}_1^0, \dots, \boldsymbol{\sigma}_{6-p}^0, 0, \dots, 0).$$

In a variational principle similar to (4-4) (i.e., treating the complementary Avellaneda material and the *p*-mode material both as homogeneous materials with effective tensors C_*^1 and C_*^2 , respectively) we choose trial stress fields that are constant,

$$\underline{\boldsymbol{\sigma}}_j(\boldsymbol{x}) = \boldsymbol{\sigma}_j^0, \tag{5-7}$$

thus trivially fulfilling the differential constraints, and trial strain fields of the form

$$\underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) = \sum_{k=1}^{p} \boldsymbol{\epsilon}_{i,k} \eta_{k}(\boldsymbol{x}) / \boldsymbol{\epsilon}, \qquad (5-8)$$

which are required to have the average values

$$\boldsymbol{\epsilon}_{i}^{0} = \langle \underline{\boldsymbol{\epsilon}}_{i} \rangle = \sum_{k=1}^{p} \boldsymbol{\epsilon}_{i,k}, \qquad (5-9)$$

and the matrices $\epsilon_{i,k}$ have the form

$$\boldsymbol{\epsilon}_{i,k} = a_{i,k} \boldsymbol{\epsilon}^{(k)}, \tag{5-10}$$

for some choice of constants $a_{i,k}$ which ensures they are symmetrized rank 1 matrices lying in the space \mathcal{V}_p (so they cost very little energy), and which ensures that the ϵ_i^0 given by (5-9) are orthogonal. This symmetrized rank 1 form ensures that $\underline{\epsilon}_i(\mathbf{x})$ derives from a displacement field. Specifically we have

$$\underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) = \frac{1}{2} \Big(\nabla \underline{\boldsymbol{u}}_{i}(\boldsymbol{x}) + (\nabla \underline{\boldsymbol{u}}_{i}(\boldsymbol{x}))^{T} \Big), \tag{5-11}$$

with

$$\underline{\boldsymbol{u}}_{i}(\boldsymbol{x}) = \sum_{k=1}^{p} a_{i,k} \boldsymbol{a}_{k} \big\{ (\boldsymbol{n}_{k} \cdot \boldsymbol{x}) \eta_{k}(\boldsymbol{x}) / \epsilon + ([\boldsymbol{n}_{k} \cdot \boldsymbol{x}] + 1)(1 - \eta_{k}(\boldsymbol{x})) \big\}, \quad (5-12)$$

where, as before, $[n_j \cdot x]$ is the greatest integer less than $n_j \cdot x$. One can easily check that this displacement field is continuous at the wall interfaces.

To find upper bounds on the energy associated with this trial strain field, first consider those parts of the walled structure that are outside of any junction regions, i.e., where for some k we have $\eta_k(\mathbf{x}) = 1$, while $\eta_s(\mathbf{x}) = 0$ for all $s \neq k$. An upper bound for the volume fraction occupied by the region where $\eta_k(\mathbf{x}) = 1$ while $\eta_s(\mathbf{x}) = 0$ for all $s \neq k$ is of course ϵ , as this represents the volume of the region

where $\eta_k(\mathbf{x}) = 1$. The associated energy per unit volume of the trial strain field in those parts of the walled structure that are outside of any junction regions is bounded above by

$$\sum_{k=1}^{p} \boldsymbol{\epsilon}_{i,k} : \boldsymbol{C}_{*}(\boldsymbol{\mathcal{V}}_{p}) \boldsymbol{\epsilon}_{i,k} / \boldsymbol{\epsilon}.$$
(5-13)

We will see in Section 8 that with an appropriate choice of multimode material, $\epsilon_{i,k} : C_*(\mathcal{V}_p)\epsilon_{i,k}$ is bounded above by a quantity proportional to δ , essentially because all the strain is concentrated in phase 2. So we require that the limits $\delta \to 0$ and $\epsilon \to 0$ be taken so that $\delta/\epsilon \to 0$ to ensure that the quantity (5-13) goes to zero in this limit.

Next, consider those junction regions where only two walls meet, i.e., where for some k_1 and $k_2 > k_1$, \mathbf{x} is such that $\eta_{k_1}(\mathbf{x}) = \eta_{k_2}(\mathbf{x}) = 1$ while $\eta_s(\mathbf{x}) = 0$ for all s not equal to k_1 or k_2 . Provided $\mathbf{n}_{k_1} \neq \mathbf{n}_{k_2}$, an upper bound for the volume fraction occupied by each such junction region is ϵ^2 . Then the associated energy per unit volume of the trial strain field in these junction regions where only two walls meet is bounded above by

$$\sum_{k_1=1}^{p} \sum_{k_2=k_1+1}^{p} (\boldsymbol{\epsilon}_{i,k_1} + \boldsymbol{\epsilon}_{i,k_2}) : \boldsymbol{C}_*(\boldsymbol{\mathcal{V}}_p)(\boldsymbol{\epsilon}_{i,k_1} + \boldsymbol{\epsilon}_{i,k_2}).$$
(5-14)

Thus, the powers of ϵ cancel and this energy density goes to zero if the multimode material is easily compliant to the strains $\epsilon_{i,k_1} + \epsilon_{i,k_2}$ for all k_1 and k_2 with $k_2 > k_1$.

Finally, consider those junction regions where three or more walls meet, i.e., for some $k_1, k_2 > k_1$, and $k_3 > k_2, \mathbf{x}$ is such that $\eta_{k_i}(\mathbf{x}) = 1$ for i = 1, 2, 3. For a given choice of $k_1, k_2 > k_1$, and $k_3 > k_2$ such that the three vectors $\mathbf{n}_{k_1}, \mathbf{n}_{k_2}$, and \mathbf{n}_{k_3} are not coplanar, an upper bound for the volume fraction occupied by this region is ϵ^3 . In the case that the three vectors $\mathbf{n}_{k_1}, \mathbf{n}_{k_2}$, and \mathbf{n}_{k_3} are coplanar, we can ensure that the volume fraction occupied by this region is ϵ^3 or less by appropriately translating one or two wall structures, i.e., by replacing $\eta_{k_m}(\mathbf{x})$ with $\eta_{k_m}(\mathbf{x} + \alpha_i \mathbf{n}_{k_m})$ for m = 2, 3, for an appropriate choice of α_2 and α_3 between 0 and 1. Since the energy density of the trial field in these regions scales as $\epsilon^3/\epsilon^2 = \epsilon$, we can ignore this contribution in the limit $\epsilon \to 0$ as it goes to zero too.

From this analysis of the energy densities associated with the trial fields it follows that one does not necessarily need the pentamode, quadramode, trimode, bimode, and unimode materials as appropriate for the material inside the walled structure. Instead, by modifying the construction, it suffices to use only unimode and bimode materials. In the walled structure we now put unimode materials in those sections where for some k we have $\eta_k(\mathbf{x}) = 1$ while $\eta_{k'}(\mathbf{x}) = 0$ for all $k' \neq k$. Each unimode material is easily compliant to the single strain $\boldsymbol{\epsilon}^{(k)}$ appropriate to the wall under consideration. A prescription for constructing 3-dimensional unimode

materials that are multiple rank laminates, and which are easily compliant under any desired single strain, is given in Section 5.1 of [Milton and Cherkaev 1995]. In each junction region of the walled structure where $\eta_{k_1}(\mathbf{x}) = \eta_{k_2}(\mathbf{x}) = 1$ for some $k_1 \neq k_2$ while $\eta_k(\mathbf{x}) = 0$ for all k not equal to k_1 or k_2 , we put a bimode material which is easily compliant to any strain in the subspace spanned by $\epsilon^{(k_1)}$ and $\epsilon^{(k_2)}$ as appropriate to the junction region under consideration. At present we do not know of any recipe in three dimensions for constructing bimode materials that have any desired pair of strains as their easy modes of deformation, other than to superimpose four pentamode structures as described in Section 8. In the remaining junction regions of the walled structure (where three or more walls intersect) we put phase 1. The contribution to the average energy of the fields in these regions vanishes as $\epsilon \rightarrow 0$ as discussed above.

By these constructions we effectively obtain materials with elasticity tensors C_* such that

$$\lim_{\delta \to 0} \boldsymbol{C}_* = (\boldsymbol{I} - \boldsymbol{\Pi}_p) \widetilde{\boldsymbol{C}}_f^A (\boldsymbol{I} - \boldsymbol{\Pi}_p), \qquad (5-15)$$

where I is the fourth-order identity matrix, Π_p is the fourth-order tensor that is the projection onto the space \mathcal{V}_p , $I - \Pi_p$ is the projection onto the orthogonal complement of \mathcal{V}_p , and \widetilde{C}_f^A is the relevant complementary Avellaneda material. In the basis (5-5) $I - \Pi_p$ is represented by the 6 × 6 matrix that has the block form

$$\boldsymbol{I} - \boldsymbol{\Pi}_p = \begin{pmatrix} \boldsymbol{I}_{6-p} & \boldsymbol{0} \\ \boldsymbol{0} & \boldsymbol{0} \end{pmatrix}, \tag{5-16}$$

where I_{6-p} represents the $(6-p) \times (6-p)$ identity matrix and the 0 on the diagonal represents the $p \times p$ zero matrix.

6. Simplifications for 2-dimensional printed materials

For 2-dimensional printed materials, or any 2-dimensional two-phase composite with one phase being void, the analysis simplifies as then the space of 2×2 symmetric matrices has dimension 3, so there are only four energy functions to consider:

$$W_{f}^{0}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\sigma}_{3}^{0}) = \min_{\boldsymbol{C}_{*}\in GU_{f}} \sum_{j=1}^{3} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}_{j}^{0},$$

$$W_{f}^{1}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\epsilon}_{1}^{0}) = \min_{\boldsymbol{C}_{*}\in GU_{f}} \left[\boldsymbol{\epsilon}_{1}^{0} : \boldsymbol{C}_{*}\boldsymbol{\epsilon}_{1}^{0} + \sum_{j=1}^{2} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}_{j}^{0}\right],$$

$$W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0}) = \min_{\boldsymbol{C}_{*}\in GU_{f}} \left[\left(\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*}\boldsymbol{\epsilon}_{i}^{0}\right) + \boldsymbol{\sigma}_{1}^{0} : \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}_{1}^{0}\right],$$

$$W_{f}^{3}(\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0}) = \min_{\boldsymbol{C}_{*}\in GU_{f}} \sum_{i=1}^{3} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*}\boldsymbol{\epsilon}_{i}^{0}.$$
(6-1)

Again $W_f^0(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0)$ is attained for a "complementary Avellaneda material" consisting of a sequentially layered laminate geometry having an effective tensor $\boldsymbol{C}_* = \widetilde{\boldsymbol{C}}_f^A(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0) \in GU_f$, and we have the inequalities

$$\sum_{j=1}^{2} \boldsymbol{\sigma}_{j}^{0} : [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, 0)]^{-1} \boldsymbol{\sigma}_{j}^{0} \leq W_{f}^{1}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\epsilon}_{1}^{0}),$$

$$\boldsymbol{\sigma}_{1}^{0} : [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, 0, 0)]^{-1} \boldsymbol{\sigma}_{1}^{0} \leq W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}),$$

$$0 \leq W_{f}^{3}(\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}),$$

(6-2)

where, as before, the last inequality is sharp in the limit $\delta \rightarrow 0$ being attained when the material consists of islands of phase 1 surrounded by a phase 2.

The recipe for showing that the bound (6-1) on $W_f^1(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\epsilon}_1^0)$ is sharp for certain values of $\boldsymbol{\epsilon}_1^0$ and that the bound (6-1) on $W_f^2(\boldsymbol{\sigma}_1^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0)$ is sharp for certain values of $\boldsymbol{\epsilon}_1^0$ and $\boldsymbol{\epsilon}_2^0$ is almost exactly the same as in the 3-dimensional case: insert into the complementary Avellaneda material a thin walled structure of respectively unimode and bimode materials so that slips can occur along these walls, allowing with very little energetic cost the average strain $\boldsymbol{\epsilon}_1^0$ in the case of W_f^1 , or any strain in the space spanned by $\boldsymbol{\epsilon}_1^0$ and $\boldsymbol{\epsilon}_2^0$ in the case of W_f^2 .

7. The algebraic problem: characterizing those symmetric matrix pencils spanned by symmetrized rank 1 matrices

We are interested in the following question: Given k linearly independent symmetric $d \times d$ matrices A_1, A_2, \ldots, A_k , find necessary and sufficient conditions such that there exist linearly independent matrices $\{B_i\}_{i=1}^k$ spanned by the basis elements A_i so that each matrix B_i is a symmetrized rank 1 matrix, i.e., there exist vectors a_i and b_i , with $|b_i| = 1$, such that

$$\boldsymbol{B}_i = \frac{1}{2} (\boldsymbol{b}_i \boldsymbol{a}_i^T + \boldsymbol{a}_i \boldsymbol{b}_i^T).$$

It is assumed that d = 2 or 3 and $1 \le k \le k_d$, where $k_2 = 2$ and $k_3 = 5$. Here we are working in the generic situation, i.e., we prove the algebraic result for a dense set of matrices. The continuity result of Section 9 will allow us to conclude for the whole set of matrices. Actually, the proof below also shows that the algebraic result holds for the complement of a zero measure set of matrices.

Theorem 7.1. The above problem is solvable if and only if the matrices A_i for i = 1, ..., k satisfy the following conditions:

(i)
$$\det(A_1) \le 0$$
, if $k = 1, d = 2$, (7-1)

 A_1 has two eigenvalues of opposite signs and one zero eigenvalue,

or has two zero eigenvalues, if k = 1, d = 3. (7-2)
(ii) If k = d = 2,

$$\det(\boldsymbol{A}_1) < 0$$

or

 $f(t) = \det(A_1 + tA_2) \text{ is quadratic and has two distinct roots for } t,$ or is linear in t with a nonzero coefficient of t. (7-3)

(iii) If
$$k = 2$$
 and $d = 3$, defining $A(\eta, \mu) = \eta A_1 + \mu A_2$, the numbers

$$\det(\boldsymbol{A}(\eta,\mu)), \quad \{\boldsymbol{A}(\eta,\mu)\}_{11}\{\boldsymbol{A}(\eta,\mu)\}_{22} - \{\boldsymbol{A}(\eta,\mu)\}_{12}^2, \quad \{\boldsymbol{A}(\eta,\mu)\}_{11} \quad (7-4)$$

are never simultaneously nonnegative for any choice of η and μ not both zero (equivalently $\mathbf{A}(\eta, \mu)$ is never strictly positive definite for any values of η and μ), and

$$\Delta = 18 \det(\mathbf{A}_1) \det(\mathbf{A}_2) S_1 S_2 - 4S_1^3 \det(\mathbf{A}_2) + S_1^2 S_2^2 - 4S_2^3 \det(\mathbf{A}_1) - 27 \det(\mathbf{A}_1)^2 \det(\mathbf{A}_2)^2 > 0, \quad (7-5)$$

where $S_i = \sum_{j=1}^{3} s_{ij}$ for i = 1, 2 and s_{ij} is the determinant of the matrix obtained by replacing the *j*-th row of A_i by the *j*-th row of A_{i+1} , where by convention we have $A_3 = A_1$ (equivalently $A(\eta, \mu)$ has three distinct roots).

(iv) Always solvable if $k \ge 3$, d = 3. (7-6)

Remark. In fact, the condition (7-2) and the last condition in (7-3), that f(t) is linear in t, could be withdrawn since we are considering the generic case. They are inserted because we can treat them explicitly.

Proof. Case (i): k = 1, d = 2 or 3. In this case A_1 must be a multiple of B_1 and hence must be a symmetrized rank 1 matrix. To see more clearly the condition for a matrix **B** to be a symmetrized rank 1 matrix, i.e., have the form $B = \frac{1}{2}(ba^T + ab^T)$, let us, without loss of generality, choose our coordinates so that $b = [1, 0]^T$ when d = 2 and $b = [1, 0, 0]^T$ when d = 3. Then **B** has the representation

$$\boldsymbol{B} = \begin{pmatrix} a_1 & \frac{1}{2}a_2 \\ \frac{1}{2}a_2 & 0 \end{pmatrix} \text{ when } d = 2, \quad \boldsymbol{B} = \begin{pmatrix} a_1 & \frac{1}{2}a_2 & \frac{1}{2}a_3 \\ \frac{1}{2}a_2 & 0 & 0 \\ \frac{1}{2}a_3 & 0 & 0 \end{pmatrix} \text{ when } d = 3.$$
 (7-7)

These have eigenvalues

$$\lambda = \frac{1}{2} \left(a_1 \pm \sqrt{a_1^2 + a_2^2} \right) \text{ when } d = 2,$$

$$\lambda = \frac{1}{2} \left(a_1 \pm \sqrt{a_1^2 + a_2^2 + a_3^2} \right) \text{ and } \lambda = 0 \text{ when } d = 3.$$
(7-8)

So, clearly **B** is a symmetrized rank 1 matrix in two dimensions if and only if $det(B) \le 0$, and is a symmetrized rank 1 matrix in three dimensions if and only if

it has two eigenvalues of opposite signs and one zero eigenvalue, or has two zero eigenvalues.

Case (ii): k = 2, d = 2. In this case there should be two distinct values of t such that det $(A_1 + tA_2) < 0$, which by continuity of this determinant as a function of t is guaranteed if any of the conditions in (7-3) are met. Note that the case where det $(A_1 + tA_2) = 0$ for all t can be ruled out from consideration since this can only happen when A_2 is proportional to A_1 , as can be easily seen by working in a basis where A_2 is diagonal.

Case (iii): k = 2, d = 3. Consider the matrix pencil (over reals η and μ) $A(\eta, \mu) = \eta A_1 + \mu A_2$. Assuming that det $A(\eta, \mu)$ is not zero for all η and μ , there are at least two matrices on the pencil which have nonzero determinant. Let us relabel them as A_1 and A_2 . Then the equation det $(A(1, \mu)) = 0$ must have either two or three roots $\mu = z_i$ for i = 1, 2 or i = 1, 2, 3, where the z_i are obtained by changing the sign of the generalized eigenvalues. This gives Cardan's condition:

$$\Delta = 18 \det(A_1) \det(A_2) S_1 S_2 - 4S_1^3 \det(A_2) + S_1^2 S_2^2 - 4S_2^3 \det(A_1) - 27 \det(A_1)^2 \det(A_2)^2 \ge 0. \quad (7-9)$$

Suppose that $A_1 + \mu A_2$ contains a symmetric matrix with two zero eigenvalues (a rank 1 matrix) as μ is varied. Then by redefining A_2 we can assume A_2 is this matrix, now with zero determinant, and by using a basis where A_2 is diagonal, we see that det $(A_1 + \mu A_2)$ depends linearly on μ and det $(A_1 + \mu A_2)$ can only have one root: (7-9) must be violated. So we can exclude this possibility: $A_1 + \mu A_2$ has at most one zero eigenvalue for any value of μ . Now consider the eigenvalues of $A(\theta) \equiv A(\cos \theta, \sin \theta)$ as θ is varied. As $A(-\theta) = -A(\theta)$ it suffices to consider the interval of θ between 0 and π . Some scenarios for the eigenvalue trajectories are plotted in Figure 5. At the values $\theta_i = \arctan^{-1}(z_i)$ at least one of the eigenvalues must be zero, and the favorable situation is when there are two remaining eigenvalues of opposite signs or only one nonzero eigenvalue. Such angles θ_i are marked by the vertical dashed lines in the figure. The unfavorable situation is when there are two nonzero eigenvalues of the same sign, marked by the red vertical lines in Figure 5 (left). First suppose that $A(\theta)$ is positive definite for some $\theta = \theta_0$. By refining θ as the old θ minus θ_0 , let us suppose A(0) is positive definite. Then the scenario is that in Figure 5 (left), or some variant of it in which eigenvalues cross, which is unfavorable. The only way to avoid this is for $A(\theta)$ to have two zero eigenvalues at the smallest and largest values of $\theta \in [0, \pi]$ for which det $A(\theta) = 0$, as in Figure 5 (middle), but we have ruled out the possibility that $A(\theta)$ has two zero eigenvalues for any value of θ . We are left with Figure 5 (right) as being the only possible suitable scenario. In conclusion, we require that the matrix $A(\theta)$ not be positive semidefinite for any choice of θ ; i.e., the three quantities



Figure 5. Some scenarios for the eigenvalues λ of $A(\theta) = \cos \theta A_1 + \sin \theta A_2$ as θ is varied.

$$\det(A(\eta,\mu)), \quad \{A(\eta,\mu)\}_{11}\{A(\eta,\mu)\}_{22} - \{A(\eta,\mu)\}_{12}^2, \quad \{A(\eta,\mu)\}_{11} \quad (7-10)$$

are never simultaneously nonnegative for any choice of η and μ not both zero. This condition could be made explicit by using the formula for the roots of a cubic to determine the generalized eigenvalues $-z_i$.

Case (iv): $k \ge 3$, d = 3. The case k = 3 is a straightforward consequence of Lemma 7.2 below.

It remains to consider $k \ge 4$ and d = 3. By the previous step, in the space spanned by A_1 , A_2 , and A_3 there are three matrices A'_1 , A'_2 , and $B_3 = A_3 + \eta_3 A'_1 + \mu_3 A'_2$ that are linearly independent, symmetrized and of rank 1. Then, again by the previous step, we can find linearly independent matrices B_1, \ldots, B_k that have the form $B_1 = A'_1$, $B_2 = A'_2$, and $B_i = A_i + \eta_i A'_1 + \mu_i A'_2$ for $3 \le i \le k$ and that are of rank 1.

In the sequel we write

$$\boldsymbol{a} \otimes \boldsymbol{b} := \boldsymbol{a} \boldsymbol{b}^T$$
 and $\boldsymbol{a} \odot \boldsymbol{b} := \frac{1}{2} (\boldsymbol{a} \otimes \boldsymbol{b} + \boldsymbol{b} \otimes \boldsymbol{a})$ for $\boldsymbol{a}, \boldsymbol{b} \in \mathbb{R}^3$. (7-11)

Lemma 7.2. Let A, B, C be three symmetric matrices of $\mathbb{R}^{3\times 3}$.

(i) Up to small perturbations of A, B, C, there exist a basis (x, y, z) of \mathbb{R}^3 and three vectors a, b, c of \mathbb{R}^3 satisfying

$$\begin{cases} \boldsymbol{a} \in \{\boldsymbol{A}\boldsymbol{x}, \boldsymbol{B}\boldsymbol{x}, \boldsymbol{C}\boldsymbol{x}\}^{\perp} \setminus \{\boldsymbol{0}\}, \\ \boldsymbol{b} \in \{\boldsymbol{A}\boldsymbol{y}, \boldsymbol{B}\boldsymbol{y}, \boldsymbol{C}\boldsymbol{y}\}^{\perp} \setminus \{\boldsymbol{0}\}, \\ \boldsymbol{c} \in \{\boldsymbol{A}\boldsymbol{z}, \boldsymbol{B}\boldsymbol{z}, \boldsymbol{C}\boldsymbol{z}\}^{\perp} \setminus \{\boldsymbol{0}\}, \end{cases}$$
(7-12)

or equivalently,

$$a \odot x, b \odot y, c \odot z \in \{A, B, C\}^{\perp} \setminus \{0\}.$$
 (7-13)

(ii) Up to small perturbations of A, B, C, there exist three independent symmetrized rank 1 matrices in the space $\{A, B, C\}^{\perp}$.

Proof. (i) Let F be the cubic function defined by

$$F(\mathbf{x}) := \det(A\mathbf{x}, B\mathbf{x}, C\mathbf{x}) \quad \text{for } \mathbf{x} \in \mathbb{R}^3.$$
(7-14)

If $F \equiv 0$ in \mathbb{R}^3 , then condition (7-12) is immediately satisfied. Otherwise, there exists a basis $(\mathbf{x}_0, \mathbf{u}_0, \mathbf{v}_0)$ of \mathbb{R}^3 in the nonempty open set $\{F \neq 0\}$. Since we have

$$F(\boldsymbol{x}_0 + s\boldsymbol{u}_0) \underset{|s| \to \infty}{\sim} s^3 F(\boldsymbol{u}_0) \underset{|s| \to \infty}{\longrightarrow} \pm \infty,$$
(7-15)

there exists $s, t \in \mathbb{R} \setminus \{0\}$ such that $\mathbf{x} := \mathbf{x}_0 + s\mathbf{u}_0$ and $\mathbf{y} := \mathbf{x}_0 + t\mathbf{v}_0$ are two independent vectors in the set $\{F = 0\}$.

First, assume that the set $\{F = 0\}$ is not contained in the plane Span $\{x, y\}$. Then there exists a basis (x, y, z) of \mathbb{R}^3 in the set $\{F = 0\}$. Therefore, there exist three vectors a, b, c of \mathbb{R}^3 satisfying (7-12), or equivalently (7-13).

Now, assume that $\{F = 0\} \subset \text{Span}\{x, y\}$. First of all, up to small perturbations we can assume that the matrices A, B, C are invertible. Since $B^{-1}C$ is a 3×3 real matrix, it has at least a real eigenvalue λ . The perturbation procedure is now divided into two cases.

First case: The matrix $B^{-1}C$ has two complex conjugate eigenvalues.

Then the eigenspace Ker($B^{-1}C - \lambda I_3$) is a line of \mathbb{R}^3 spanned by $e \in \mathbb{R}^3 \setminus \{0\}$. Consider a basis (x_0, u_0, v_0) of \mathbb{R}^3 in the set $\{F \neq 0\}$ such that (e, x_0, u_0) and (e, x_0, v_0) are also two bases of \mathbb{R}^3 . As previously there exist $s, t \in \mathbb{R} \setminus \{0\}$ such that $x := x_0 + su_0$ and $y := x_0 + tv_0$ are two independent vectors of the set $\{F = 0\}$. Moreover, since (e, x) and (e, y) are two families of independent vectors and $\mathbb{R}e$ is the unique real eigenspace of the matrix $B^{-1}C$, we have

$$Bx \times Cx \neq 0$$
 and $By \times Cy \neq 0$. (7-16)

Now, consider a vector $u \in \{x, y\}^{\perp} \setminus \{0\}$ and the matrix $M \in \mathbb{R}^{3 \times 3}$ defined by

$$Mx = \xi, \quad My = \eta, \quad Mu = 0, \tag{7-17}$$

where the vectors $\boldsymbol{\xi}$, $\boldsymbol{\eta}$ will be chosen later. Define for $\tau > 0$ the perturbed function

$$F_{\tau}(z) := \det(Az + \tau Mz, Bz, Cz) \quad \text{for } z \in \mathbb{R}^3.$$
(7-18)

We have

$$\begin{cases} F_{\tau}(\boldsymbol{x} + \tau \boldsymbol{u}) = \tau \,\boldsymbol{\xi} \cdot \left(\boldsymbol{B}\boldsymbol{x} \times \boldsymbol{C}\boldsymbol{x} + \boldsymbol{O}(\tau)\right) + O(\tau), \\ F_{\tau}(\boldsymbol{y} + \tau \boldsymbol{u}) = \tau \,\boldsymbol{\eta} \cdot \left(\boldsymbol{B}\boldsymbol{y} \times \boldsymbol{C}\boldsymbol{y} + \boldsymbol{O}(\tau)\right) + O(\tau), \end{cases}$$
(7-19)

where the $O(\tau)$ denote some first-order vectors in τ and $O(\tau)$ some first-order real numbers in τ which are independent of ξ , η . Condition (7-16) then allows us to choose $\xi = \xi_{\tau}$ and $\eta = \eta_{\tau}$ such that $F_{\tau}(x + \tau u) = F_{\tau}(y + \tau u) = 0$. Therefore, since (x, y, u) is a basis of \mathbb{R}^3 , $(x, x + \tau u, y + \tau u)$ is also a basis of \mathbb{R}^3 , which in addition lies in the set $\{F_{\tau} = 0\}$. This leads us to condition (7-12) with the matrices $A + \tau M$, B, C. Second case: The matrix $B^{-1}C$ has only real eigenvalues.

Then there exists a small perturbation C_{τ} of C such that the perturbed matrix $B^{-1}C_{\tau}$ has three distinct real eigenvalues. Hence, the matrix $B^{-1}C_{\tau}$ admits a basis (x, y, z) of eigenvectors, which implies that

$$C_{\tau} x - \lambda B x = C_{\tau} y - \lambda B y = C_{\tau} z - \lambda B z = 0.$$
(7-20)

Therefore, the perturbed function

$$F_{\tau}(\boldsymbol{u}) := \det(\boldsymbol{A}\boldsymbol{u}, \boldsymbol{B}\boldsymbol{u}, \boldsymbol{C}_{\tau}\boldsymbol{u}) \quad \text{for } \boldsymbol{u} \in \mathbb{R}^3$$
(7-21)

satisfies $F_{\tau}(\mathbf{x}) = F_{\tau}(\mathbf{y}) = F_{\tau}(\mathbf{z}) = 0$, which again leads us to condition (7-12) with the matrices A, B, C_{τ} .

(ii) We will distinguish four cases according to whether the following conditions are satisfied by the basis (x, y, z) of \mathbb{R}^3 and the vectors $a, b, c \in \mathbb{R}^3 \setminus \{0\}$ obtained in step (i):

$$\begin{cases} a \in \operatorname{Span}\{x, y\} \cap \operatorname{Span}\{x, z\}, \\ b \in \operatorname{Span}\{y, x\} \cap \operatorname{Span}\{y, z\}, \\ c \in \operatorname{Span}\{z, x\} \cap \operatorname{Span}\{z, y\}. \end{cases}$$
(7-22)

First case: *a*, *b* and *c* satisfy conditions (7-22).

Then, since (x, y, z) is a basis of \mathbb{R}^3 , we have necessarily $a \in \mathbb{R}x$, $b \in \mathbb{R}y$, $c \in \mathbb{R}z$. Therefore, $x \odot x$, $y \odot y$, $z \odot z$ are clearly three independent matrices of $\{A, B, C\}^{\perp}$.

Second case: b and c satisfy conditions (7-22) but a does not.

Then, for example, (a, x, y) is a basis of \mathbb{R}^3 , and $b \in \mathbb{R}y$, $c \in \mathbb{R}z$. Let $u \in \{y, z\}^{\perp} \setminus \{0\}$, and let α , β , $\gamma \in \mathbb{R}$ be such that $\alpha a \odot x + \beta y \odot y + \gamma z \odot z = 0$. Multiplying by u we get that $\alpha(x \cdot u)a + \alpha(a \cdot u)x = 0$; hence $\alpha = 0$ since $x \cdot u \neq 0$. We deduce immediately that $\beta = \gamma = 0$. Therefore, $a \odot x$, $y \odot y$, $z \odot z$ are three independent matrices of $\{A, B, C\}^{\perp}$.

Third case: **a** and **b** do not satisfy conditions (7-22), with $a \notin \text{Span}\{x, y\}$ and $b \notin \mathbb{R}a \cup \mathbb{R}x$ (respectively $a \notin \text{Span}\{x, z\}$ and $c \notin \mathbb{R}a \cup \mathbb{R}x$).

Then (a, x, y) is a basis of \mathbb{R}^3 . Let $u \in \{x, y\}^{\perp} \setminus \{0\}$, and let $\alpha, \beta \in \mathbb{R}$ be such that $\alpha a \odot x + \beta b \odot y = 0$. Multiplying by u we get that $\alpha(a \cdot u)x + \beta(b \cdot u)y = 0$; hence $\alpha = 0$ since $a \cdot u \neq 0$, and thus $\beta = 0$. Therefore, $a \odot x, b \odot y$ are two independent matrices of $\{A, B, C\}^{\perp}$, which have two eigenvalues of opposite sign and one 0 eigenvalue.

Let us prove by contradiction that

$$\exists t \in \mathbb{R} \setminus \{0\}, \quad \det(\boldsymbol{a} \odot \boldsymbol{x} + t \, \boldsymbol{b} \odot \boldsymbol{y}) \neq 0. \tag{7-23}$$

Otherwise, for any $t \neq 0$, there exists $z_t \in \text{Ker}(a \odot x + t b \odot y) \setminus \{0\}$; hence

$$(\mathbf{x} \cdot \mathbf{z}_t)\mathbf{a} + (\mathbf{a} \cdot \mathbf{z}_t)\mathbf{x} + t(\mathbf{y} \cdot \mathbf{z}_t)\mathbf{b} + t(\mathbf{b} \cdot \mathbf{z}_t)\mathbf{y} = \mathbf{0}.$$
 (7-24)

Since (a, x, y) is a basis of \mathbb{R}^3 and $z_t \neq 0$, we have necessarily $y \cdot z_t \neq 0$, which implies that

$$-\boldsymbol{b} = \frac{\boldsymbol{x} \cdot \boldsymbol{z}_t}{t(\boldsymbol{y} \cdot \boldsymbol{z}_t)} \boldsymbol{a} + \frac{\boldsymbol{a} \cdot \boldsymbol{z}_t}{t(\boldsymbol{y} \cdot \boldsymbol{z}_t)} \boldsymbol{x} + \frac{\boldsymbol{b} \cdot \boldsymbol{z}_t}{\boldsymbol{y} \cdot \boldsymbol{z}_t} \boldsymbol{y} = \alpha \, \boldsymbol{a} + \beta \, \boldsymbol{x} + \gamma \, \boldsymbol{y}, \tag{7-25}$$

where α , β , γ are independent of *t*, and

$$(\boldsymbol{x} - \alpha t \, \boldsymbol{y}) \cdot \boldsymbol{z}_t = (\boldsymbol{a} - \beta t \, \boldsymbol{y}) \cdot \boldsymbol{z}_t = (\boldsymbol{b} - \gamma \, \boldsymbol{y}) \cdot \boldsymbol{z}_t = 0. \tag{7-26}$$

Since $z_t \neq \mathbf{0}$ there exists $(p_t, q_t, r_t) \in \mathbb{R}^3 \setminus \{0\}$ such that

$$p_t(\mathbf{x} - \alpha t \, \mathbf{y}) + q_t(\mathbf{a} - \beta t \, \mathbf{y}) + r_t(\mathbf{b} - \gamma \, \mathbf{y})$$

= $(q_t - \alpha r_t)\mathbf{a} + (p_t - \beta r_t)\mathbf{x} - (\alpha t p_t + \beta t q_t + 2\gamma r_t)\mathbf{y} = \mathbf{0}, \quad (7-27)$

which implies that $q_t = \alpha r_t$, $p_t = \beta r_t$ and $r_t(\alpha \beta t + \gamma) = 0$. Since $(p_t, q_t, r_t) \neq \mathbf{0}$, we have $r_t \neq 0$ and $\alpha \beta t + \gamma = 0$ for any $t \neq 0$; hence $\alpha \beta = 0$ and $\gamma = 0$. This yields a contradiction between (7-25) and $\mathbf{b} \notin \mathbb{R}\mathbf{a} \cup \mathbb{R}\mathbf{x}$.

By virtue of (7-23) there exist two nonzero real numbers $\alpha \neq \beta$ such that the matrices

$$M := a \odot x + \alpha b \odot y \quad \text{and} \quad N := a \odot x + \beta b \odot y \tag{7-28}$$

are invertible. The function $p(t) := \det(\beta M - tN)$ is a polynomial of degree 3 whose α , β are two distinct roots. Then the polynomial p(t) must change sign by crossing α , for example (the conclusion is similar for β). Let $\lambda_1(t) \le \lambda_2(t) \le \lambda_3(t)$ be the well-ordered eigenvalues of the symmetric matrix $\beta M - tN$. Since the vectors a, x are independent, $a \odot x$ has two eigenvalues of opposite sign and one 0 eigenvalue; hence $\lambda_1(\alpha) < \lambda_2(\alpha) = 0 < \lambda_3(\alpha)$.

Now, let P_{τ} for a small $\tau > 0$ be a symmetric matrix in the space $\{A, B, C\}^{\perp}$, such that $|P_{\tau} - a \odot x| = O(\tau)$, and such that the three matrices $a \odot x, b \odot y, P_{\tau}$ are independent (note that the dimension of $\{A, B, C\}^{\perp}$ is ≥ 3). Define the two perturbed matrices

$$\boldsymbol{M}_{\tau} := \boldsymbol{P}_{\tau} + \alpha \, \boldsymbol{b} \odot \, \boldsymbol{y} \quad \text{and} \quad \boldsymbol{N}_{\tau} := \boldsymbol{P}_{\tau} + \beta \, \boldsymbol{b} \odot \, \boldsymbol{y}.$$
 (7-29)

Since the well-ordered eigenvalues of a real symmetric matrix *S* are Lipschitzcontinuous with respect to *S* (see, e.g., [Ciarlet 1989], Theorem 2.3-2), the eigenvalues $\lambda_1^{\tau}(t) \le \lambda_2^{\tau}(t) \le \lambda_3^{\tau}(t)$ of $\beta M_{\tau} - tN_{\tau}$ converge uniformly as $\tau \to 0$ to the eigenvalues $\lambda_1(t) \le \lambda_2(t) \le \lambda_3(t)$ of $\beta M - tN$, with respect to *t* in a neighborhood of α . Hence, for $\tau > 0$ small enough, there exist α_{τ} close to α such that $\alpha_{\tau} \neq \beta$ and

$$\lambda_1^{\tau}(\alpha_{\tau}) < \lambda_2^{\tau}(\alpha_{\tau}) = 0 < \lambda_3^{\tau}(\alpha_{\tau}). \text{ Then by (7-29) there exist } \boldsymbol{c}_{\tau}, \boldsymbol{z}_{\tau} \in \mathbb{R}^3 \text{ such that}$$
$$\beta \boldsymbol{M}_{\tau} - \alpha_{\tau} \boldsymbol{N}_{\tau} = \boldsymbol{c}_{\tau} \odot \boldsymbol{z}_{\tau} = (\beta - \alpha_{\tau}) \boldsymbol{P}_{\tau} + \beta(\alpha - \alpha_{\tau}) \boldsymbol{b} \odot \boldsymbol{y}, \text{ with } \beta - \alpha_{\tau} \neq 0.$$
(7-30)

Therefore, $a \odot x$, $b \odot y$, $c_{\tau} \odot z_{\tau}$ are three independent symmetrized rank 1 matrices in the space $\{A, B, C\}^{\perp}$.

Fourth case: *a* and *b* do not satisfy conditions (7-22), with $a \notin \text{Span}\{x, y\}$ and $b \in \mathbb{R}a \cup \mathbb{R}x$ (respectively $a \notin \text{Span}\{x, z\}$ and $c \in \mathbb{R}a \cup \mathbb{R}x$).

For example, we have $b \in \mathbb{R}a$. We thus start from the matrices $a \odot x$ and $a \odot y$ in the space $\{A, B, C\}^{\perp}$, where (a, x, y) is a basis of \mathbb{R}^3 . We will consider a perturbation of A, B, C for leading us to the third case.

Let $t \in \{a, x\}^{\perp} \setminus \{0\}$, let $d \in \mathbb{R}^3 \setminus (\mathbb{R}a + \mathbb{R}x)$, and consider, for a small $\tau > 0$, the perturbed vector $b_{\tau} := a + \tau d \notin \mathbb{R}a \cup \mathbb{R}x$ and the perturbed matrices

$$A_{\tau} := A + \tau t \odot u_{\tau}, \quad B_{\tau} := B + \tau t \odot v_{\tau}, \quad C_{\tau} := C + \tau t \odot w_{\tau}, \quad (7-31)$$

where the vectors u_{τ} , v_{τ} , w_{τ} will be chosen later. Clearly, $a \odot x \in \{A_{\tau}, B_{\tau}, C_{\tau}\}^{\perp}$. On the other hand, we have

$$A_{\tau}: \boldsymbol{b}_{\tau} \odot \boldsymbol{y} = \tau(\boldsymbol{A}: \boldsymbol{d} \odot \boldsymbol{y} + \boldsymbol{t} \odot \boldsymbol{u}_{\tau}: \boldsymbol{a} \odot \boldsymbol{y} + \tau \boldsymbol{t} \odot \boldsymbol{u}_{\tau}: \boldsymbol{d} \odot \boldsymbol{y}).$$
(7-32)

Since $2t \odot u_{\tau} : a \odot y = (t \cdot y)a \cdot u_{\tau}$ with $t \cdot y \neq 0$, we can choose $u_{\tau} = O(1)$ with respect to τ such that $A_{\tau} : b_{\tau} \odot y = 0$. Hence, choosing v_{τ} and w_{τ} similarly, we get that $b_{\tau} \odot y \in \{A_{\tau}, B_{\tau}, C_{\tau}\}^{\perp}$. Therefore, the vectors a, b_{τ}, x, y satisfy the conditions of the third case with the perturbed matrices $A_{\tau}, B_{\tau}, C_{\tau}$.

8. Constructing suitable multimode materials for the wall microstructure

Let us specify the construction of the desired multimode materials in two dimensions and then move to three dimensions. We begin by constructing bimode materials that can only support one stress. One could use the fourth-rank laminate structure described in detail in Section 30.7 of [Milton 2002]. The analysis would then be essentially a repeat of that analysis, which builds the appropriate trial stress and strain fields at each length scale. The key feature is that these trial fields need to be chosen so the trial stress associated with the average stress σ^0 we want to achieve at the macroscopic scale is concentrated entirely in phase 1 (apart from boundary layers that we ignore, whose contribution to the energy vanishes in the homogenization limit), and so the trial strain associated with an average strain that is orthogonal to σ^0 is concentrated entirely in phase 2.

Rather than doing this, it is more instructive to build trial stress and strain fields that are concentrated in phase 1 and phase 2, respectively, for the honeycomb and inverted honeycomb bimode structures of Figure 6, as the ideas here carry over to pentamode materials. The trial stress is easy. It is taken to be macroscopically



Figure 6. 2-dimensional bimode materials that can only support one average stress field σ^0 , and which are easily compliant to any strain orthogonal to σ^0 . Here the red struts are laminates of the two phases with the interfaces in the laminate parallel to the direction of the struts. The geometry on the left is appropriate if det $\sigma^0 > 0$, the geometry on the right is appropriate if det $\sigma^0 < 0$, and if det $\sigma^0 = 0$ it suffices to use a simple laminate with the layer surfaces perpendicular to the null vector of σ^0 .

constant with a value $\alpha_i a_i a_i^T$ in each strut which is parallel to the unit vector a_i in Figure 7. Let w_i denote the width of the strut parallel to a_i , for i = 0, 1, 2. Since the net "force" on the black junction regions in the top left and top right of Figure 7 must be zero, we obtain

$$0 = -\sum_{k=0}^{2} w_i(\alpha_i \boldsymbol{a}_i \boldsymbol{a}_i^T) \boldsymbol{a}_i = -\sum_{k=0}^{2} w_i \alpha_i \boldsymbol{a}_i.$$
(8-1)

Since $w_1 = w_2$ and a_0 points in the horizontal direction, while a_1 and a_2 have the same horizontal component and equal but opposite vertical components, we get

$$\alpha_1 = \alpha_2 = -w_0 \alpha_0 / [2w_1(a_1 \cdot a_2)].$$
(8-2)

The symmetry of the trial stress field implies there is no associated torque acting on the junction regions. The trial stress in the junction regions is really not that important. One choice is the stress field that satisfies the elasticity equations appropriate to phase 1 filling the junction region when constant tractions act on the three sides. The average value of the trial stress does not depend on the choice of trial stress in the junctions. Indeed, since $\nabla \cdot (\sigma) = 0$ it follows from integration by parts of $\nabla \cdot (\sigma x)$ (where σx is a third-order tensor) that

$$\int_{\Omega} \boldsymbol{\sigma} \, d\boldsymbol{x} = \int_{\partial \Omega} \boldsymbol{t} \boldsymbol{x}^T \, dS, \quad \text{where } \boldsymbol{t} = \boldsymbol{\sigma} \boldsymbol{n} \text{ is the surface traction}, \tag{8-3}$$

in which Ω is any region with boundary $\partial \Omega$. For example, the boundary of Ω could be the outermost boundary of the shape in the top left or top right of Figure 7, where we include the dashed lines as part of the boundary.



Figure 7. The honeycomb structure of Figure 6 (left) can be taken to have the unit cell shown at top left. Similarly the inverted honeycomb structure of Figure 6 (right) can be taken to have the unit cell shown at top right. The space outside the struts and junction regions (which is occupied by phase 2) has been triangulated with boundaries marked by the dashed lines to make the construction of the trial stress fields easy.

In passing, we remark that if σ^0 is proportional to the identity matrix, then the microstructure of Figure 7 (top left) resembles a Sigmund microstructure (see the last subfigure in Figure 2 in [Sigmund 2000]). However, we do not require the tuning of layer widths in the struts that makes his structure optimal. Suboptimal structures are perfectly fine in the walls, since the walls ultimately occupy a vanishingly small volume fraction in the final material.

To obtain a trial easy strain it suffices to specify the trial displacement in the unit cell. We only choose motions so the junction regions (triangular in Figure 7 (bottom left) and quadrilateral in Figure 7 (bottom right)) undergo rigid body translations, so there is no strain inside them. Thus associated with Figure 7 (bottom left) one can clearly identify two independent macroscopic modes of motion. The first is where the line RS moves vertically upwards while the line PU remains fixed, and Q and T move in such a way that the lengths QR, QP, TS, and TU remain equal and preserved in length. One can choose the displacement to be linear in each of the three regions *A*, *B*, and *C* so that it matches the displacement on the boundary. The second is where the line RS moves horizontally while the line PU remains fixed, and Q and T move in such a way that the lengths QR, QP, TS, and TU remain equal and preserved in length. In either case inside the horizontal laminate arm there is no



Figure 8. 2-dimensional unimode materials that are easily compliant to one average strain field ϵ^0 , and which can support any stress orthogonal to ϵ^0 . In both, the red region represents a laminate as indicated by the inserts. The second-rank laminate geometry on the left is appropriate if det $\epsilon^0 < 0$ and the third-rank geometry on the right is appropriate for any ϵ^0 .

strain, while inside the inclined laminate arms there is an infinitesimal shear so the junction at P remains fixed, while the junction at Q moves perpendicular to a_1 and the junction at V moves perpendicular to a_2 . We also note that there is also an easy microscopic motion which results in no macroscopic motion. Define the center of each triangular junction to be the point which is at the junction of the perpendicular bisector of the three faces. Then if all the triangular junctions undergo the same infinitesimal rotation about these centers while the laminate material in the struts shears at the same time, it will cost very little energy. The trial strain field is bounded and nonzero only in phase 2, and therefore the associated upper bound on the elastic energy scales in proportion to δ .

The situation in Figure 7 (bottom right) is basically similar. The two black quadrilateral junction regions at the bottom of the figure can remain fixed. Then one mode is the symmetric one, where the region A undergoes uniaxial compression in the horizontal direction and at the same time moves downwards. The second is where the region A undergoes pure shear, so the junction on the left side of it moves up, while the right side moves down. The strain field can be taken constant in the regions A, B, C, D, and E, and in the inclined laminate strut arms is also constant and corresponds to pure shear. These strains are easily determined from the value of the trial displacement field at the boundaries of each region. Again, the trial strain field is bounded and nonzero only in phase 2, and therefore the associated upper bound on the elastic energy scales in proportion to δ .

The structures of Figure 8 give suitable 2-dimensional unimode materials. We will not specify the appropriate trial stress and strain fields which prove that these structures have the desired elastic behavior, as they are exactly the same as those given in Section 30.6 of [Milton 2002].

We now describe the pentamodes and the trial fields in them. Given four vectors a_0, a_1, a_2 , and a_4 (no longer required to be unit vectors) we position a point P at



Figure 9. The procedure for constructing the desired pentamodes. In (d) a shearable section is inserted into each strut. This section, shown in red, has the structure of parallel square fibers, as illustrated in Figure 10, with the fibers aligned parallel to the strut.

the origin, and join P to the four points $x = a_i$ for i = 0, 1, 2, 3, with four infinitesimally thin rods, as in Figure 9(a). We then take as our unit cell of periodicity the parallelepiped with the eight points $x = a_i$, $x = a_1 + a_2 + a_3 - a_i - a_0$ for i = 0, 1, 2, 3 (the three vectors $v_i = a_i - a_0$ for i = 1, 2, 3 are the primitive lattice vectors). We require that a_0, a_1, a_2, a_4 be chosen so P lies within this parallelepiped. After periodically extending the rod structure (with rods joining $k_1 v_1 + k_2 v_1 + k_3 v_1$ with the four points $k_1 v_1 + k_2 v_1 + k_3 v_1 + a_i$ for i = 0, 1, 2, 3, 3, 3for any integers k_1 , k_2 , and k_3), we then coat this periodic rod structure with phase 1, as illustrated in Figure 9(b), so that any point x is in phase 1 if and only if it is within a distance r of the rod structure. Here r should be chosen appropriately small so that the coatings of each rod contain a cylindrical section that we refer to as a strut. Figure 9(b) is misleading as it suggests that the unit cell only contains one junction region. The true structure which should be periodically repeated (by making copies shifted by vectors $k_1 v_1 + k_2 v_1 + k_3 v_1$ for all combinations of integers k_1, k_2 , and k_3 is shown in Figure 9(c) and contains the junction of Figure 9(b) plus the one obtained by inverting it under the transformation $x \to -x$. The final step, illustrated in Figure 9(d), is to take a cylindrical subsection of each cylindrical section between junctions and replace it with a pentamode material that supports any stress proportional to $a_i a_i^T$. It is convenient to take end faces of the cylindrical subsection to be perpendicular to the cylinder axis, i.e., perpendicular to the vector a_i that is parallel to the cylinder axis. Now we define the junction regions to be those connected regions of phase 1 that are bounded by the cylindrical subsections.

To obtain the trial stress field, we first solve for the tensions in the rods of Figure 9(a) when the rods are completely rigid and supporting a stress. These are found just by balance of forces at the junctions. If the rods parallel to a_i have a tension T_i (which could be negative) then we take in the cylindrical subsection of the corresponding strut of the final pentamode a trial stress field $T_i a_i a_i^T / (|a_i|^2 \pi r^2)$ giving rise to a net force T_i pulling (pushing if T_i is negative) on the adjacent junction regions. Inside the junction region we take a stress field that satisfies the



Figure 10. A detailed view of the square beam array microstructure which is used as the easily shearable section in the pentamode cylindrical struts. The vector *a* is chosen to be one of the four vectors a_k for k = 0, 1, 2, 3, as appropriate to each pentamode strut orientated parallel to a_k . The square beams can support tension (or compression) in the direction of the beam, and in particular can support a constant macroscopic stress $\sigma_k^B = \alpha_k a_k a_k^T$. As we are working in the framework of linear elasticity, we ignore the very real possibility that the beams will buckle.

elasticity equations appropriate to phase 1 filling the junction region when constant tractions $T_i/(\pi r^2)$ act on the four disks that border the cylindrical subsections, and there are no forces on the remaining surface of the junction regions.

Obtaining appropriate trial strain fields is also not too difficult. We first consider an infinitesimal motion that the rod model with Figure 9(a) as the unit cell can undergo when the rods are rigid but the pin junctions are flexible. Then in the final pentamode the junction regions are taken to undergo a rigid body translation which is the same as that of the corresponding pin junction in the rod model. The cylindrical subsections undergo appropriate shears to ensure continuity of the displacement. The trial displacement in the remaining multiconnected region of phase 2 bordered by the junction regions and the cylindrical subsection can be somewhat arbitrary, and is not really important. One could take it as the solution for the displacement field when phase 2 has some nonzero elastic moduli, and the displacement at the boundary of the junctions and cylindrical subsections matches that of the trial field just specified. The trial strain field is bounded and nonzero only in phase 2, and therefore the associated upper bound on the elastic energy scales in proportion to δ .

It is clear from the choice of these trial stress and strain fields that the macroscopic stress the material supports and the easy motions it permits are exactly the same as those for the ideal model with rods and pin junctions that has the unit cell pictured in Figure 9(a), and which provided the basis for our construction. That this structure can support any desired average stress, and only that average stress, is then a direct consequence of the analysis in Section 5.2 of [Milton and Cherkaev 1995].



Figure 11. Some of the replacements that are needed to obtain desired unimode, bimode, trimode, or quadramode materials.

To obtain any desired unimode, bimode, trimode, or quadramode material, having respectively p = 1, 2, 3, 4 independent easy modes of deformation, and supporting respectively 6 – p applied stresses σ_i^0 for j = 1, ..., 6 - p, we follow the prescription given by Milton and Cherkaev [1995]. That is, we superimpose, one at a time, 6-p pentamode structures, each supporting one of the stresses σ_i^0 , with struts which are sufficiently thin to ensure that one can (with appropriate modification specified below) superimpose the structures without collision. When doing this superimposition we first remove phase 2 and shift the lattice structures to try to avoid unwanted intersections of phase 1. This may not always be possible, so in the event two vertices clash we make the replacement in Figure 11 (left) in one of the structures (which may of course then cause additional unwanted intersections of the struts). Then if two (or more) struts intersect we make the replacement in Figure 11 (right) in all but one of the struts (which then passes through each hole). The remaining possibility we want to avoid is that two pentamode struts are parallel and intersect when we superimpose the structures. Due to the freedom in the choice of the a_k that give a desired σ_i^0 , we can always choose our 6 - ppentamode structures to avoid such clashes. Finally, the shearable section in each pentamode strut should be placed in a section that has not been modified, so it still is parallel to one of the a_k . At the very end any remaining space that is not filled by phase 1 should be filled by the extremely compliant phase 2.

9. Continuity of the energy functions

It follows from the preceding analysis that we can determine the three energy functions $2^{-1} + 2$

$$W_{f}^{5}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}), W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}), W_{f}^{5}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0})$$

in the limit $\delta \to 0$ for almost all combinations of applied fields. Here we establish that these energy functions are continuous functions of the applied fields in the limit $\delta \to 0$, and therefore we obtain expressions for the energy functions for all combinations of applied fields in this limit.

Recall that the set GU_f is characterized by its *W*-transform. For example, part of it is described by the function

$$W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}) = \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\sum_{i=1}^{4} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{2} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0} \right].$$
(9-1)

Here we want to show that such energy functions are continuous in their arguments. Let the tensor $C_*(\sigma_1^0, \sigma_2^0, \epsilon_1^0, \epsilon_2^0, \epsilon_3^0, \epsilon_4^0)$ be a minimizer of (9-1), and suppose we perturb the applied stress fields σ_j^0 by $\delta\sigma_j^0$ and the applied strain fields ϵ_i^0 by $\delta\epsilon_i^0$. Now consider the walled material with a geometry described by the characteristic function

$$\chi_w(\mathbf{x}) = \prod_{k=1}^3 (1 - H_{\epsilon'}(\mathbf{x} \cdot \mathbf{n}_k)), \qquad (9-2)$$

where n_1 , n_2 , and n_3 are the three orthogonal unit vectors

$$\boldsymbol{n}_1 = \begin{pmatrix} 1\\0\\0 \end{pmatrix}, \quad \boldsymbol{n}_2 = \begin{pmatrix} 0\\1\\0 \end{pmatrix}, \quad \boldsymbol{n}_3 = \begin{pmatrix} 0\\0\\1 \end{pmatrix}, \quad (9-3)$$

and ϵ' is a small parameter that gives the thickness of the walls. Inside the walls, where $\chi_w(\mathbf{x}) = 0$, we put an isotropic composite of phase 1 and phase 2, mixed in the proportions f and 1 - f with isotropic effective elasticity tensor $C(\kappa_0, \mu_0)$, where κ_0 is the effective bulk modulus and μ_0 is the effective shear modulus, which are assumed to have nonzero limits as $\delta \to 0$. (The isotropic composite could consist of islands of void surrounded by phase 1.) Outside the walls, where $\chi_w(\mathbf{x}) = 1$, we put the material that has an effective tensor

$$\boldsymbol{C}_*^1 = \boldsymbol{C}_*(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0, \boldsymbol{\epsilon}_4^0).$$

Let C'_* be the effective tensor of the composite. We have the variational principle

$$\sum_{i=1}^{4} (\boldsymbol{\epsilon}_{i}^{0} + \delta \boldsymbol{\epsilon}_{i}^{0}) : \boldsymbol{C}_{*}^{\prime} (\boldsymbol{\epsilon}_{i}^{0} + \delta \boldsymbol{\epsilon}_{i}^{0}) + \sum_{j=1}^{2} (\boldsymbol{\sigma}_{j}^{0} + \delta \boldsymbol{\sigma}_{j}^{0}) : (\boldsymbol{C}_{*}^{\prime})^{-1} (\boldsymbol{\sigma}_{j}^{0} + \delta \boldsymbol{\sigma}_{j}^{0})$$

$$= \min_{\underline{\boldsymbol{\epsilon}}_{1}, \underline{\boldsymbol{\epsilon}}_{2}, \underline{\boldsymbol{\epsilon}}_{3}, \underline{\boldsymbol{\epsilon}}_{4}, \underline{\boldsymbol{\sigma}}_{1}, \underline{\boldsymbol{\sigma}}_{2}} \left\langle \sum_{i=1}^{4} \underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) : [\boldsymbol{\chi}_{w}(\boldsymbol{x}) \boldsymbol{C}_{*}^{1} + (1 - \boldsymbol{\chi}_{w}(\boldsymbol{x})) \boldsymbol{C}(\boldsymbol{\kappa}_{0}, \boldsymbol{\mu}_{0})] \underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x})$$

$$+ \sum_{j=1}^{2} \underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x}) : [\boldsymbol{\chi}_{w}(\boldsymbol{x}) \boldsymbol{C}_{*}^{1} + (1 - \boldsymbol{\chi}_{w}(\boldsymbol{x})) \boldsymbol{C}(\boldsymbol{\kappa}_{0}, \boldsymbol{\mu}_{0})]^{-1} \underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x}) \right\rangle, \quad (9-4)$$

where the minimum is over fields subject to the appropriate average values and differential constraints. We choose constant trial strain fields

$$\underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) = \boldsymbol{\epsilon}_{i}^{0} + \delta \boldsymbol{\epsilon}_{i}^{0}, \quad i = 1, 2, 3, 4,$$
(9-5)

and trial stress fields

$$\underline{\sigma}_{j}(\mathbf{x}) = \sigma_{j}^{0} + \delta \underline{\sigma}_{j}(\mathbf{x}), \quad j = 1, 2,$$
(9-6)

where $\delta \underline{\sigma}_j(\mathbf{x})$ has average value $\delta \sigma_j^0$ and is concentrated in the walls. Specifically, if $\{\delta \sigma_j^0\}_{k\ell}$ denote the matrix elements of $\delta \sigma_j^0$, and letting

$$\delta \boldsymbol{\sigma}_{j}^{1} = \begin{pmatrix} 0 & 0 & 0 \\ 0 & 0 & \{\delta \boldsymbol{\sigma}_{j}^{0}\}_{23} \\ 0 & \{\delta \boldsymbol{\sigma}_{j}^{0}\}_{32} & \{\delta \boldsymbol{\sigma}_{j}^{0}\}_{33} \end{pmatrix},$$

$$\delta \boldsymbol{\sigma}_{j}^{2} = \begin{pmatrix} \{\delta \boldsymbol{\sigma}_{j}^{0}\}_{11} & 0 & \{\delta \boldsymbol{\sigma}_{j}^{0}\}_{13} \\ 0 & 0 & 0 \\ \{\delta \boldsymbol{\sigma}_{j}^{0}\}_{31} & 0 & 0 \end{pmatrix},$$

$$\delta \boldsymbol{\sigma}_{j}^{3} = \begin{pmatrix} 0 & \{\delta \boldsymbol{\sigma}_{j}^{0}\}_{12} & 0 \\ \{\delta \boldsymbol{\sigma}_{j}^{0}\}_{21} & \{\delta \boldsymbol{\sigma}_{j}^{0}\}_{22} & 0 \\ 0 & 0 & 0 \end{pmatrix},$$
(9-7)

then we choose

$$\delta \underline{\sigma}_{j}(\boldsymbol{x}) = \sum_{k=1}^{3} \delta \boldsymbol{\sigma}_{j}^{k} H_{\epsilon'}(\boldsymbol{x} \cdot \boldsymbol{n}_{k}) / \epsilon', \qquad (9-8)$$

which has the required average value $\delta \sigma_j^0$ and satisfies the differential constraints appropriate to a stress field because $\delta \sigma_j^k n_k = 0$.

Hence, there exist positive constants α and β such that for sufficiently small ϵ' and for sufficiently small variations $\delta \sigma_i^0$ and $\delta \epsilon_i^0$ in the applied fields, we have

$$\left\langle \sum_{i=1}^{4} \underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) : [\boldsymbol{\chi}_{w}(\boldsymbol{x})\boldsymbol{C}_{*}^{1} + (1 - \boldsymbol{\chi}_{w}(\boldsymbol{x}))\boldsymbol{C}(\boldsymbol{\kappa}_{0}, \boldsymbol{\mu}_{0})]\underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) + \sum_{j=1}^{2} \underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x}) : [\boldsymbol{\chi}_{w}(\boldsymbol{x})\boldsymbol{C}_{*}^{1} + (1 - \boldsymbol{\chi}_{w}(\boldsymbol{x}))\boldsymbol{C}(\boldsymbol{\kappa}_{0}, \boldsymbol{\mu}_{0})]^{-1}\underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x}) \right\rangle \\ \leq W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}) + \alpha\boldsymbol{\epsilon}' + \beta K/\boldsymbol{\epsilon}', \quad (9-9)$$

where K represents the norm

$$K = \sqrt{\sum_{i=1}^{4} \delta \boldsymbol{\epsilon}_{i}^{0} : \delta \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{2} \delta \boldsymbol{\sigma}_{j}^{0} : \delta \boldsymbol{\sigma}_{j}^{0}}, \qquad (9-10)$$

of the field variations. Choosing $\epsilon' = \sqrt{\beta K/\alpha}$ to minimize the right-hand side of (9-9), we obtain

$$W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0}+\delta\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0}+\delta\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\epsilon}_{1}^{0}+\delta\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0}+\delta\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0}+\delta\boldsymbol{\epsilon}_{3}^{0},\boldsymbol{\epsilon}_{4}^{0}+\delta\boldsymbol{\epsilon}_{4}^{0})$$

$$\leq W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0},\boldsymbol{\epsilon}_{4}^{0})+2\sqrt{\alpha\beta K}.$$
 (9-11)

In obtaining the bound (9-9) we have used the fact that K^2 is less than K for sufficiently small K, specifically K < 1. Clearly the right-hand side of (9-11) approaches $W_f^4(\sigma_1^0, \sigma_2^0, \epsilon_1^0, \epsilon_2^0, \epsilon_3^0, \epsilon_4^0)$ as $K \to 0$. On the other hand, by repeating the same argument with the roles of

$$W_f^4(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0, \boldsymbol{\epsilon}_4^0)$$

and

$$W_f^4(\boldsymbol{\sigma}_1^0 + \delta\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0 + \delta\boldsymbol{\sigma}_2^0, \boldsymbol{\epsilon}_1^0 + \delta\boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0 + \delta\boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0 + \delta\boldsymbol{\epsilon}_3^0, \boldsymbol{\epsilon}_4^0 + \delta\boldsymbol{\epsilon}_4^0)$$

reversed, and with

$$\boldsymbol{C}_{*}(\boldsymbol{\sigma}_{1}^{0}+\delta\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0}+\delta\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\epsilon}_{1}^{0}+\delta\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0}+\delta\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0}+\delta\boldsymbol{\epsilon}_{3}^{0},\boldsymbol{\epsilon}_{4}^{0}+\delta\boldsymbol{\epsilon}_{4}^{0})$$

replacing

$$\boldsymbol{C}_*(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0, \boldsymbol{\epsilon}_4^0),$$

we deduce that

$$W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0},\boldsymbol{\epsilon}_{4}^{0}) \leq W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0}+\delta\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0}+\delta\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\epsilon}_{1}^{0}+\delta\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0}+\delta\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0}+\delta\boldsymbol{\epsilon}_{3}^{0},\boldsymbol{\epsilon}_{4}^{0}+\delta\boldsymbol{\epsilon}_{4}^{0}) + 2\sqrt{\alpha\beta K}.$$
 (9-12)

This, together with (9-11), establishes the continuity of $W_f^4(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0, \boldsymbol{\epsilon}_4^0)$. The continuity of the other energy functions follows by the same argument.

10. Conclusion

We have established the following two theorems.

Theorem 10.1. Consider composites in three dimensions of two materials with positive definite elasticity tensors C_1 and $C_2 = \delta C_0$ mixed in proportions f and 1 - f. Let the seven energy functions W_f^k , for k = 0, 1, ..., 6, that characterize the set GU_f (with $U = (C_1, \delta C_0)$) of possible elastic tensors be defined by (3-9). These energy functions involve a set of applied strains ϵ_i^0 and applied stresses σ_j^0 meeting the orthogonality condition (3-10). The energy function W_f^0 is given by

$$W_{f}^{0}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\sigma}_{5}^{0}, \boldsymbol{\sigma}_{6}^{0}) = \sum_{j=1}^{6} \boldsymbol{\sigma}_{j}^{0} : \widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\sigma}_{5}^{0}, \boldsymbol{\sigma}_{6}^{0}) \boldsymbol{\sigma}_{j}^{0} \quad (10\text{-}1)$$

(as proved by Avellaneda [1987b]). Here $\tilde{C}_{f}^{A}(\sigma_{1}^{0}, \sigma_{2}^{0}, \sigma_{3}^{0}, \sigma_{4}^{0}, \sigma_{5}^{0}, \sigma_{6}^{0})$ is the effective elasticity tensor of a complementary Avellaneda material that is a sequentially layered laminate with the minimum value of the sum of complementary energies

$$\sum_{j=1}^{6} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0}.$$
(10-2)

Additionally, we now have

$$\lim_{\delta \to 0} W_{f}^{3}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}) = \sum_{j=1}^{3} \boldsymbol{\sigma}_{j}^{0} : [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, 0, 0, 0)]^{-1} \boldsymbol{\sigma}_{j}^{0},$$

$$\lim_{\delta \to 0} W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}) = \sum_{j=1}^{2} \boldsymbol{\sigma}_{j}^{0} : [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, 0, 0, 0, 0)]^{-1} \boldsymbol{\sigma}_{j}^{0}, \quad (10\text{-}3)$$

$$\lim_{\delta \to 0} W_{f}^{5}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}) = \boldsymbol{\sigma}_{1}^{0} : [\widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, 0, 0, 0, 0, 0)]^{-1} \boldsymbol{\sigma}_{1}^{0}, \quad (10\text{-}3)$$

$$\lim_{\delta \to 0} W_{f}^{6}(\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}) = 0.$$

for all combinations of applied stresses $\boldsymbol{\sigma}_{j}^{0}$ and applied strains $\boldsymbol{\epsilon}_{i}^{0}$. When det $\boldsymbol{\epsilon}_{1}^{0} = 0$ but $\boldsymbol{\epsilon}_{1}^{0}$ is not positive semidefinite or negative semidefinite, we have

$$\lim_{\delta \to 0} W_f^1(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0, \boldsymbol{\sigma}_4^0, \boldsymbol{\sigma}_5^0, \boldsymbol{\epsilon}_1^0) = \sum_{j=1}^5 \boldsymbol{\sigma}_j^0 : [\widetilde{\boldsymbol{C}}_f^A(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0, \boldsymbol{\sigma}_4^0, \boldsymbol{\sigma}_5^0, 0)]^{-1} \boldsymbol{\sigma}_j^0, \quad (10\text{-}4)$$

while when the equation $\det(\epsilon_1^0 + t\epsilon_2^0)$ has at least two distinct roots for t (the condition for which is given by (7-5)), and additionally, the matrix pencil $\epsilon(t) = \epsilon_1^0 + t\epsilon_2^0$ does not contain any positive definite or negative definite matrices as t is varied (which requires that the quantities in (7-4) are never all positive, or all negative), we have

$$\lim_{\delta \to 0} W_f^2(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0, \boldsymbol{\sigma}_4^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0) = \sum_{j=1}^4 \boldsymbol{\sigma}_j^0 : [\widetilde{\boldsymbol{C}}_f^A(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0, \boldsymbol{\sigma}_4^0, 0, 0)]^{-1} \boldsymbol{\sigma}_j^0.$$
(10-5)

Theorem 10.2. For 2-dimensional composites, the four energy functions W_f^k , for k = 0, 1, 2, 3, are defined by (6-1), and these characterize the set GU_f , with $U = (C_1, \delta C_0)$, of possible elastic tensors C_* of composites of two phases with positive definite elasticity tensors C_1 and $C_2 = \delta C_0$. These energy functions involve a set of applied strains ϵ_i^0 and applied stresses σ_j^0 meeting the orthogonality condition (3-10). The energy function W_f^0 is given by

$$W_{f}^{0}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}) = \sum_{j=1}^{3} \boldsymbol{\sigma}_{j}^{0} : \widetilde{\boldsymbol{C}}_{f}^{A}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}) \boldsymbol{\sigma}_{j}^{0}$$
(10-6)

(as proved by Avellaneda [1987b]), where $\widetilde{C}_{f}^{A}(\sigma_{1}^{0}, \sigma_{2}^{0}, \sigma_{3}^{0})$ is the effective elasticity tensor of a complementary Avellaneda material that is a sequentially layered laminate with the minimum value of the sum of complementary energies

$$\sum_{j=1}^{5} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0}.$$
(10-7)

We also have the trivial result that

$$\lim_{\delta \to 0} W_f^3(\boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0) = 0.$$
(10-8)

When det $\epsilon_1^0 \leq 0$ we have

$$\lim_{\delta \to 0} W_f^1(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\epsilon}_1^0) = \sum_{j=1}^2 \boldsymbol{\sigma}_j^0 : [\widetilde{\boldsymbol{C}}_f^A(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, 0)]^{-1} \boldsymbol{\sigma}_j^0,$$
(10-9)

while when det $\epsilon_1^0 < 0$ or when $f(t) = det(\epsilon_1^0 + t\epsilon_2^0)$ is quadratic in t with two distinct roots, or when f(t) is linear in t with a nonzero t coefficient, we have

$$\lim_{\delta \to 0} W_f^2(\boldsymbol{\sigma}_1^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0) = \boldsymbol{\sigma}_1^0 : [\widetilde{\boldsymbol{C}}_f^A(\boldsymbol{\sigma}_1^0, 0, 0)]^{-1} \boldsymbol{\sigma}_1^0.$$
(10-10)

These theorems, and the accompanying microstructures, help define what sort of elastic behaviors are theoretically possible in 2- and 3-dimensional printed materials. They should serve as benchmarks for the construction of more realistic microstructures that can be manufactured. We have found the minimum over all microstructures of various sums of energies and complementary energies. More realistic designs can be obtained by adding to this sum a term that penalizes the surface area as done for a single energy minimization by Kohn and Wirth [2014; 2016].

It remains an open problem to find expressions for the energy functions in the cases not covered by these theorems. Even for an isotropic composite with a bulk modulus κ_* and a shear modulus μ_* , the set of all possible pairs (κ_* , μ_*) is still not completely characterized either in the limit $\delta \rightarrow 0$ or in the limit $\delta \rightarrow \infty$. In these limits the bounds of Berryman and Milton [1988] and Cherkaev and Gibiansky [1993] decouple and provide no extra information beyond that provided by the Hashin–Shtrikman–Hill bounds [Hashin and Shtrikman 1963; Hashin 1965; Hill 1963; 1964]. While the results of this paper show that in the limit $\delta \rightarrow 0$ one can obtain 2- or 3-dimensional structures attaining the Hashin–Shtrikman–Hill upper bound on κ_* , while having $\mu_* = 0$, it is not clear what the maximum value for μ_* is, given that $\kappa_* = 0$.

One important corollary of this work is that it gives a complete characterization of the possible triplets (ϵ^0 , σ^0 , f) of average strain ϵ^0 , average stress σ^0 , and volume fraction f that can occur in 2-dimensional and 3-dimensional printed materials in

the limit $\delta \rightarrow 0$. This will be discussed in a separate paper [Milton and Camar-Eddine 2016].

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TOWARDS A COMPLETE CHARACTERIZATION OF THE EFFECTIVE ELASTICITY TENSORS OF MIXTURES OF AN ELASTIC PHASE AND AN ALMOST RIGID PHASE

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The set GU_f of possible effective elastic tensors of composites built from two materials with positive definite elasticity tensors C_1 and $C_2 = \delta C_0$ comprising the set $U = \{C_1, \delta C_0\}$ and mixed in proportions f and 1 - f is partly characterized in the limit $\delta \to \infty$. The material with tensor C_2 corresponds to a material which (for technical reasons) is almost rigid in the limit $\delta \to \infty$. This paper, and the underlying microgeometries, has many aspects in common with the companion paper "On the possible effective elasticity tensors of 2-dimensional and 3-dimensional printed materials". The chief difference is that one has a different algebraic problem to solve: determining the subspaces of stress fields for which the thin walled structures can be rigid, rather than determining, as in the companion paper, the subspaces of strain fields for which the thin walled structure is compliant. Recalling that GU_f is completely characterized through minimums of sums of energies, involving a set of applied strains, and complementary energies, involving a set of applied stresses, we provide descriptions of microgeometries that in appropriate limits achieve the minimums in many cases. In these cases the calculation of the minimum is reduced to a finite-dimensional minimization problem that can be done numerically. Each microgeometry consists of a union of walls in appropriate directions, where the material in the wall is an appropriate p-mode material that is almost rigid to $6 - p \le 5$ independent applied stresses, yet is compliant to any strain in the orthogonal space. Thus the walls, by themselves, can support stress with almost no deformation. The region outside the walls contains "Avellaneda material", which is a hierarchical laminate that minimizes an appropriate sum of elastic energies.

1. Introduction

This paper is a companion to "On the possible effective elasticity tensors of 2dimensional and 3-dimensional printed materials" [Milton et al. 2017], which gives a partial characterization of the set GU_f of effective elasticity tensors that can be

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produced in the limit $\delta \to 0$ if we mix in prescribed proportions f and 1-f two materials with positive definite and bounded elasticity tensors C_1 and $C_2 = \delta C_0$. Here we consider the opposite limit $\delta \to \infty$, which corresponds to mixing in prescribed proportions an elastic phase and an almost rigid phase. Our results are summarized in the theorem in the conclusion section. For a complete introduction and summary of previous results the reader is urged to read at least the first three sections of the companion paper. The essential ideas presented here are much the same as those contained in the companion paper. However, the algebraic problem relevant to this paper, of determining when the set of walls can support a set of stress fields, is quite different from the algebraic problem encountered in the companion paper of determining when the set of walls is compliant to a set of strain fields.

The microstructures we consider involve taking three limits. First, as they have structure on multiple length scales, the homogenization limit where the ratio between length scales goes to infinity needs to be taken. Second, the limit $\delta \rightarrow \infty$ needs to be taken. Third, as the structure involves walls of width ϵ , which are very stiff to certain applied stresses, the limit $\epsilon \rightarrow 0$ needs to be taken so the contribution to the elastic energy of these walls goes to zero, when the structure is compliant to an applied strain. The limits should be taken in this order, as, for example, standard homogenization theory is justified only if δ is positive and finite, so we need to take the homogenization limit before taking the limit $\delta \rightarrow \infty$.

As in the companion paper we emphasize that our analysis is valid only for linear elasticity, and ignores nonlinear effects such as buckling, which may be important even for small deformations. It is also important to emphasize that to apply our results when phase 2 is perfectly rigid (rather than almost rigid) requires special care. Indeed, if phase 2 is perfectly rigid, then many of the microgeometries considered here do not permit the kind of motions that are permitted for any finite value of δ , no matter how large. In particular, the structures considered in Figures 6, 8, and 9(d) of the companion paper would be completely rigid if phase 2 was perfectly rigid. To permit the required motions, one has to first replace the rigid phase 2 with a composite with a small amount of phase 1 as the matrix phase, so that its effective elasticity tensor is finite but approaches infinity as the proportion of phase 1 in it tends to zero. The microgeometry in this composite needs to be much smaller than the scales in the geometries discussed here, which would involve mixtures of it and phase 1.

2. Characterizing *G* closures through sums of energies and complementary energies

Cherkaev and Gibiansky [1992; 1993] found that bounding sums of energies and complementary energies could lead to very useful bounds on *G*-closures. It was

subsequently proved in [Francfort and Milton 1994; Milton 1994] that minimums over $C_* \in GU_f$ of such sums of energies and complementary energies completely characterize GU_f in much the same way that Legendre transforms characterize convex sets: the stability under lamination of GU_f is what allows one to recover GU_f from the values of these minimums (see also Chapter 30 in [Milton 2002]). Specifically, in the case of 3-dimensional elasticity, the set GU_f is completely characterized if we know the seven "energy functions",

$$\begin{split} W_{f}^{0}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\sigma}_{3}^{0},\boldsymbol{\sigma}_{4}^{0},\boldsymbol{\sigma}_{5}^{0},\boldsymbol{\sigma}_{6}^{0}) &= \min_{\boldsymbol{C}_{*}\in GU_{f}} \sum_{j=1}^{6} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}_{j}^{0}, \\ W_{f}^{1}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\sigma}_{3}^{0},\boldsymbol{\sigma}_{4}^{0},\boldsymbol{\sigma}_{5}^{0},\boldsymbol{\epsilon}_{1}^{0}) &= \min_{\boldsymbol{C}_{*}\in GU_{f}} \left[\boldsymbol{\epsilon}_{1}^{0} : \boldsymbol{C}_{*}\boldsymbol{\epsilon}_{1}^{0} + \sum_{j=1}^{5} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}_{j}^{0} \right], \\ W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\sigma}_{3}^{0},\boldsymbol{\sigma}_{4}^{0},\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0}) &= \min_{\boldsymbol{C}_{*}\in GU_{f}} \left[\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*}\boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{4} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}_{j}^{0} \right], \\ W_{f}^{3}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\sigma}_{3}^{0},\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0}) &= \min_{\boldsymbol{C}_{*}\in GU_{f}} \left[\sum_{i=1}^{3} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*}\boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{3} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}_{j}^{0} \right], \\ W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0},\boldsymbol{\epsilon}_{4}^{0}) &= \min_{\boldsymbol{C}_{*}\in GU_{f}} \left[\sum_{i=1}^{4} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*}\boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{2} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}_{j}^{0} \right], \\ W_{f}^{5}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0},\boldsymbol{\epsilon}_{4}^{0},\boldsymbol{\epsilon}_{5}^{0}) &= \min_{\boldsymbol{C}_{*}\in GU_{f}} \left[\sum_{i=1}^{5} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*}\boldsymbol{\epsilon}_{i}^{0} \right] + \boldsymbol{\sigma}_{1}^{0} : \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}_{1}^{0} \right], \\ W_{f}^{6}(\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0},\boldsymbol{\epsilon}_{4}^{0},\boldsymbol{\epsilon}_{5}^{0}) &= \min_{\boldsymbol{C}_{*}\in GU_{f}} \left[\sum_{i=1}^{5} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*}\boldsymbol{\epsilon}_{i}^{0} \right] + \boldsymbol{\sigma}_{1}^{0} : \boldsymbol{C}_{*}^{-1}\boldsymbol{\sigma}_{1}^{0} \right], \end{aligned}$$

In fact, Milton and Cherkaev [1995] showed it suffices to know these functions for sets of applied strains ϵ_i^0 and applied stresses σ_j^0 that are mutually orthogonal:

$$(\boldsymbol{\epsilon}_i^0, \boldsymbol{\sigma}_j^0) = 0, \quad (\boldsymbol{\epsilon}_i^0, \boldsymbol{\epsilon}_k^0) = 0, \quad (\boldsymbol{\sigma}_j^0, \boldsymbol{\sigma}_\ell^0) = 0,$$

for all i, j, k, ℓ with $i \neq j, i \neq k, j \neq \ell$. (2-2)

The terms appearing in the minimums have a physical significance. For example, in the expression for W_f^2 ,

$$\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{4} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0}$$
(2-3)

has the physical interpretation of being the sum of energies per unit volume stored in the composite with effective elasticity tensor C_* when successively subjected to the two applied strains ϵ_1^0 and ϵ_2^0 and then to the four applied stresses $\sigma_1^0, \sigma_2^0, \sigma_3^0$, and σ_4^0 . To distinguish the terms $\epsilon_i^0 : C_* \epsilon_i^0$ and $\sigma_j^0 : C_*^{-1} \sigma_j^0$, the first is called an energy (it is really an energy per unit volume associated with the applied strain ϵ_i^0) and the second is called a complementary energy, although it too physically represents an energy per unit volume associated with the applied stress σ_i^0 .

For well-ordered materials with $C_2 \ge C_1$ (or the reverse), Avellaneda [1987] showed that there exist sequentially layered laminates of finite rank having an effective elasticity tensor $C_* = C_f^A(\epsilon_1^0, \epsilon_2^0, \epsilon_3^0, \epsilon_4^0, \epsilon_5^0, \epsilon_6^0)$ (not to be confused with the elasticity tensor $C_* = \widetilde{C}_f^A(\sigma_1^0, \sigma_2^0, \sigma_3^0, \sigma_4^0, \sigma_5^0, \sigma_6^0)$ used in the companion paper) that attains the minimum in the above expression for $W_f^6(\epsilon_1^0, \epsilon_2^0, \epsilon_3^0, \epsilon_4^0, \epsilon_5^0)$:

$$W_{f}^{6}(\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}, \boldsymbol{\epsilon}_{6}^{0}) = \sum_{i=1}^{6} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{f}^{A}(\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}, \boldsymbol{\epsilon}_{6}^{0}) \boldsymbol{\epsilon}_{i}^{0}.$$
(2-4)

The effective tensor $C_* = C_f^A(\epsilon_1^0, \epsilon_2^0, \epsilon_3^0, \epsilon_4^0, \epsilon_5^0, \epsilon_6^0)$ of the Avellaneda material is found by finding a combination of the parameters entering the formula for the effective tensor of sequentially layered laminates that minimizes the sum of six elastic energies. In general this has to be done numerically, but it suffices to consider laminates of rank at most 6 if C_1 is isotropic [Francfort et al. 1995], or, using an argument of Avellaneda [1987], to consider laminates of rank at most 21 if C_1 is anisotropic (see Section 2 in the companion paper).

In the case of 2-dimensional elasticity, the set GU_f is similarly completely characterized if we know the 4 "energy functions",

$$W_{f}^{0}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}) = \min_{\boldsymbol{C}_{*} \in GU_{f}} \sum_{j=1}^{3} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0},$$

$$W_{f}^{1}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\epsilon}_{1}^{0}) = \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\boldsymbol{\epsilon}_{1}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{1}^{0} + \sum_{j=1}^{2} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0} \right],$$

$$W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}) = \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\left(\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} \right) + \boldsymbol{\sigma}_{1}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{1}^{0} \right],$$

$$W_{f}^{3}(\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}) = \min_{\boldsymbol{C}_{*} \in GU_{f}} \sum_{i=1}^{3} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0}.$$

$$(2-5)$$

Again $W_f^3(\boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0)$ is attained for an "Avellaneda material" consisting of a sequentially layered laminate geometry having an effective tensor $\boldsymbol{C}_* = \boldsymbol{C}_f^A(\boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0)$, i.e.,

$$W_f^3(\boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0) = \sum_{i=1}^3 \boldsymbol{\epsilon}_i^0 : \boldsymbol{C}_f^A(\boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0) \boldsymbol{\epsilon}_i^0.$$
(2-6)

The effective tensor $C_* = C_f^A(\epsilon_1^0, \epsilon_2^0, \epsilon_3^0)$ of the Avellaneda material is found by finding a combination of the parameters entering the formula for the effective tensor of sequentially layered laminates that minimizes the sum of three elastic energies. In general this has to be done numerically, but it suffices to consider laminates of rank at most three if C_1 is isotropic [Avellaneda and Milton 1989], or, using an argument of Avellaneda [1987], to consider laminates of rank at most 6 if C_1 is anisotropic (see Section 2 in the companion paper).

3. Microgeometries which are associated with sharp bounds on many sums of energies and complementary energies

The analysis here of mixtures of an almost rigid phase mixed with an elastic phase is very similar to the analysis in the companion paper for mixtures of an extremely compliant phase and an elastic phase. The roles of stresses and strains are interchanged and now the challenge is to identify matrix pencils that are spanned by matrices with zero determinant, rather than symmetrized rank 1 matrices. We now have the inequalities

$$0 \leq W_{f}^{0}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\sigma}_{5}^{0}, \boldsymbol{\sigma}_{6}^{0}),$$

$$\boldsymbol{\epsilon}_{1}^{0} : [\boldsymbol{C}_{f}^{A}(0, 0, 0, 0, 0, \boldsymbol{\epsilon}_{1}^{0})]\boldsymbol{\epsilon}_{1}^{0} \leq W_{f}^{1}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\sigma}_{5}^{0}, \boldsymbol{\epsilon}_{1}^{0}),$$

$$\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : [\boldsymbol{C}_{f}^{A}(0, 0, 0, 0, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0})]\boldsymbol{\epsilon}_{i}^{0} \leq W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}),$$

$$\sum_{i=1}^{3} \boldsymbol{\epsilon}_{i}^{0} : [\boldsymbol{C}_{f}^{A}(0, 0, 0, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0})]\boldsymbol{\epsilon}_{i}^{0} \leq W_{f}^{3}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}), \qquad (3-1)$$

$$\sum_{i=1}^{4} \boldsymbol{\epsilon}_{i}^{0} : [\boldsymbol{C}_{f}^{A}(0, 0, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0})]\boldsymbol{\epsilon}_{i}^{0} \leq W_{f}^{4}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}),$$

$$\sum_{i=1}^{5} \boldsymbol{\epsilon}_{i}^{0} : [\boldsymbol{C}_{f}^{A}(0, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0})]\boldsymbol{\epsilon}_{i}^{0} \leq W_{f}^{5}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}),$$

The first inequality is clearly sharp, being attained when the composite consists of islands of phase 1 surrounded by a phase 2 (so that $(C_*)^{-1}$ approaches 0 as $\delta \to \infty$). Again the objective is to show that many of the other inequalities are also sharp in the limit $\delta \to \infty$, at least when the spaces spanned by the applied stresses σ_j^0 for $j = 1, 2, \ldots, 6 - p$ satisfy certain properties. This space of applied stresses associated with W_f^p has dimension 6 - p and its orthogonal complement defines the *p*-dimensional space V_p .

The recipe for doing this is to simply insert into a relevant Avellaneda material a microstructure occupying a thin walled region containing a p-mode material,

such that the walled structure, by itself, is very stiff when the applied stress lies in the (6-p)-dimensional subspace spanned by the σ_i^0 , yet allows strains in the orthogonal *p*-dimensional subspace \mathcal{V}_p spanned by the $\boldsymbol{\epsilon}_i^0$. We say a composite with effective tensor C_* built from the two materials C_1 and $C_2 = \delta C_0$ is very stiff to a stress σ_j^0 if the complementary energy $\sigma_j^0 : C_*^{-1} \sigma_j^0$ goes to zero as $\delta \to \infty$, and allows a strain ϵ_i^0 if the elastic energy $\epsilon_i^0 : C_* \epsilon_i^0$ has a finite limit as $\delta \to \infty$. These p-mode materials have exactly the same construction as that specified in Section 8 of the companion paper, only now the region that was occupied by the elastic phase is now occupied by the rigid phase, and the material that was occupied by the extremely compliant phase (which becomes void in the limit $\delta \rightarrow 0$) is occupied by the elastic phase. If we happened to choose $C_0 = C_1$, all the moduli (and effective moduli) are simply rescaled, i.e., for any δ , and in particular for large values of δ , if a mixture of two materials with effective tensors C_1 and C_1/δ has effective tensor C_* , then when rescaling the elasticity tensors of the two phases to δC_1 and C_1 , the resulting effective elasticity tensor will be δC_* . Thus, the analysis of the response of the *p*-mode materials is essentially the same as in the companion paper. Exactly the same trial fields can be chosen to bound the response of the *p*-mode material. Hence we do not repeat this analysis but instead the reader is referred to Section 8 of the companion paper.

The subspace orthogonal to \mathcal{V}_p is now required to be spanned by matrices $\boldsymbol{\sigma}^{(k)}$, for $k = 1, \dots, 6 - p$, such that

$$\boldsymbol{\sigma}^{(k)}\boldsymbol{n}_k = 0 \tag{3-2}$$

for some unit vector n_k . Thus the identifying feature of these matrices $\sigma^{(k)}$ is that they have zero determinant, and then n_k can be chosen as a null vector of $\sigma^{(k)}$. The existence of such matrices $\sigma^{(k)}$ is proved in Section 4. The proof uses small perturbations of the applied stresses and strains. But, due the continuity of the energy functions W_f^k established in Section 5, the small perturbations do not modify the generic result. The vectors n_k determine the orientation of the walls in the structure since a set of walls orthogonal to n can support any stress σ such that $\sigma n = 0$.

To define the thin walled structure, introduce the periodic function $H_c(x)$ with period 1 which takes the value 1 if $x - [x] \le c$, where [x] is the greatest integer less than x, and $c \in [0, 1]$ gives the relative thickness of each wall. Then for the unit vectors $\mathbf{n}_1, \mathbf{n}_2, \ldots, \mathbf{n}_{6-p}$ appearing in (3-2), and for a small relative thickness $c = \epsilon$, define the characteristic functions

$$\eta_k(\boldsymbol{x}) = H_\epsilon(\boldsymbol{x} \cdot \boldsymbol{n}_k + k/p). \tag{3-3}$$

This characteristic function defines a series of parallel walls, as shown on the left in Figure 1, each perpendicular to the vector \mathbf{n}_j , where $\eta_j(\mathbf{x}) = 1$ in the wall material. The additional shift term k/p in (3-3) ensures the walls associated with



Figure 1. Example of walled structures. On the left we have a "rank 1" walled structure and on the right a "rank 2" walled structure. The generalization to walled structures of any rank is obvious, and precisely defined by the characteristic function (3-4) that is 0 in the walls, and 1 in the remaining material.

 k_1 and k_2 do not intersect when it happens that $n_{k_1} = n_{k_2}$, at least when ϵ is small. We emphasize that ϵ is not a homogenization parameter, but rather represents a volume fraction of walls.

Now define the characteristic function

$$\chi_*(\mathbf{x}) = \prod_{k=1}^p (1 - \eta_k(\mathbf{x})).$$
(3-4)

If $p \le 3$, this is usually a periodic function of x, an exception being if p = 3 and there are no nonzero integers z_1 , z_2 , and z_3 such that $z_1n_1 + z_2n_2 + z_3n_3 = 0$. More generally, $\chi_*(x)$ is a quasiperiodic function of x. The walled structure is where $\chi_*(x)$ takes the value 0. In the case p = 2 the walled structure is illustrated on the right in Figure 1.

The walled structure is where $\chi_*(\mathbf{x})$ given by (3-4) takes the value 0. Inside it we put a *p*-mode material with effective tensor $C_*^2 = C_*(\mathcal{V}_p)$ that allows any applied strain ϵ^0 in the space \mathcal{V}_p but which is very stiff to any stress σ^0 orthogonal to the space \mathcal{V}_p . Using the six matrices

$$\boldsymbol{v}_1 = \boldsymbol{\sigma}_1^0 / |\boldsymbol{\sigma}_1^0|, \dots, \boldsymbol{v}_{6-p} = \boldsymbol{\sigma}_{6-p}^0 / |\boldsymbol{\sigma}_{6-p}^0|, \boldsymbol{v}_{7-p} = \boldsymbol{\epsilon}_1^0 / |\boldsymbol{\epsilon}_1^0|, \dots, \boldsymbol{v}_6 = \boldsymbol{\epsilon}_p^0 / |\boldsymbol{\epsilon}_p^0| \quad (3-5)$$

as our basis for the 6-dimensional space of 3×3 symmetric matrices, the compliance tensor $[C_*(\mathcal{V}_p)]^{-1}$ in this basis takes the limiting form

$$\lim_{\delta \to \infty} [\boldsymbol{C}_*(\boldsymbol{\mathcal{V}}_p)]^{-1} = \begin{pmatrix} 0 & 0\\ 0 & \boldsymbol{B} \end{pmatrix}, \qquad (3-6)$$

where **B** represents a (strictly) positive definite $p \times p$ matrix and the 0 on the diagonal represents the $(6-p) \times (6-p)$ zero matrix. Inside the walled structure, where $\chi_*(\mathbf{x}) = 1$, we put the Avellaneda material with effective elasticity tensor

$$\boldsymbol{C}_*^1 = \boldsymbol{C}_f^A(0,\ldots,0,\boldsymbol{\epsilon}_1^0,\ldots,\boldsymbol{\epsilon}_p^0).$$

In a variational principle similar to (4-4) in the companion paper (i.e., treating the Avellaneda material and the *p*-mode material both as homogeneous materials with effective tensors $C_*^1 = C_f^A$ and $C_*^2 = C_*(\mathcal{V}_p)$, respectively) we choose trial strain fields that are constant,

$$\underline{\boldsymbol{\epsilon}}_i(\boldsymbol{x}) = \boldsymbol{\epsilon}_i^0, \quad \text{for } i = 1, 2, \dots, p, \tag{3-7}$$

thus trivially fulfilling the differential constraints, and trial stress fields of the form

$$\underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x}) = \sum_{k=1}^{6-p} \boldsymbol{\sigma}_{j,k} \eta_{k}(\boldsymbol{x}) / \epsilon, \qquad (3-8)$$

which are required to have the average values

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$$\boldsymbol{\sigma}_{j}^{0} = \langle \underline{\boldsymbol{\sigma}}_{j} \rangle = \sum_{k=1}^{6-p} \boldsymbol{\sigma}_{j,k}, \qquad (3-9)$$

and the matrices $\sigma_{i,j}$ are additionally required to lie in the space orthogonal to \mathcal{V}_p (so they cost very little energy) and satisfy

$$\boldsymbol{\sigma}_{j,k} = c_{j,k} \boldsymbol{\sigma}^{(k)}, \qquad (3-10)$$

for some choice of parameters $c_{j,k}$ to ensure that $\sigma_{j,k}n_k = 0$ and hence that $\underline{\sigma}_j(x)$ satisfies the differential constraints of a stress field — this requires $\underline{\sigma}_j(x)n_k$ to be continuous across any interface with normal n_k . Additionally, the $c_{j,k}$ in (3-10) should be chosen so the σ_j^0 given by (3-9) are orthogonal.

To find upper bounds on the energy associated with this trial stress field, first consider those parts of the walled structure that are outside of any junction regions, i.e., where for some k we have $\eta_k(\mathbf{x}) = 1$, while $\eta_s(\mathbf{x}) = 0$ for all $s \neq k$. An upper bound for the volume fraction occupied by the region where $\eta_k(\mathbf{x}) = 1$ while $\eta_s(\mathbf{x}) = 0$ for all $s \neq k$ is of course ϵ , as this represents the volume of the region where $\eta_k(\mathbf{x}) = 1$. The associated energy per unit volume of the trial stress field in those parts of the walled structure that are outside of any junction regions is bounded above by

$$\sum_{k=1}^{6-p} \boldsymbol{\sigma}_{j,k} : [\boldsymbol{C}_*(\boldsymbol{\mathcal{V}}_p)]^{-1} \boldsymbol{\sigma}_{j,k} / \epsilon.$$
(3-11)

With an appropriate choice of multimode material, one can construct bounded trial stress fields that are essentially concentrated in phase 2, and consequently, $\sigma_{j,k} : [C_*(\mathcal{V}_p)]^{-1}\sigma_{j,k}$ is bounded above by a quantity proportional to $1/\delta$. Our assumption that we take the limit $\delta \to \infty$ before taking the limit $\epsilon \to 0$ ensures that $1/(\delta\epsilon) \to 0$, and thus ensures that the quantity (3-11) goes to zero in this limit.

Next, consider those junction regions where only two walls meet, i.e., where for some k_1 and $k_2 > k_1$, \mathbf{x} is such that $\eta_{k_1}(\mathbf{x}) = \eta_{k_2}(\mathbf{x}) = 1$ while $\eta_s(\mathbf{x}) = 0$ for all s not equal to k_1 or k_2 . Provided $\mathbf{n}_{k_1} \neq \mathbf{n}_{k_2}$, an upper bound for the volume fraction occupied by each such junction region is ϵ^2 . Then the associated energy per unit volume of the trial stress field in these junction regions where only two walls meet is bounded above by

$$\sum_{k_1=1}^{6-p} \sum_{k_2=k_1+1}^{6-p} (\boldsymbol{\sigma}_{i,k_1} + \boldsymbol{\sigma}_{i,k_2}) : [\boldsymbol{C}_*(\boldsymbol{\mathcal{V}}_p)]^{-1} (\boldsymbol{\sigma}_{j,k_1} + \boldsymbol{\sigma}_{j,k_2}).$$
(3-12)

Thus, the powers of ϵ cancel and this energy density goes to zero if the multimode material is easily compliant to the strains $\sigma_{j,k_1} + \sigma_{j,k_2}$ for all k_1 and k_2 with $k_2 > k_1$.

Finally, consider those junction regions where three or more walls meet, i.e., for some $k_1, k_2 > k_1$, and $k_3 > k_2$, \mathbf{x} is such that $\eta_{k_i}(\mathbf{x}) = 1$ for i = 1, 2, 3. For a given choice of $k_1, k_2 > k_1$, and $k_3 > k_2$ such that the three vectors $\mathbf{n}_{k_1}, \mathbf{n}_{k_2}$, and \mathbf{n}_{k_3} are not coplanar, an upper bound for the volume fraction occupied by this region is ϵ^3 . In the case that the three vectors $\mathbf{n}_{k_1}, \mathbf{n}_{k_2}$, and \mathbf{n}_{k_3} are coplanar, we can ensure that the volume fraction occupied by this region is ϵ^3 or less by appropriately translating one or two walled structures, i.e., by replacing $\eta_{k_m}(\mathbf{x})$ with $\eta_{k_m}(\mathbf{x} + \alpha_i \mathbf{n}_{k_m})$ for m = 2, 3, for an appropriate choice of α_2 and α_3 between 0 and 1. Since the energy density of the trial field in these regions scales as $\epsilon^3/\epsilon^2 = \epsilon$, we can ignore this contribution in the limit $\epsilon \rightarrow 0$ as it goes to zero too.

From this analysis of the energy densities associated with the trial fields it follows that one does not necessarily need the pentamode, quadramode, trimode, bimode, and unimode materials as appropriate for the material inside the walled structure. Instead, by modifying the construction, it suffices to use only pentamode and quadramode materials. In the walled structure we now put pentamode materials in those sections where for some k, we have $\eta_k(\mathbf{x}) = 1$ while $\eta_{k'}(\mathbf{x}) = 0$ for all $k' \neq k$. Each pentamode material is very stiff to the single stress $\sigma^{(k)}$ appropriate to the wall under consideration. In each junction region of the walled structure where $\eta_{k_1}(\mathbf{x}) = \eta_{k_2}(\mathbf{x}) = 1$ for some $k_1 \neq k_2$ while $\eta_k(\mathbf{x}) = 0$ for all k not equal to k_1 or k_2 , we put a quadramode material which is very stiff to any stress in the subspace spanned by $\sigma^{(k_1)}$ and $\sigma^{(k_2)}$ as appropriate to the junction region under consideration. In the remaining junction regions of the walled structure (where three or more walls intersect) we put phase 1. The contribution to the average energy of the fields in these regions vanishes as $\epsilon \rightarrow 0$ as discussed above.

By these constructions we effectively obtain materials with elasticity tensors C_* such that

$$\lim_{\delta \to \infty} (\boldsymbol{C}_*)^{-1} = \Pi_p (\boldsymbol{C}_f^A)^{-1} \Pi_p,$$
(3-13)

where I is the fourth-order identity matrix, Π_p is the fourth-order tensor that is the

projection onto the space \mathcal{V}_p , and C_f^A is the relevant Avellaneda material. In the basis (3-5), Π_p is represented by the 6 × 6 matrix that has the block form

$$\Pi_p = \begin{pmatrix} 0 & 0\\ 0 & \boldsymbol{I}_p \end{pmatrix}, \tag{3-14}$$

where I_p represents the $p \times p$ identity matrix and the 0 on the diagonal represents the $(6-p) \times (6-p)$ zero matrix.

In the case d = 2 the analysis simplifies in the obvious way. We have the inequalities

$$0 \leq W_{f}^{0}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}),$$

$$\boldsymbol{\epsilon}_{1}^{0} : [\boldsymbol{C}_{f}^{A}(0, 0, \boldsymbol{\epsilon}_{1}^{0})]\boldsymbol{\epsilon}_{1}^{0} \leq W_{f}^{1}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\epsilon}_{1}^{0}),$$

$$\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{f}^{A}(0, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0})\boldsymbol{\epsilon}_{i}^{0} \leq W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}),$$
(3-15)

the first one of which is sharp in the limit $\delta \to \infty$ being attained when the material consists of islands of phase 1 surrounded by phase 2. The recipe for showing that the bound (3-15) on $W_f^1(\sigma_1^0, \sigma_2^0, \epsilon_1^0)$ is sharp for certain values of σ_1^0 and σ_2^0 and that the bound (2-5) on $W_f^2(\sigma_1^0, \epsilon_1^0, \epsilon_2^0)$ is sharp for certain values of σ_1^0 is almost exactly the same as in the 3-dimensional case: insert into the Avellaneda material a thin walled structure of unimode and bimode materials, respectively, so that it is very stiff to any stress in the space spanned by σ_1^0 and σ_2^0 in the case of W_f^1 , or so that it is very stiff to the stress σ_1^0 in the case of W_f^2 .

4. The algebraic problem: characterizing those symmetric matrix pencils spanned by zero determinant matrices

Now we are interested in the following question: Given k linearly independent symmetric $d \times d$ matrices A_1, A_2, \ldots, A_k , find necessary and sufficient conditions such that there exists linearly independent matrices $\{B_i\}_{i=1}^k$ spanned by the basis elements A_i such that $\det(B_i) = 0$. It is assumed that d = 2 or 3 and $1 \le k \le k_d$, where $k_2 = 2$ and $k_3 = 5$. Here we are working in the generic situation, i.e., we prove the algebraic result for a dense set of matrices. The continuity result of Section 5 will allow us to conclude for the whole set of matrices. Actually, the proof below also shows that the algebraic result holds for the complement of a zero measure set of matrices.

Theorem 4.1. The above problem is solvable if and only if the matrices A_i for i = 1, ..., k satisfy the following conditions:

(i)
$$det(A_1) = 0$$
, if $k = 1, d = 2, 3$. (4-1)

(ii)
$$(\alpha_1 \gamma_2 + \alpha_2 \gamma_1 - 2\beta_1 \beta_2)^2 > 4 \det(A_1) \det(A_2), \quad \text{if } k = d = 2,$$
 (4-2)

where

$$A_i = \begin{pmatrix} \alpha_i & \beta_i \\ \beta_i & \gamma_i \end{pmatrix}. \tag{4-3}$$

(iii)
$$\triangle = 18 \det(A_1) \det(A_2) S_1 S_2 - 4 S_1^3 \det(A_2) + S_1^2 S_2^2 - 4 S_2^3 \det(A_1) - 27 \det(A_1)^2 \det(A_2)^2 > 0, \quad if \ k = 2, \ d = 3, \quad (4-4)$$

where $S_i = \sum_{j=1}^{3} s_{ij}$ for i = 1, 2 and s_{ij} is the determinant of the matrix obtained by replacing the *j*-th row of A_i by the *j*-th row of A_{i+1} , where by convention we have $A_3 = A_1$.

(iv) Always solvable if
$$k \ge 3$$
, $d = 3$. (4-5)

Remark. In fact, the condition (4-1), that $det(A_1) = 0$, could be excluded since we are considering the generic case. It is inserted because we can treat it explicitly.

Proof. We consider all the cases separately.

Case (i): k = 1. In this case one must obviously have $det(A_1) = 0$.

Case (ii): k = 2, d = 2. We can without loss of generality assume that (by small perturbations) det $(A_i) \neq 0$ for i = 1, 2. For $\eta, \mu \in \mathbb{R}^2$, denote $A(\eta, \mu) = \eta A_1 + \mu A_2$, and thus for the equality

$$\det(A(\eta, \mu)) = \det(A_1)\eta^2 + (\alpha_1\gamma_2 + \alpha_2\gamma_1 - 2\beta_1\beta_2)\eta\mu + \det(A_2)\mu^2$$
(4-6)

to happen, one must first of all have $\mu \neq 0$; thus, dividing by μ^2 and setting $t = \eta/\mu$, we get that the quadratic equation

$$\frac{1}{\mu^2} \det(A(\eta, \mu)) = \det(A_1)t^2 + (\alpha_1\gamma_2 + \alpha_2\gamma_1 - 2\beta_1\beta_2)t + \det(A_2) = 0 \quad (4-7)$$

must have two different solutions, i.e., the discriminant is strictly positive, which amounts to exactly (4-2).

Case (iii): k = 2, d = 3. Again, we can without loss of generality assume that $det(A_i) \neq 0$ for i = 1, 2. Set again $A(\eta, \mu) = \eta A_1 + \mu A_2$; thus we must have that the equation

$$\det(A(\eta,\mu)) = \det(A_1)\eta^3 + S_1\eta^2\mu + S_2\eta\mu^2 + \det(A_2)\mu^3 = 0$$
(4-8)

has at least two different real roots, which by Cardan's condition gives

$$\Delta = 18 \det(A_1) \det(A_2) S_1 S_2 - 4S_1^3 \det(A_2) + S_1^2 S_2^2 - 4S_2^3 \det(A_1) - 27 \det(A_1)^2 \det(A_2)^2 > 0, \quad (4-9)$$

which is exactly (4-4).

Case (iv): $k \ge 3$, d = 3. Let us consider the case k = 3 first. Let us show that we can assume, without loss of generality, that $\det(A_1) = \det(A_2) = 0$, by proving that there exist numbers $\eta_i \ne 0$ for i = 1, 2 such that the matrices $B_1 = \eta_1 A_1 + A_2$ and $B_2 = \eta_2 A_1 + A_3$ have zero determinant. Indeed, we assume without loss of generality that $\det(A_i) \ne 0$ for i = 1, 2, 3. We would then like to have

$$\det(\boldsymbol{B}_1)(\eta_1) = \eta_1^3 \det(\boldsymbol{A}_1) + \eta_1^2(\cdot) + \eta_1(\cdot) + \det(\boldsymbol{A}_2) = 0, \quad (4-10)$$

which has a nonzero root η_1 , being a cubic equation with det $(\boldsymbol{B}_1)(0) = \det(\boldsymbol{A}_2) \neq 0$. Similarly, the equation det $(\boldsymbol{B}_2)(\eta_2) = 0$ has a nonzero solution η_2 . The matrices \boldsymbol{B}_1 , \boldsymbol{B}_2 , and \boldsymbol{A}_1 are linearly independent, because the linear independence of \boldsymbol{B}_1 , \boldsymbol{B}_2 , and \boldsymbol{A}_1 is equivalent to the condition

$$\det \begin{pmatrix} \eta_1 & 1 & 0\\ \eta_2 & 0 & 1\\ 1 & 0 & 0 \end{pmatrix} = 1 \neq 0.$$
(4-11)

Assume now that A_1 , A_2 , and A_3 are linearly independent and

$$\det(A_1) = \det(A_2) = 0. \tag{4-12}$$

For any $\eta, \mu \in \mathbb{R}$, consider the matrix

$$\boldsymbol{B}_3 = \boldsymbol{B}(\eta, \mu) = \boldsymbol{A}_3 + \eta \boldsymbol{A}_1 + \mu \boldsymbol{A}_2.$$

It is clear that the triple A_1, A_2, B_3 is linearly independent, so we would like to show that there exist $\eta, \mu \in \mathbb{R}$, such that $det(B_3) = 0$. Assume, by contradiction, that

$$\det(\boldsymbol{B}_3) \neq 0, \quad \text{for all } \eta, \, \mu \in \mathbb{R}. \tag{4-13}$$

Let us then show that the condition (4-13) implies that $c_1 = c_2 = 0$, where, taking into account the condition (4-12), we have that

$$\det(\boldsymbol{B}_3) = c_1 \eta^2 \mu + c_2 \eta \mu^2 + c_3 \eta \mu + c_4 \eta^2 + c_5 \mu^2 + c_6 \eta + c_7 \mu + \det(\boldsymbol{A}_3).$$
(4-14)

Indeed, if $c_1 \neq 0$, then taking $\eta = \mu^2$ we get that the equation $\det(\boldsymbol{B}(\mu^2, \mu)) = 0$ would have a solution $\mu \in \mathbb{R}$, being a fifth-order equation; thus, we get $c_1 = c_2 = 0$. Next, by perturbing the elements of A_1 and A_2 if necessary, we can reach the situation where no entries and second-order minors of both A_1 and A_2 vanish, by first reaching the situation where A_1 and A_2 have no zero entries. If we now perturb any *ij* and *ik* elements of A_1 by small numbers ϵ and δ , where $j \neq k$, then to keep the condition $\det(A_1) = 0$, we must have ϵ and δ satisfying

$$\epsilon \cdot \operatorname{cof}_{ij}(A_1) + \delta \cdot \operatorname{cof}_{ik}(A_1) = 0.$$
(4-15)
On the other hand, the condition $c_2 = 0$ must not be violated by that perturbation, thus we must have as well

$$\epsilon \cdot \operatorname{cof}_{ii}(A_2) + \delta \cdot \operatorname{cof}_{ik}(A_2) = 0.$$
(4-16)

The last two conditions then imply that the cofactor matrix $cof A_1$ is a multiple of the cofactor matrix $cof A_2$, i.e.,

$$\operatorname{cof}(A_2) = t \cdot \operatorname{cof}(A_1), \quad t \neq 0. \tag{4-17}$$

Again, a small perturbation of the 11 and 12 elements of A_1 by ϵ and δ satisfying (4-15) with i = j = 1, k = 2 does not violate the condition det $(A_1) = 0$, thus it must not violate the condition (4-16). Observe that the above perturbation does not change the cofactor cof₁₁ (A_1) , but it changes the cofactor element cof₃₃ (A_1) , which means that the desired condition det $(B_3) = 0$ can be reached by small perturbations. The case k = d = 3 is now done.

Assume now $k \ge 4$ and d = 3. By the previous step, in the space spanned by A_1, A_2 , and A_3 there are three matrices A'_1, A'_2 , and $B_3 = A_3 + \eta_3 A'_1 + \mu_3 A'_2$ that are linearly independent with zero determinant. Then, again by the previous step, we can find linearly independent matrices B_1, \ldots, B_k that have the form $B_1 = A'_1$, $B_2 = A'_2$, and $B_i = A_i + \eta_i A'_1 + \mu_i A'_2$ for $3 \le i \le k$ and that are linearly independent and have zero determinant.

5. Continuity of the energy functions

It follows from the preceding analysis that we can determine the three energy functions

$$\begin{split} & W_f^1(\pmb{\sigma}_1^0, \pmb{\sigma}_2^0, \pmb{\sigma}_3^0, \pmb{\sigma}_4^0, \pmb{\sigma}_5^0, \pmb{\epsilon}_1^0), \\ & W_f^2(\pmb{\sigma}_1^0, \pmb{\sigma}_2^0, \pmb{\sigma}_3^0, \pmb{\sigma}_4^0, \pmb{\epsilon}_1^0, \pmb{\epsilon}_2^0), \\ & W_f^3(\pmb{\sigma}_1^0, \pmb{\sigma}_2^0, \pmb{\sigma}_3^0, \pmb{\epsilon}_1^0, \pmb{\epsilon}_2^0, \pmb{\epsilon}_3^0) \end{split}$$

in the limit $\delta \to \infty$ for almost all combinations of applied fields. Here we establish that these energy functions are continuous functions of the applied fields in the limit $\delta \to \infty$, and therefore we obtain expressions for the energy functions for all combinations of applied fields in this limit.

Recall that the set GU_f is characterized by its *W*-transform. For example, part of it is described by the function

$$W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}) = \min_{\boldsymbol{C}_{*} \in GU_{f}} \left[\sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{4} \boldsymbol{\sigma}_{j}^{0} : \boldsymbol{C}_{*}^{-1} \boldsymbol{\sigma}_{j}^{0} \right].$$
(5-1)

Here we want to show that such energy functions are continuous in their arguments. Let the compliance tensor $[C_*(\sigma_1^0, \sigma_2^0, \sigma_3^0, \sigma_4^0, \epsilon_1^0, \epsilon_2^0)]^{-1}$ be a minimizer of (5-1),

and suppose we perturb the applied stress fields σ_j^0 by $\delta \sigma_j^0$ and the applied strain fields ϵ_i^0 by $\delta \epsilon_i^0$. Now consider the walled material with a geometry described by the characteristic function

$$\chi_w(\boldsymbol{x}) = \prod_{k=1}^3 (1 - H_{\epsilon'}(\boldsymbol{x} \cdot \boldsymbol{n}_k)), \qquad (5-2)$$

where n_1 , n_2 , and n_3 are the three orthogonal unit vectors

$$\boldsymbol{n}_1 = \begin{pmatrix} 1\\0\\0 \end{pmatrix}, \quad \boldsymbol{n}_2 = \begin{pmatrix} 0\\1\\0 \end{pmatrix}, \quad \boldsymbol{n}_3 = \begin{pmatrix} 0\\0\\1 \end{pmatrix}, \quad (5-3)$$

and ϵ' is a small parameter that gives the thickness of the walls. Inside the walls, where $\chi_w(\mathbf{x}) = 0$, we put an isotropic composite of phase 1 and phase 2, mixed in the proportions f and 1 - f with isotropic effective elasticity tensor $C(\kappa_0, \mu_0)$, where κ_0 is the effective bulk modulus and μ_0 is the effective shear modulus, which are assumed to have finite limits as $\delta \to \infty$. (The isotropic composite could consist of islands of void surrounded by phase 1.) Outside the walls, where $\chi_w(\mathbf{x}) = 1$, we put the material that has effective compliance tensor

$$[\boldsymbol{C}_*^1]^{-1} = [\boldsymbol{C}_*(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0, \boldsymbol{\sigma}_4^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0)]^{-1}.$$

Let C'_* be the effective tensor of the composite. We have the variational principle

$$\sum_{i=1}^{2} (\boldsymbol{\epsilon}_{i}^{0} + \delta \boldsymbol{\epsilon}_{i}^{0}) : \boldsymbol{C}_{*}'(\boldsymbol{\epsilon}_{i}^{0} + \delta \boldsymbol{\epsilon}_{i}^{0}) + \sum_{j=1}^{4} (\boldsymbol{\sigma}_{j}^{0} + \delta \boldsymbol{\sigma}_{j}^{0}) : (\boldsymbol{C}_{*}')^{-1} (\boldsymbol{\sigma}_{j}^{0} + \delta \boldsymbol{\sigma}_{j}^{0})$$

$$= \min_{\boldsymbol{\epsilon}_{1}, \boldsymbol{\epsilon}_{2}, \boldsymbol{\epsilon}_{3}, \boldsymbol{\epsilon}_{4}, \boldsymbol{\sigma}_{1}, \boldsymbol{\sigma}_{2}} \left\langle \sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}(\boldsymbol{x}) : [\boldsymbol{\chi}_{w}(\boldsymbol{x}) \boldsymbol{C}_{*}^{1} + (1 - \boldsymbol{\chi}_{w}(\boldsymbol{x})) \boldsymbol{C}(\boldsymbol{\kappa}_{0}, \boldsymbol{\mu}_{0})] \boldsymbol{\epsilon}_{i}(\boldsymbol{x})$$

$$+ \sum_{j=1}^{4} \boldsymbol{\sigma}_{j}(\boldsymbol{x}) : [\boldsymbol{\chi}_{w}(\boldsymbol{x}) \boldsymbol{C}_{*}^{1} + (1 - \boldsymbol{\chi}_{w}(\boldsymbol{x})) \boldsymbol{C}(\boldsymbol{\kappa}_{0}, \boldsymbol{\mu}_{0})]^{-1} \boldsymbol{\sigma}_{j}(\boldsymbol{x}) \right\rangle, \quad (5-4)$$

where the minimum is over fields subject to the appropriate average values and differential constraints. We choose constant trial stress fields

$$\underline{\sigma}_j(\mathbf{x}) = \sigma_j^0 + \delta \sigma_j^0, \quad j = 1, 2, 3, 4,$$
(5-5)

and trial strain fields

$$\underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) = \boldsymbol{\epsilon}_{i}^{0} + \delta \underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}), \quad i = 1, 2,$$
(5-6)

where $\delta \underline{\epsilon}_i(\mathbf{x})$ has average value $\delta \epsilon_i^0$ and is concentrated in the walls. Specifically,

if $\{\delta \epsilon_i^0\}_{k\ell}$ denote the matrix elements of $\delta \epsilon_i^0$, and letting

$$\delta \boldsymbol{\epsilon}_{i}^{1} = \begin{pmatrix} \{\delta \boldsymbol{\epsilon}_{i}^{0}\}_{11} & \{\delta \boldsymbol{\epsilon}_{i}^{0}\}_{12} & 0\\ \{\delta \boldsymbol{\epsilon}_{i}^{0}\}_{21} & 0 & 0\\ 0 & 0 & 0 \end{pmatrix},$$

$$\delta \boldsymbol{\epsilon}_{j}^{2} = \begin{pmatrix} 0 & 0 & 0\\ 0 & \{\delta \boldsymbol{\epsilon}_{i}^{0}\}_{22} & \{\delta \boldsymbol{\epsilon}_{i}^{0}\}_{23}\\ 0 & \{\delta \boldsymbol{\epsilon}_{i}^{0}\}_{32} & 0 \end{pmatrix},$$

$$\delta \boldsymbol{\epsilon}_{j}^{3} = \begin{pmatrix} 0 & 0 & \{\delta \boldsymbol{\epsilon}_{i}^{0}\}_{13}\\ 0 & 0 & 0\\ \{\delta \boldsymbol{\epsilon}_{i}^{0}\}_{31} & 0 & \{\delta \boldsymbol{\epsilon}_{i}^{0}\}_{33} \end{pmatrix},$$

(5-7)

then we choose

$$\delta \underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) = \sum_{k=1}^{3} \delta \boldsymbol{\epsilon}_{i}^{k} H_{\epsilon'}(\boldsymbol{x} \cdot \boldsymbol{n}_{k}) / \epsilon', \qquad (5-8)$$

which has the required average value $\delta \sigma_j^0$ and satisfies the differential constraints appropriate to a strain field because $\delta \epsilon_i^k = a_{i,k} n_k^T + n_k a_{i,k}^T$ for some vector $a_{i,k}$. Hence, there exist constants α and β such that for sufficiently small ϵ' and for

sufficiently small variations $\delta \sigma_i^0$ and $\delta \epsilon_i^0$ in the applied fields, we have

$$\left\langle \sum_{i=1}^{2} \underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) : [\boldsymbol{\chi}_{w}(\boldsymbol{x})\boldsymbol{C}_{*}^{1} + (1 - \boldsymbol{\chi}_{w}(\boldsymbol{x}))\boldsymbol{C}(\boldsymbol{\kappa}_{0}, \boldsymbol{\mu}_{0})]\underline{\boldsymbol{\epsilon}}_{i}(\boldsymbol{x}) + \sum_{j=1}^{4} \underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x}) : [\boldsymbol{\chi}_{w}(\boldsymbol{x})\boldsymbol{C}_{*}^{1} + (1 - \boldsymbol{\chi}_{w}(\boldsymbol{x}))\boldsymbol{C}(\boldsymbol{\kappa}_{0}, \boldsymbol{\mu}_{0})]^{-1}\underline{\boldsymbol{\sigma}}_{j}(\boldsymbol{x}) \right\rangle$$
$$\leq W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}) + \alpha\boldsymbol{\epsilon}' + \beta K/\boldsymbol{\epsilon}', \quad (5-9)$$

where K represents the norm

$$K = \sqrt{\sum_{i=1}^{2} \delta \boldsymbol{\epsilon}_{i}^{0} : \delta \boldsymbol{\epsilon}_{i}^{0} + \sum_{j=1}^{4} \delta \boldsymbol{\sigma}_{j}^{0} : \delta \boldsymbol{\sigma}_{j}^{0}}$$
(5-10)

of the field variations. Choosing $\epsilon' = \sqrt{\beta K/\alpha}$ to minimize the right-hand side of (5-9), we obtain

$$W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0}+\delta\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0}+\delta\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\sigma}_{3}^{0}+\delta\boldsymbol{\sigma}_{3}^{0},\boldsymbol{\sigma}_{4}^{0}+\delta\boldsymbol{\sigma}_{4}^{0},\boldsymbol{\epsilon}_{1}^{0}+\delta\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0}+\delta\boldsymbol{\epsilon}_{2}^{0})$$

$$\leq W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\sigma}_{3}^{0},\boldsymbol{\sigma}_{4}^{0},\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0})+2\sqrt{\alpha\beta K}.$$
 (5-11)

Clearly the right-hand side approaches $W_f^2(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0, \boldsymbol{\sigma}_4^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0)$ as $K \to 0$. On the other hand, by repeating the same argument with the roles of

$$W_f^2(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0, \boldsymbol{\sigma}_4^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0)$$

and

$$W_f^2(\boldsymbol{\sigma}_1^0 + \delta\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0 + \delta\boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0 + \delta\boldsymbol{\sigma}_3^0, \boldsymbol{\sigma}_4^0 + \delta\boldsymbol{\sigma}_4^0, \boldsymbol{\epsilon}_1^0 + \delta\boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0 + \delta\boldsymbol{\epsilon}_2^0)$$

reversed, and with the compliance tensor

$$[\boldsymbol{C}_{*}(\boldsymbol{\sigma}_{1}^{0}+\delta\boldsymbol{\sigma}_{1}^{0},\boldsymbol{\sigma}_{2}^{0}+\delta\boldsymbol{\sigma}_{2}^{0},\boldsymbol{\epsilon}_{1}^{0}+\delta\boldsymbol{\epsilon}_{1}^{0},\boldsymbol{\epsilon}_{2}^{0}+\delta\boldsymbol{\epsilon}_{2}^{0},\boldsymbol{\epsilon}_{3}^{0}+\delta\boldsymbol{\epsilon}_{3}^{0},\boldsymbol{\epsilon}_{4}^{0}+\delta\boldsymbol{\epsilon}_{4}^{0})]^{-1}$$

replacing the compliance tensor

$$[\boldsymbol{C}_*(\boldsymbol{\sigma}_1^0,\boldsymbol{\sigma}_2^0,\boldsymbol{\sigma}_3^0,\boldsymbol{\sigma}_4^0,\boldsymbol{\epsilon}_1^0,\boldsymbol{\epsilon}_2^0)]^{-1},$$

we deduce that

$$W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}) \\ \leq W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0} + \delta\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0} + \delta\boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0} + \delta\boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0} + \delta\boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\epsilon}_{1}^{0} + \delta\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0} + \delta\boldsymbol{\epsilon}_{2}^{0}) \\ + 2\sqrt{\alpha\beta K}. \quad (5-12)$$

This, together with (5-11), establishes the continuity of $W_f^2(\sigma_1^0, \sigma_2^0, \sigma_3^0, \sigma_4^0, \epsilon_1^0, \epsilon_2^0)$. The continuity of the other energy functions follows by the same argument.

6. Conclusion

We have established the following theorems.

Theorem 6.1. Consider composites in three dimensions of two materials with positive definite elasticity tensors C_1 and $C_2 = \delta C_0$ mixed in proportions f and 1 - f. Let the seven energy functions W_f^k , for k = 0, 1, ..., 6, that characterize the set GU_f (with $U = (C_1, \delta C_0)$) of possible elastic tensors be defined by (2-1). These energy functions involve a set of applied strains ϵ_i^0 and applied stresses σ_j^0 meeting the orthogonality condition (2-2). The energy function W_f^6 is given by

$$W_{f}^{6}(\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}, \boldsymbol{\epsilon}_{6}^{0}) = \sum_{i=1}^{6} \boldsymbol{\epsilon}_{i}^{0} : \boldsymbol{C}_{f}^{A}(\boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}, \boldsymbol{\epsilon}_{4}^{0}, \boldsymbol{\epsilon}_{5}^{0}, \boldsymbol{\epsilon}_{6}^{0}) \boldsymbol{\epsilon}_{i}^{0}$$
(6-1)

(as established by Avellaneda [1987]), where $C_f^A(\epsilon_1^0, \epsilon_2^0, \epsilon_3^0, \epsilon_4^0, \epsilon_5^0, \epsilon_6^0)$ is the effective elasticity tensor of an Avellaneda material that is a sequentially layered laminate with the minimum value of the sum of elastic energies

$$\sum_{i=1}^{6} \boldsymbol{\epsilon}_{j}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{j}^{0}.$$
(6-2)

Again some of the applied stresses $\boldsymbol{\sigma}_{j}^{0}$ or applied strains $\boldsymbol{\epsilon}_{i}^{0}$ could be zero. Additionally, we have

$$\lim_{\delta \to \infty} W_{f}^{0}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\sigma}_{5}^{0}, \boldsymbol{\sigma}_{6}^{0}) = 0,$$

$$\lim_{\delta \to \infty} W_{f}^{1}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\sigma}_{5}^{0}, \boldsymbol{\epsilon}_{1}^{0}) = \boldsymbol{\epsilon}_{1}^{0} : [\boldsymbol{C}_{f}^{A}(0, 0, 0, 0, 0, \boldsymbol{\epsilon}_{1}^{0})]\boldsymbol{\epsilon}_{1}^{0},$$

$$\lim_{\delta \to \infty} W_{f}^{2}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\sigma}_{4}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}) = \sum_{i=1}^{2} \boldsymbol{\epsilon}_{i}^{0} : [\boldsymbol{C}_{f}^{A}(0, 0, 0, 0, 0, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0})]\boldsymbol{\epsilon}_{i}^{0}, \qquad (6-3)$$

$$\lim_{\delta \to \infty} W_{f}^{3}(\boldsymbol{\sigma}_{1}^{0}, \boldsymbol{\sigma}_{2}^{0}, \boldsymbol{\sigma}_{3}^{0}, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0}) = \sum_{i=1}^{3} \boldsymbol{\epsilon}_{i}^{0} : [\boldsymbol{C}_{f}^{A}(0, 0, 0, 0, \boldsymbol{\epsilon}_{1}^{0}, \boldsymbol{\epsilon}_{2}^{0}, \boldsymbol{\epsilon}_{3}^{0})]\boldsymbol{\epsilon}_{i}^{0},$$

for all combinations of applied stresses σ_j^0 and applied strains ϵ_i^0 . In the case that $det(\sigma_1^0) = 0$, we have

$$\lim_{\delta \to \infty} W_f^5(\boldsymbol{\sigma}_1^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0, \boldsymbol{\epsilon}_4^0, \boldsymbol{\epsilon}_5^0) = \sum_{i=1}^5 \boldsymbol{\epsilon}_i^0 : [\boldsymbol{C}_f^A(0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0, \boldsymbol{\epsilon}_4^0, \boldsymbol{\epsilon}_5^0)] \boldsymbol{\epsilon}_i^0, \quad (6-4)$$

while, when $f(t) = \det(\sigma_1^0 + t\sigma_2^0)$ has at least two roots (the condition for which is given by (4-4)),

$$\lim_{\delta \to \infty} W_f^4(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0, \boldsymbol{\epsilon}_4^0) = \sum_{i=1}^4 \boldsymbol{\epsilon}_i^0 : [\boldsymbol{C}_f^A(0, 0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0, \boldsymbol{\epsilon}_4^0)] \boldsymbol{\epsilon}_i^0.$$
(6-5)

Theorem 6.2. For 2-dimensional composites, the four energy functions W_f^k , for k = 0, 1, 2, 3, are defined by (2-5), and these characterize the set GU_f , with $U = (C_1, \delta C_0)$, of possible elastic tensors C_* of composites of two phases with positive definite elasticity tensors C_1 and $C_2 = \delta C_0$. These energy functions involve a set of applied strains ϵ_i^0 and applied stresses σ_j^0 meeting the orthogonality condition (2-2). The energy function W_f^3 is given by

$$W_f^3(\boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0) = \sum_{i=1}^3 \boldsymbol{\epsilon}_i^0 : \boldsymbol{C}_f^A(\boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0, \boldsymbol{\epsilon}_3^0) \boldsymbol{\epsilon}_i^0$$
(6-6)

(as proved by Avellaneda [1987]), where $C_f^A(\epsilon_1^0, \epsilon_2^0, \epsilon_3^0)$ is the effective elasticity tensor of an Avellaneda material that is a sequentially layered laminate with the minimum value of the sum of elastic energies

$$\sum_{j=1}^{3} \boldsymbol{\epsilon}_{j}^{0} : \boldsymbol{C}_{*} \boldsymbol{\epsilon}_{j}^{0}.$$
(6-7)

We also have the trivial result that

$$\lim_{\delta \to \infty} W_f^0(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\sigma}_3^0) = 0.$$
(6-8)

When det $\sigma_1^0 = 0$, we have

$$\lim_{\delta \to \infty} W_f^2(\boldsymbol{\sigma}_1^0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0) = \sum_{i=1}^2 \boldsymbol{\epsilon}_i^0 : [\boldsymbol{C}_f^A(0, \boldsymbol{\epsilon}_1^0, \boldsymbol{\epsilon}_2^0)] \boldsymbol{\epsilon}_i^0,$$
(6-9)

while when $f(t) = \det(\sigma_1^0 + t\sigma_2^0)$ has exactly two roots (the condition for which is given by (4-2)),

$$\lim_{\delta \to \infty} W_f^1(\boldsymbol{\sigma}_1^0, \boldsymbol{\sigma}_2^0, \boldsymbol{\epsilon}_1^0) = \boldsymbol{\epsilon}_1^0 : [\boldsymbol{C}_f^A(0, 0, \boldsymbol{\epsilon}_1^0)] \boldsymbol{\epsilon}_1^0.$$
(6-10)

These theorems, and the accompanying microstructures, help define what sort of elastic behaviors are theoretically possible in 2- and 3-dimensional materials consisting of a very stiff phase and an elastic phase (possibly anisotropic, but with fixed orientation). They should serve as benchmarks for the construction of more realistic microstructures that can be manufactured. We have found the minimum over all microstructures of various sums of energies and complementary energies.

It remains an open problem to find expressions for the energy functions in the cases not covered by these theorems. Notice that for 3-dimensional composites the function W_f^5 is only determined when the special condition $\det(\sigma_1^0) = 0$ is satisfied exactly. Similarly, for 2-dimensional composites the function W_f^2 is only determined when the special condition $\det \sigma_1^0 = 0$ is satisfied exactly. Thus these functions are only known on a set of zero measure.

Even for an isotropic composite with a bulk modulus κ_* and a shear modulus μ_* , the set of all possible pairs (κ_* , μ_*) is still not completely characterized either in the limit $\delta \to \infty$. In these limits the bounds of Berryman and Milton [1988] and Cherkaev and Gibiansky [1993] decouple and provide no extra information beyond that provided by the Hashin–Shtrikman–Hill bounds [Hashin and Shtrikman 1963; Hashin 1965; Hill 1963; 1964]. While the results of this paper show that in the limit $\delta \to \infty$ one can obtain 3-dimensional structures attaining the Hashin–Shtrikman–Hill lower bound on κ_* , while having $\mu_* = \infty$, it is not clear what the minimum value for μ_* is, given that $\kappa_* = \infty$, nor is it clear in two dimensions what the minimum value of κ_* is when $\mu_* = \infty$.

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